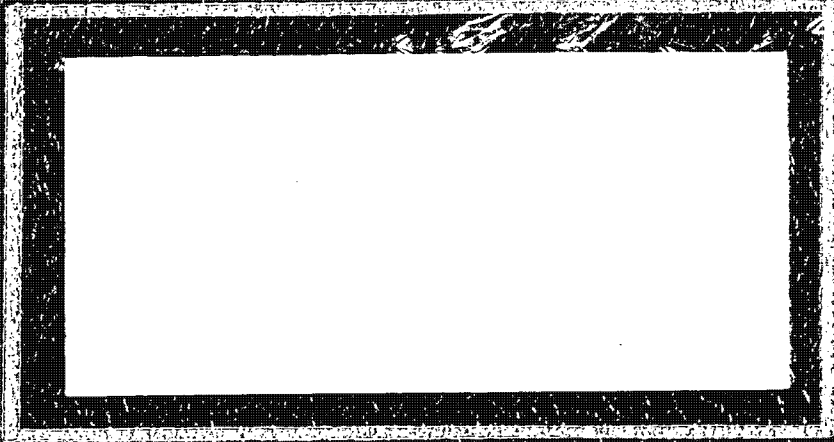


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**DIRECTORATE OF SPECIAL PROJECTS  
OFFICE OF THE SECRETARY OF THE AIR FORCE**



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SYSTEM  
PERFORMANCE EVALUATION TEAM  
(FLIGHT MISSION CHARACTERISTICS)  
MISSION 4022/65

This report consists of 142 pages.

Cy 5 of 5 copies

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PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65

FOREWORD

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Preparing Unit:

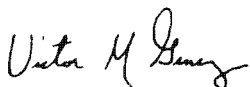
Performance Evaluation Team  
Los Angeles AF Station  
Los Angeles, California 90045

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PUBLICATION REVIEW

This report has been reviewed and is approved.



VICTOR M. GENEZ  
Colonel, USAF  
Team Manager

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PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65

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PERFORMANCE EVALUATION TEAM  
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## SECTION I

## RESUME OF MISSION 4022

## A. INTRODUCTION

Mission 4022, consisting of the GAMBIT Camera system and the Orbital Control Vehicle (OCV), was launched into orbit from Point Arguello Launch Complex, Pad 3, at 1223:38Z on 30 September 1965.

## B. OBJECTIVES

The operational objectives of this mission were to conduct:

1. A four-day reconnaissance mission to obtain high resolution photography of specific targets selected by the NRO staff.
2. R&D experiments to obtain exposure and filter data using special test filter configurations in Slit position A.
3. Continuing R&D experiments to evaluate focus sensor mechanism, film set data, exposure tests, inner shield operations, thermal control, OCV power supply, capability of augmented tracking stations to support orbital operations, and BUSS capabilities.
4. Fifth day OCV solo operations to demonstrate vehicle attitude control and full system capability over a five day period.

## C. MISSION INFORMATION

## 1. Powered Flight and Orbital Parameters

The Standard Atlas/Agena combination which boosted the satellite into orbit had the following powered flight ascent parameters:

	NOMINAL (SEC)	ACTUAL (SEC)
BOOSTER ENGINE CUT-OFF	T -/ 132.40	T -/ 130.14
BOOSTER ENGINE JETTISON	T -/ 135.40	T -/ 133.35
SUSTAINER ENGINE CUT-OFF	T -/ 273.92	T -/ 280.05
VERNIER ENGINE CUT-OFF	T -/ 293.60	T -/ 298.71
BOOSTER SEPARATION	T -/ 296.10	T -/ 301.21
AGENA IGNITION	T -/ 331.17	T -/ 337.14
AGENA CUT-OFF	T -/ 566.89	T -/ 575.04

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The satellite (OCV) achieved a near-nominal orbit with the following elements:

	NOMINAL	ACTUAL
INCLINATION (DEGREES)	95.5	95.56
PERIOD (MINUTES)	88.85	88.91
PERIGEE (NM)	85.9	85.05
APOGEE (NM)	157.11	160.8
ECCENTRICITY	.0095	.0098
ARGUMENT OF PERIGEE (DEGREES)	137.44	141.67

## 2. Recovery

The capsule was de-orbited on 4 October 1965 on Rev 67 and was recovered by air catch. The OCV solo exercise was terminated four revs early due to power depletion. The OCV was de-boosted on 5 October on Rev 77 with impact predicted in the South Atlantic at 16° 22' N 32° 25' W.

## D. RESULTS

The operational objectives of Mission 4022 were satisfactorily accomplished despite the fact that high stabilization control gas consumption required budgeting on Day 4 and resulted in the elimination of some low priority primary targets as well as several R&D operations. This mission resulted in a total of 198 stereo pairs, 95 framed stereo pairs, 9 lateral pairs and 206 strip frames utilizing 2699 feet of film in the primary camera. The GRAVY program indicates that a total of 963 COMOR targets were available, of which 543 were photographed in cloud free areas. (See NPIC comments below.)

## E. ANOMALIES

Only one serious anomaly was noted:

An apparent high consumption rate of stabilization control gas required the reduction of roll maneuvers on Day 4. The unscheduled depletion was apparently caused by a high rate of usage by this vehicle for orbit stabilization maintenance as well as the increase in numbers of a new target category type. This problem is under study by the contractor for development of specific procedures which will prevent a recurrence of unscheduled depletion. Completion of the study is scheduled prior to the next mission.

## F. OVERALL EVALUATION

The following subjective evaluation of Mission 4022 is submitted by the National Photographic Interpretation Center (NPIC).

1. The intelligence content of the Mission is one of the highest to date with 543 targets covered in cloud-free or partially cloud-free areas. Most of the important targets were covered with stereo photographs. This target coverage was equivalent with the best previous missions that were of four day

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duration, but not numerically equivalent to a five day mission.

2. The photo interpretation of this mission was the best to date of many of the prime targets that must be covered for intelligence purposes. Most of the prime targets were covered with stereo photographs and proved to be the best photography of the target areas gained to date. The target portion of the readout has been separated into two separate readouts and since the second readout is incomplete at this time the number of targets quoted is subject to change.

3. The acquisition of targets was exceptionally high with no misses due to attitude or coordinates. The only limiting factor to a higher acquisition was the 50% cloud cover experienced in the target areas.

4. The mensuration of imagery on this mission was the best to date and was enhanced by the fact that the stellar/index package worked throughout the mission and adequate stars existed on 1304 frames of the 1767 exposures programmed. Stellar reduction was possible on all of the frames with adequate stars. The mensuration of imagery with known dimensions with accurate camera attitude was within 1/4 of 1 percent on all dimensions long enough to minimize pointing errors and resolution limitations.

5. The film speed variations on this mission were higher than on some previous missions but were generally within the 1% area of the programmed speed. Some excursions beyond the 1% lines do occur but do not seem to degrade the imagery when viewed through high magnification. The excursions of the film speeds from nominal or programmed values was generally within plus or minus 1/2 of 1 percent of the programmed value. The settling times at the start of the frames were approximately .4 seconds with a significant change at about 1.2 seconds after the start of the frame. The rest of the speeds were within the limits stated above. The looper actions were normal to the previous missions with no variations outside the plus or minus 5% lines except at the very start of the looper area. The looper area extended through about 3.8 to 4 seconds in most cases and did not appear to be as violent as it has been on some previous missions.

6. The imagery on this mission was the best achieved to date and was consistent throughout the mission. There was a slight fall off in resolution before the two focus adjustments but the imagery was relatively consistent from pass to pass. Comparison of the same images from Missions 4016, 4017, and 4018 shows that this mission was slightly better than the previous best missions. The best resolution for this mission was about 1.8 feet with the average ground resolution being about 2.0 feet. Mensuration techniques for producing ground resolution show some areas where the resolution reaches 1.8 feet but the resolution decreases in a normal fashion when high obliquity angles occur or with the higher altitudes in some portions of the mission. CORN Target readings were consistent with the mensuration figures above.

7. The time track operated continually throughout the mission and provided the accurate mensuration base necessary in this type system. The yaw slits operated throughout the mission and

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provided some yaw checks against the stellar reduced yaw.

## 8. Stellar/Index Camera

a. The Stellar/Index camera operated throughout the mission in both the dependent mode (with the main camera) and the independent mode for mapping purposes with no master camera operations.

b. There were 1771 programmed frames of which 1767 were acquired. Both the stellar and index acquired the same number of frames and were correlated by both time and pass number. There were 408 frames taken with the vehicle in a positive roll position and the stellar had no chance to acquire stellar fields; of the remaining 1359 exposures 1304 had adequate stars for stellar reduction. This is the best operation of the stellar/index package to date and the stellar field generally contained 10 or more stars. High amounts of background light limited the stellar exposures to small areas near the edge of the frame in many cases and these do not contain the best geometry for reduction but can be reduced in a two star solution. The flared portion of the format was the smallest to date and the overexposure areas were smaller than on previous missions. There are 25 frames with streaked stars due to the vehicle rolling during exposure. These have all been correlated to programmed roll conditions and were caused by a lack of time between exposure cycles to complete the stellar exposure before the vehicle motion was to occur. No double exposures were found in this camera. There were four commanded stellar index firings which did not occur.

c. Time correlations were accomplished by both the M. C. D. and the Telemetry times and they both agreed except for the four commanded frames which did not occur.

d. The final analysis of the stellar fields in the vertical mode and the rolled position of -6.5 degrees from the vertical have not been completed but it appears that the stellar field recorded is slightly better in the -6.5 degree roll position.

e. The index camera provided excellent coverage of some unmapped areas of the world and should prove invaluable to the mapping community.

## 9. A final analysis of the mission shows the following facts:

a. The number of targets covered were comparable to any other four day mission.

b. The resolution was the best of any mission to date.

c. The stellar/index operation was the best to date.

d. The time correlation was the most accurate to date.

e. The intelligence content was higher than any previous mission due to the higher resolution.

f. The resolution was consistent and relatively constant throughout the mission.

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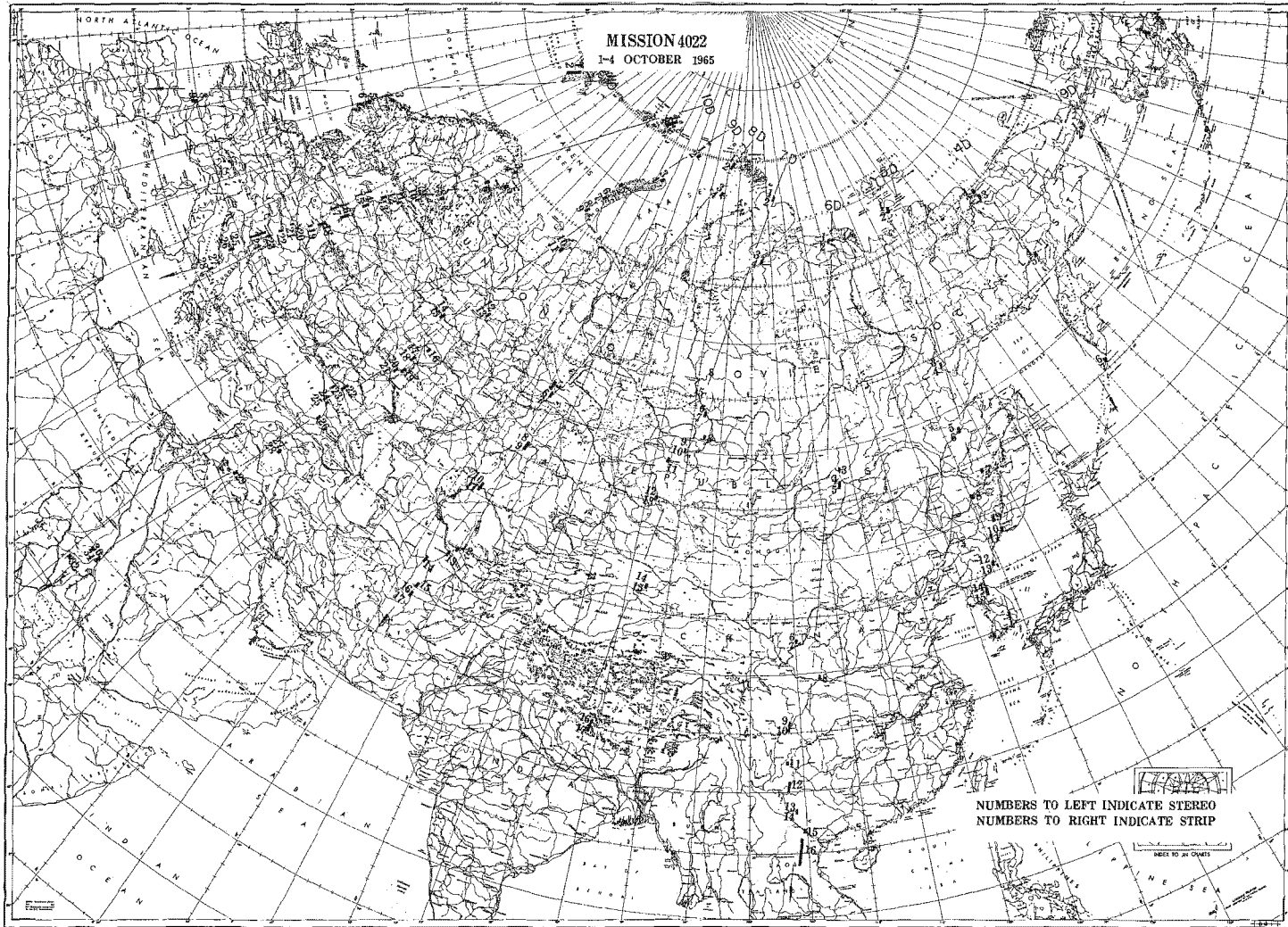


ILLUSTRATION 1

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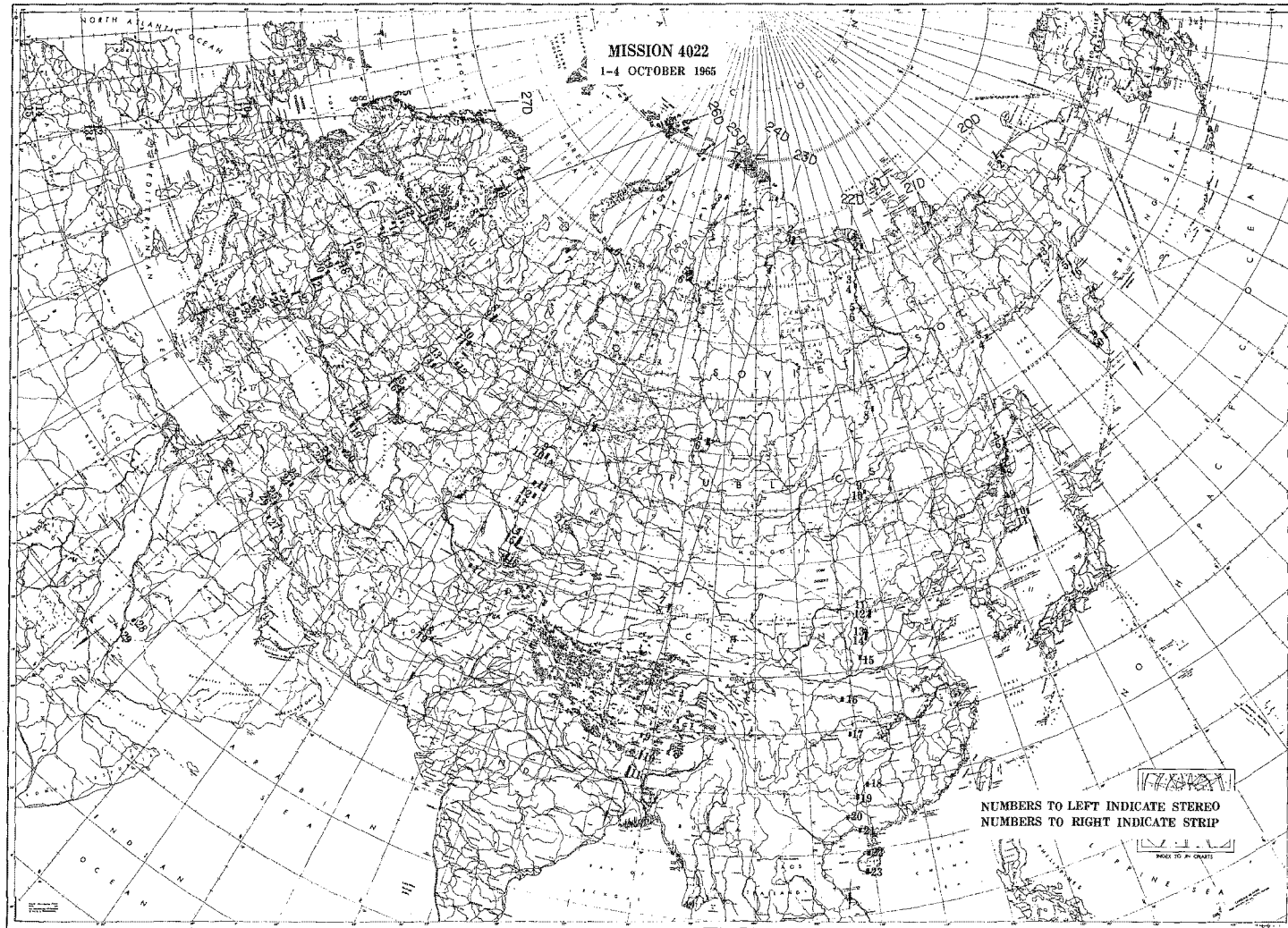


ILLUSTRATION 2

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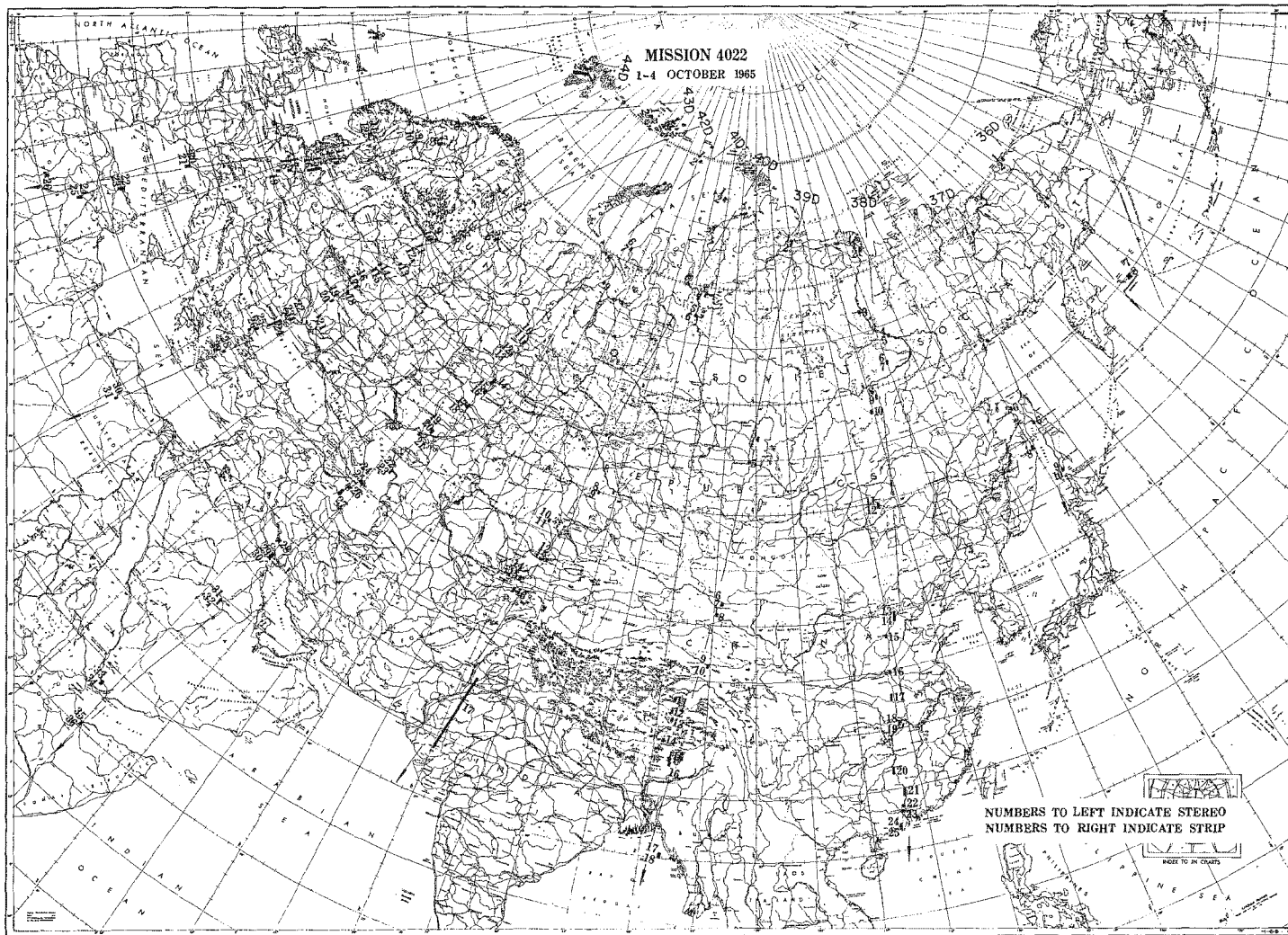


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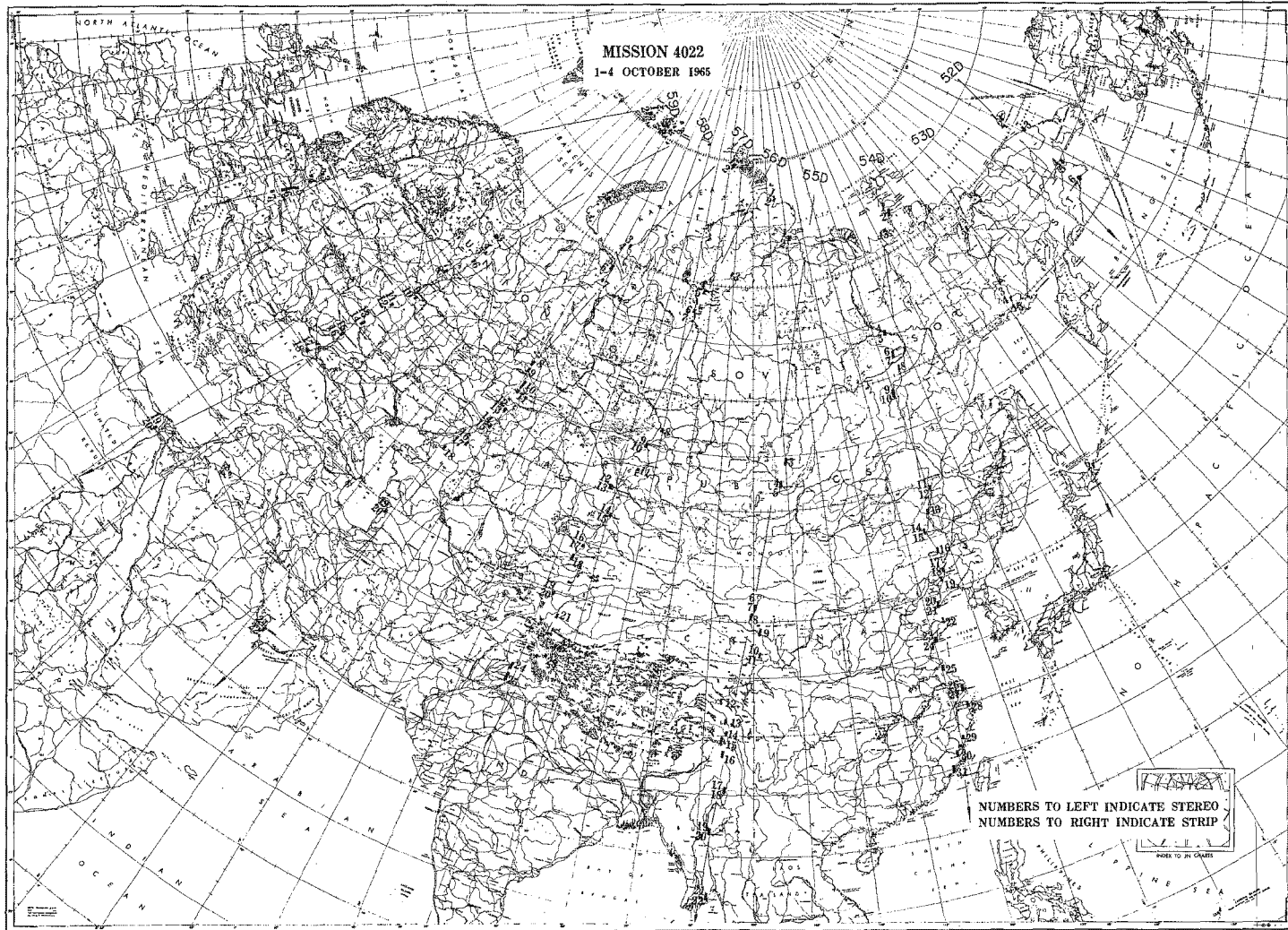
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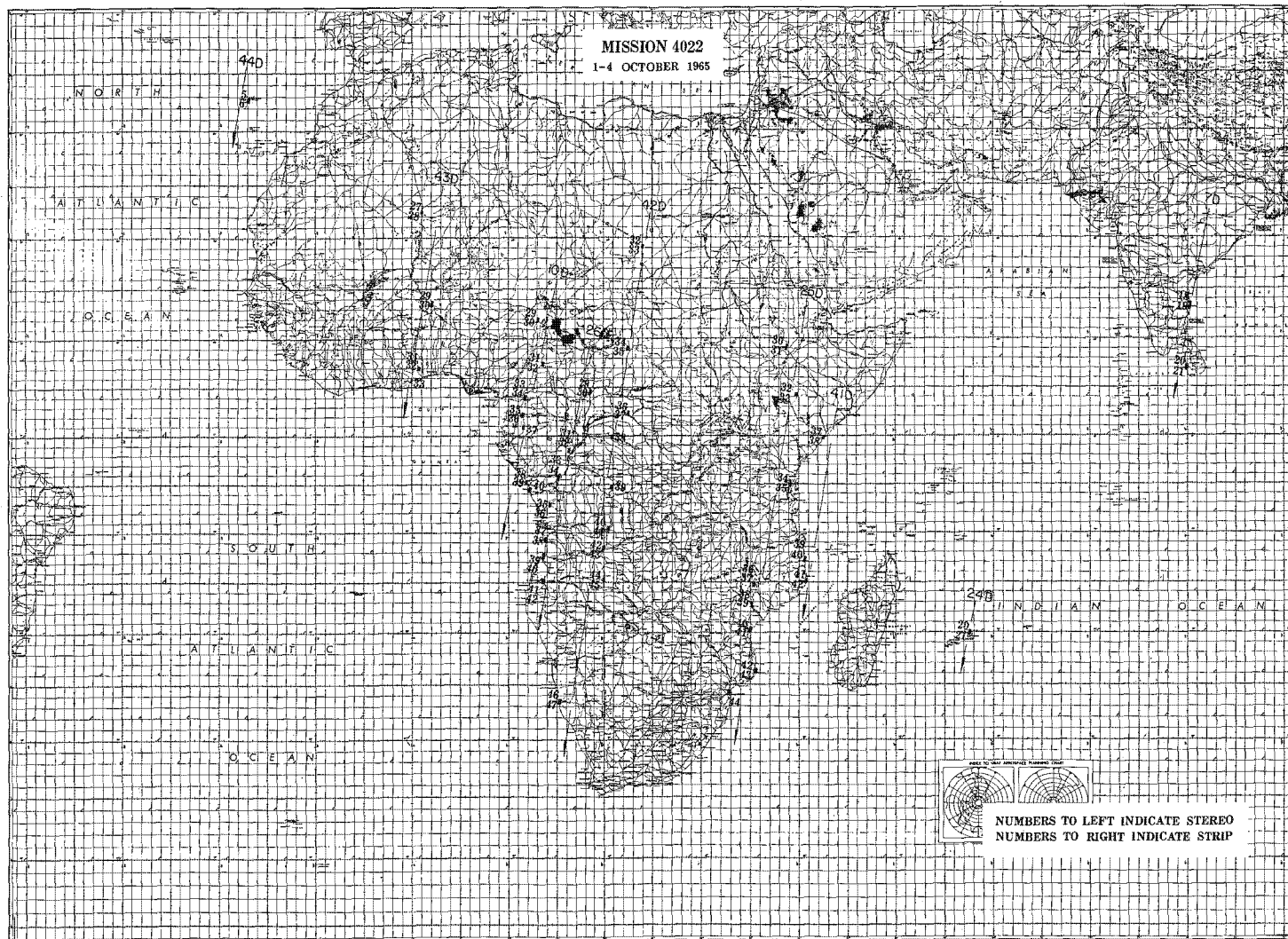
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ILLUSTRATION 4



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ILLUSTRATION 5

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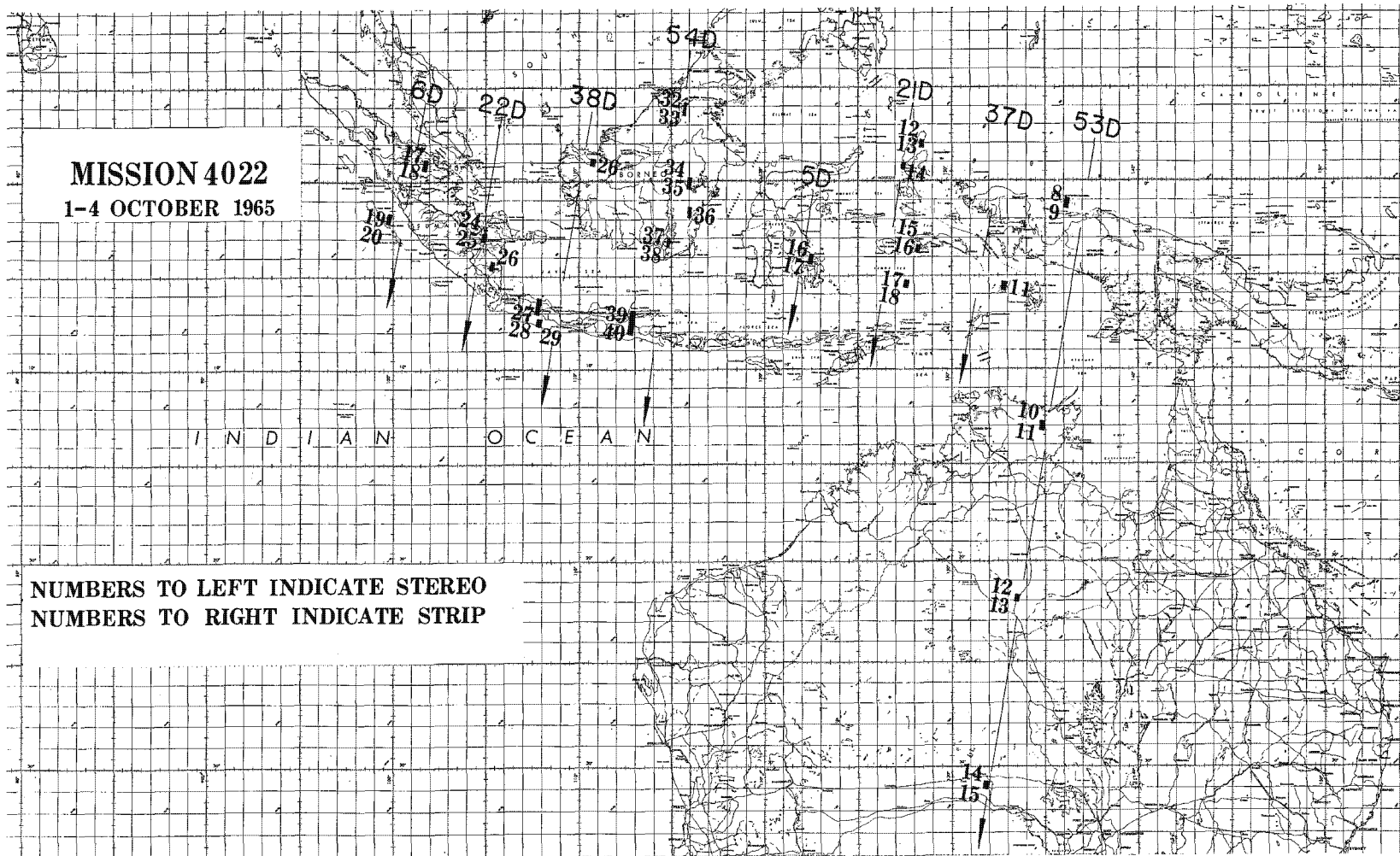


ILLUSTRATION 7

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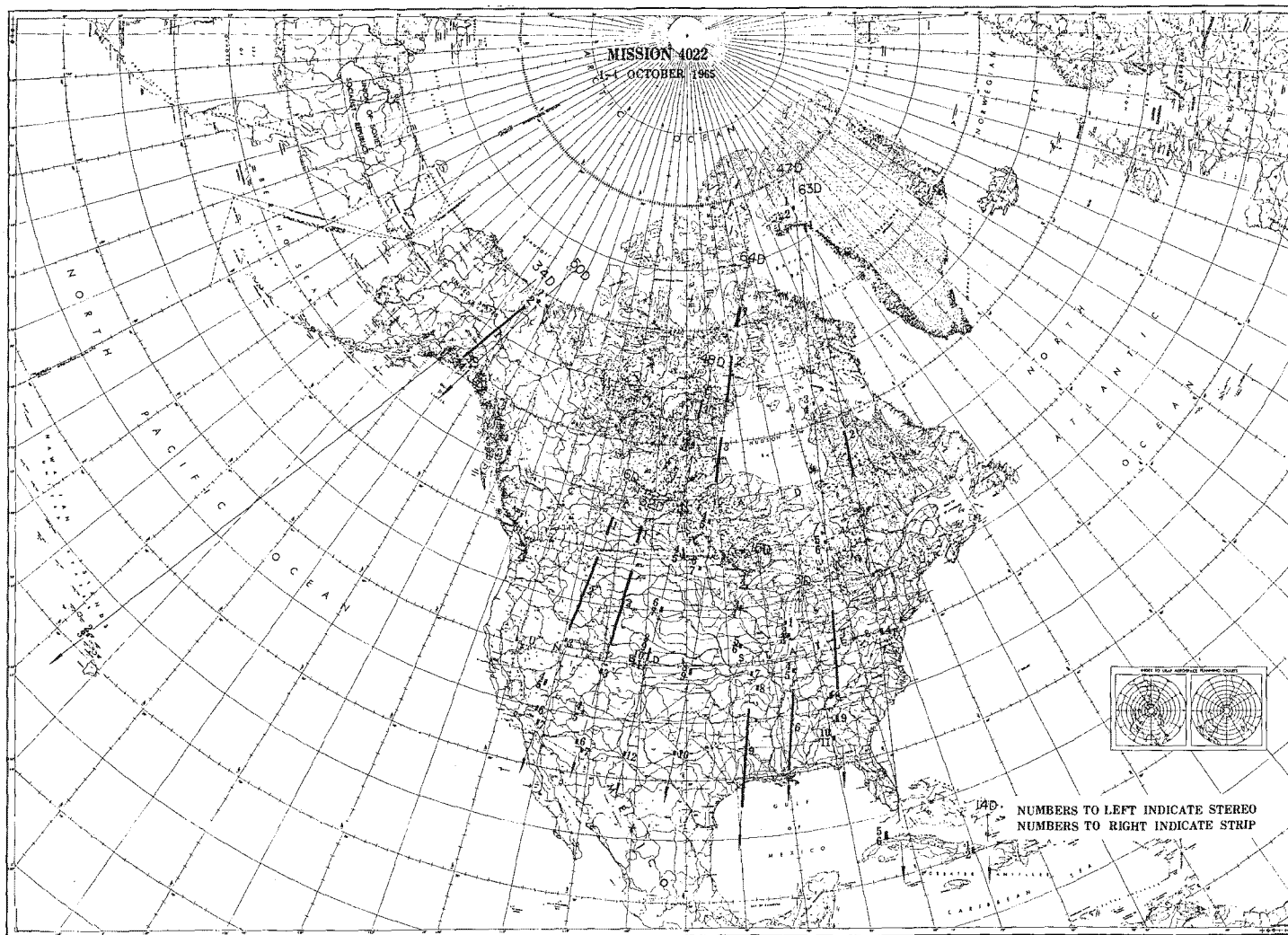


ILLUSTRATION 8

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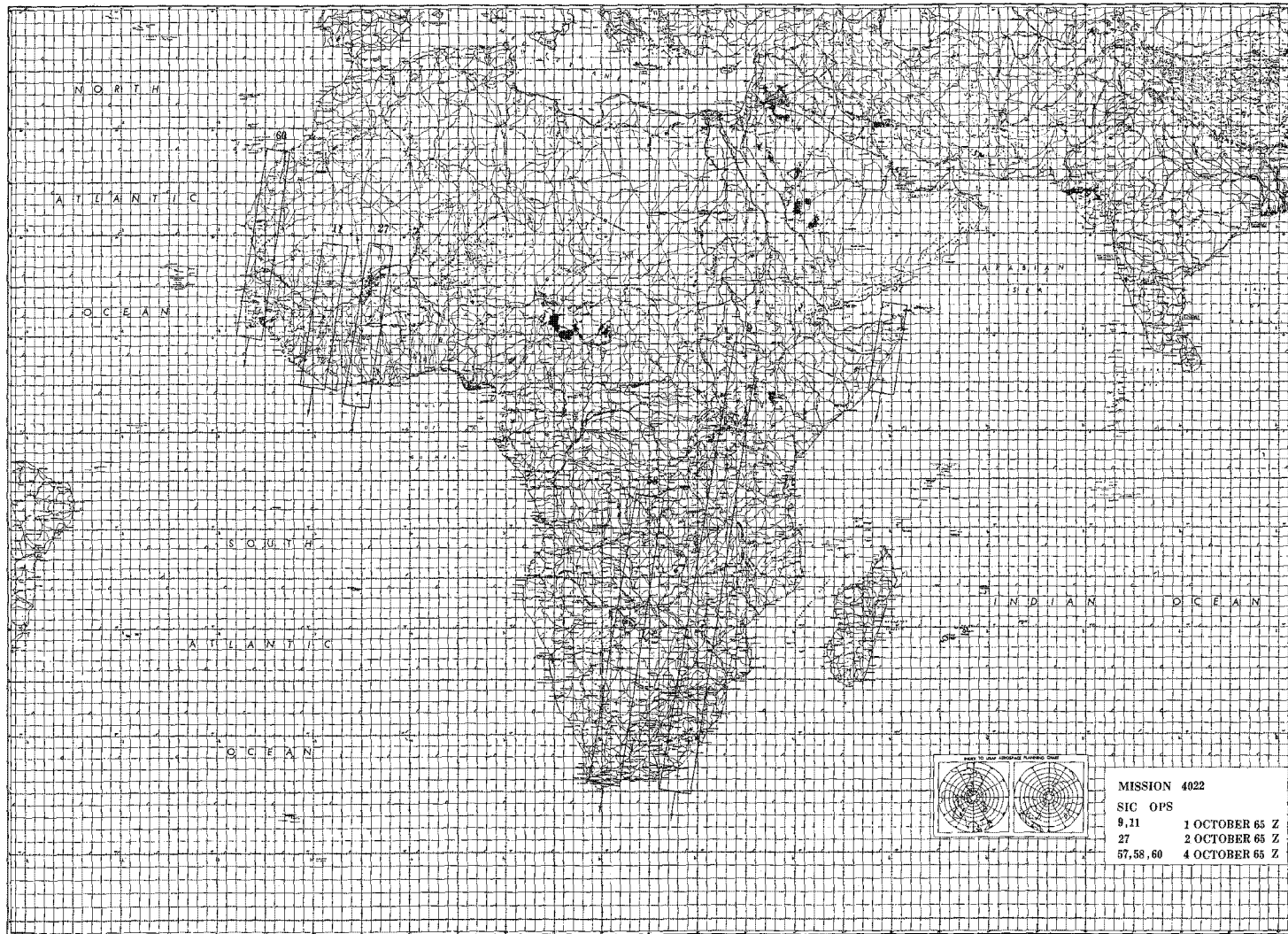
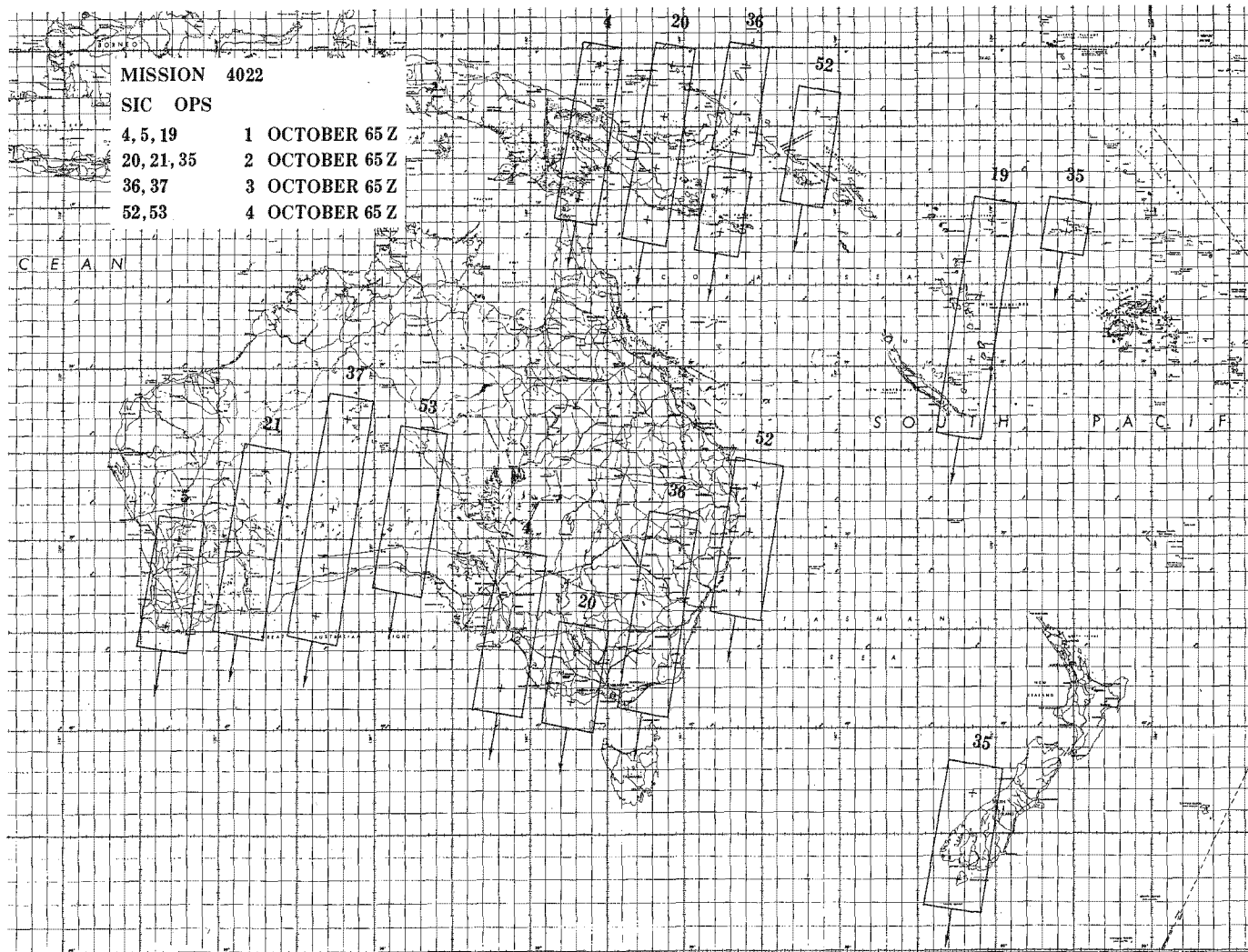


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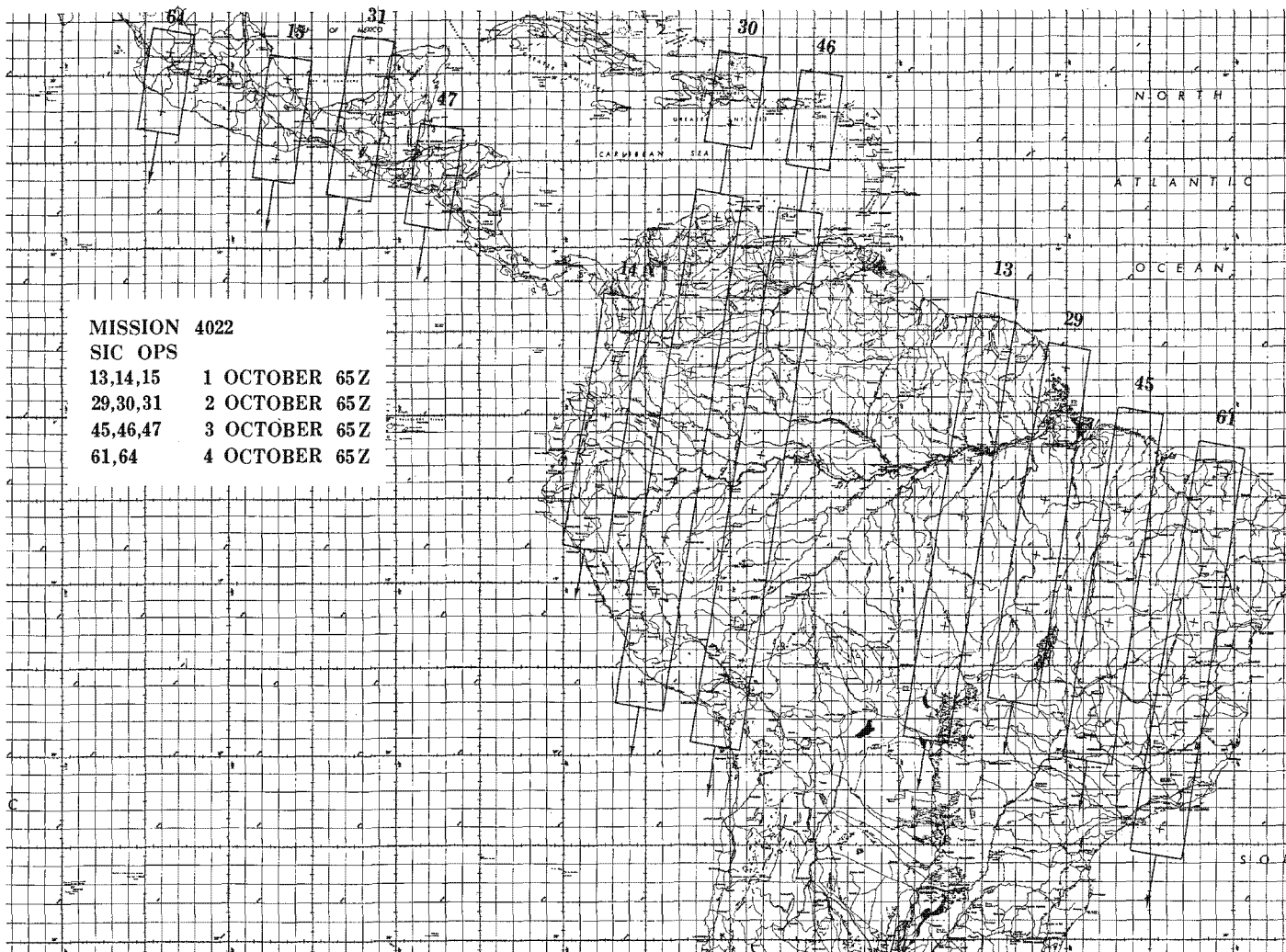
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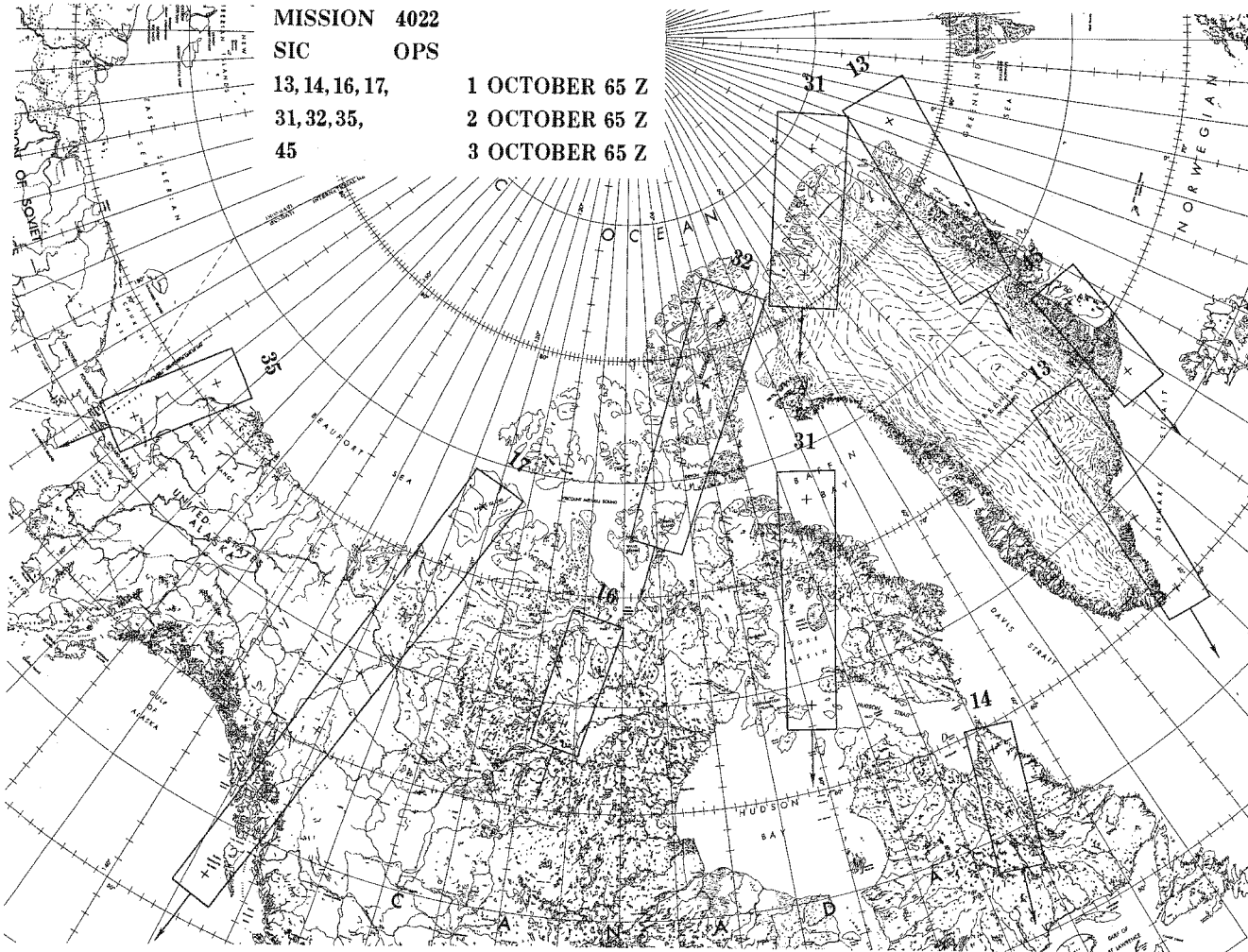
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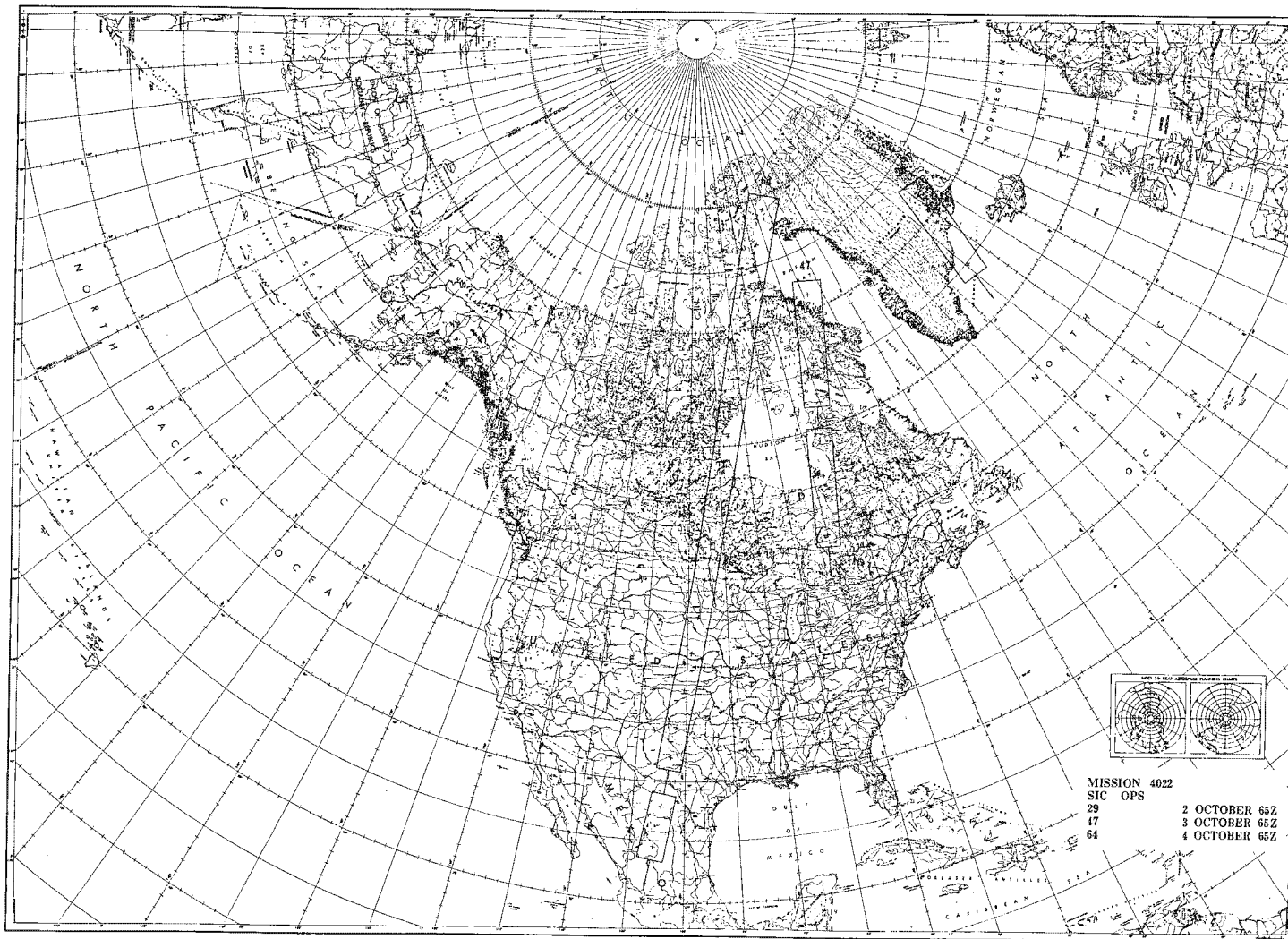
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ILLUSTRATION 12



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ILLUSTRATION 13

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ILLUSTRATION 14

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PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65SECTION II  
FLIGHT PROGRAM

## Performance of the Command System

1. For Mission 4022 all commands, except for those in conjunction with R&D experiments, for camera payload operations were generated by the Mission Profile Generation Program (TMPGP) and programmed through the GMINT and GCOMMAND modules. The following table summarizes the command messages loaded, the rev and tracking station for loading, the operating rev (s) and the results.

## COMMAND LOAD SUMMARY

<u>MESSAGES</u>	<u>LOAD REV &amp; STATION</u>	<u>OPERATING REV(S)</u>	<u>STEREO PAIRS</u>	<u>STRIPS</u>	<u>FRAMED STEREO PAIRS</u>	<u>LATERAL PAIRS</u>
103	4/POGO	4	3	1		
		5	3	3	4	
105	6/BOSS	6	4	6	2	1
		7	6	3	3	
108	8/COOK	8	3	5	2	1
		9	6	8	6	
109	10/HULA	10	10	12	4	
		11	3	1	1	
		13	2	1		
110	14/POGO	14	1			
		15	3	3		
112	16/POGO	16		4	1	
		17	1			
113	19/POGO	19	1			
		20	3	2	1	
		21	4	4	3	
114	22/BOSS	22	5	10	3	
		23	1	5	2	
		24	5	3	3	1
116	25/KODI	25	11	10	6	
		26	7	14	7	
		27	7	1		
		29	2			
117	30/POGO	31	1	2	1	
		32	1	4		
118	33/POGO	33	1			
		34		1	1	

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<u>MESSAGES</u>	<u>LOAD REV &amp; STATION</u>	<u>OPERATING REV(S)</u>	<u>STEREO PAIRS</u>	<u>STRIPS</u>	<u>FRAMED STEREO PAIRS</u>	<u>LATERAL PAIRS</u>
121	36/POGO	36	3			1
		37	2	3	2	
122	38/POGO	38	4	11	5	
		39	4	8	1	
		40	5	3	2	
123	41/KODI	41	9	8	7	1
		42	12	11	6	
		43	11	5	2	1
		44	1		2	
125	46/POGO	47	5	3	1	
		48	3	3	1	
126	49/POGO	49	1			
		50	2			
127	52/POGO	52	2	1		1
		53	7	1		
128	54/BOSS	54	8	12	6	
		55	4	8	3	
		56	5	9	3	
		57	4	6	2	2
129	58/HULA	58	9	2		
130	59/HULA	59	1	2		
132	63/POGO	63		4	1	
		64	2	3	1	

2. Stellar/Index Camera (GSIC) operations independent of the primary camera were programed for mapping purposes. A total of 27 command messages were manually programed for independent GSIC operations on 59 revs to obtain 979 photographic frames in the mapping mode. A total of 792 GSIC photographic frames were obtained during primary camera payload operations. For evaluation, see NPIC comments in Section I.

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PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65SECTION III  
PHOTOGRAPHIC CHARACTERISTICS EVALUATION SUMMARY  
(SPPL Report No. 101-1-58)

The following observations result from analysis of the original negatives from Reconnaissance Satellite Mission 4022.

## A. Physical Degradations

## 1. Imaged

a. Two fine-lined, continuous minus-density streaks were visible paralleling the major axis, approximately 1 1/2" from the titled edge, throughout this Mission.

b. Dendritic fogging (static discharges) was observed intermittently along the titled edge of the following Revs: D06, D07, D13-D15, and D52-D56.

## 2. Superficial

a. A thin, continuous base scratch occurred, approximately 2 3/4" from the non-titled edge on all frames and revs throughout the Mission.

b. Two emulsion digs, approximately 1/8" in length, were found on Frame 003 of Rev D05.

c. A pre-processing splice (sonic) was noted 4" from the head of Frame 006, Rev D09.

d. Numerous pinholes were visible on Frame 014, Rev D08 and throughout all frames of Rev D07.

e. A 1/16" wide base scratch was observed extending the entire length at the center of Frame 004, Rev D16.

f. A few minor abrasions, scratches and pinholes were noted throughout this Mission.

## B. Film Format Characteristics

The titling, time tracks, and fiducial lines appeared clear and distinguishable throughout Mission 4022.

## C. Density

There were 648 density measurements accomplished on Mission 4022. The results are portrayed by range, average, and standard deviation ( $\sigma$ ) in the following table:

	<u>Range</u>	<u>Average</u>	<u>Standard Deviation (<math>\sigma</math>)</u>
Dmin	0.22 - 1.30	0.48	0.17
Dmax	0.34 - 2.66	1.35	0.42
$\bar{D}$	0.30 - 1.56	0.92	0.24
$\Delta D$	0.06 - 2.26	0.88	0.42

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PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65

## SECTION IV

## CAMERA SYSTEM OPERATIONS

## A. Primary Camera

## 1. Film Transport

Velocity plots by NPIC showed that this transport system was not as smooth as others. Although there are many deviations from the  $\pm 1\%$  film velocity error permitted by specification, the  $3\sigma$  value lies well within the specified limits. The looper empty and start of refill did not result in velocity variations as violent as experienced previously due to excellent damping. The start up and stopping transients were nominal.

## 2. Yaw Slits

Slits were clean and produced usable imagery for image motion analysis.

3. At liftoff, 2683 feet were available for on orbit use. The load in the supply prior to going to the launch pad was 2955 feet. The film was consumed as follows:

Pad Checkout	272 feet
Orbital Health Check	35 feet
Operational Footage	1223 feet
R&D/ CORN	376 feet
Runout	649 feet
Processing Blanks	<u>400 feet</u>
Total	2955 feet

## 4. Processing

Approximate percentages of the use of the three processing levels:

Full	72%
Intermediate	20%
Primary	8%

Due to misidentification of looper marks for frame lines, some frames received two levels of processing. FM-23 will have interframe marks on the film and will eliminate this difficulty. Only one frame could be identified as underprocessed.

## 5. Exposure

Optimum exposure occurs about 0.50 density level for Dmin for this mission. Many frames have Dmin values below this level but are still within the half photographic stop below  $D = 0.50$  deemed as acceptable exposure. About 12% of the frames were below this minimum acceptable level. This is due in a large part to the low sun angles encountered this time of year and the terrain still covered with dark green vegetation. This filter-lens combination has poor green transmission.

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## 6. Resolution

Resolution was the best achieved to date. CORN target readouts indicated levels of about 1.8 feet. These readings still must be adjusted for their position in the field of the lens and the higher than 2:1 contrast of the CORN target as it appears at the aperture of the orbiting lens. This resolution was totally unexpected if judgement is based on the tri-bar readings of resolution obtained during acceptance testing and MAB validation testing. These levels were about 125 lpm at 2:1 contrast. Besides this consideration, the lens was more astigmatic than any other flight model accepted. One theory is based in the lack of adequate stiffness in the substrate as compared to later stiffer mirrors and the relieving of stress in the zero-G environment. Focus sensor operations indicated BPF as being removed from the preflight setting by about .002". Platen movements on orbit were accomplished moving the platen in two separate movements of about .001" each. No real improvement in resolution as a result of these movements is apparent in the preliminary subjective visual analysis nor the CORN target readings. Edge Trace data has not yet been compiled at this writing. See NPIC comments on resolution.

## B. Stellar/Index Camera

S/I No. 18 was flown. No shutter or transport anomalies due to mechanical operation were noted. However, four programmed frames about the middle of the mission did not occur. No explanation can be offered at this time. A new stellar baffle installed for this mission aided in obtaining a greater stellar count than on any other mission. See the NPIC comments on the results of the S/I performance and the resulting utility for mensuration.

## C. Experimental Operations

The following Secondary Flight Objectives were accomplished:

1. SFO 120: Remote Focus Record - with the door open, no film transport, the focus sensor is turned on. Resultant readings were obtained on nineteen revs. These resulted in the platen being moved about .001" on Rev 22 and the same amount on Rev 35. The evaluation of the resulting image acuity has not been completed but did not result in a degradation as a result of the platen movements.
2. SFO 109: Thermal Pulse Degradation of Optics - with the door open for varying periods of 180 seconds to 800 seconds, strips are photographed at the termination of these door open periods. Edge Trace evaluations have not been completed. Subjective judgements indicate little noticeable difference in image quality even at the end of the longest period.
3. SFO 137: Film Set Elimination Experiment - two 1.9 seconds of film transport operation separated by 23 minutes before the processing blank were programmed on four revs. Film velocity plots of these frames have not yet been completed.
4. SFO 138: Slit Change Within Stereo Pair - this was accomplished on Rev 34 between 002 and 003. Subjective evaluations will be completed by the payload contractor. These evaluations should indicate to some extent the tradeoffs between reducing smear at the expense of low light detail rendition.

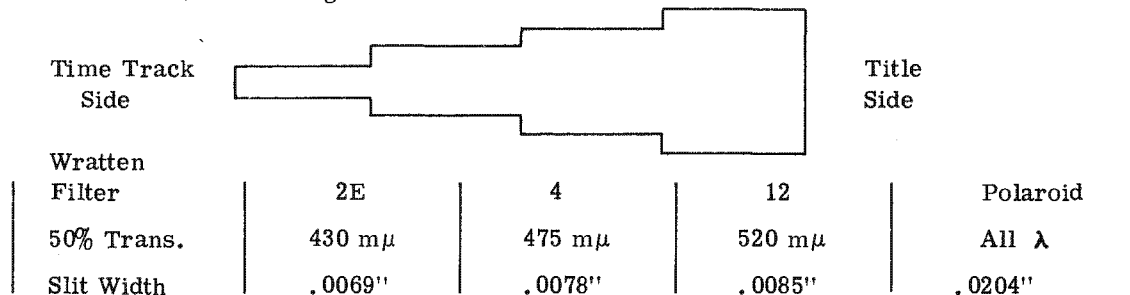
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5. SFO 139: Move Platen .002" Between Frames - this was accomplished on Rev 63. Platen was moved out on frame 003 and back to BPF on 004. Edge trace data is undergoing processing. Slit No. 3 with the glass blocks either side of the center of the slit was used. Subjective evaluation shows little change of image quality even though the BPF is increased about .002" by these glass blocks.

6. SFO 141: Test Slit Spectral Filtration - the test slit Position A on the slit aperture plate had four filters across its aperture. This slit was used on six revs. Tone traces were taken from good imagery on four revs. Edge traces were taken where well-defined images existed. This data will be reduced at a later date. The configuration of the slit was as shown below:



CORN Targets were obtained on:

Rev 15,	Frame 007	Booneville, Missouri
Rev 16	Frame 003	Wells, Nevada
Rev 31	Frame 004/005	Olney, Illinois
Rev 31	Frame 006	Mathison, Mississippi
Rev 32	Frame 003	Green River, Utah
Rev 32	Frame 006	Eloy, Arizona
Rev 47	Frame 009	Norcross, Georgia
Rev 63	Frame 004	Mechanicsburg, Pennsylvania
Rev 64	Frame 008/009	Hill City, Kansas
Rev 64	Frame 010	Big Springs, Texas

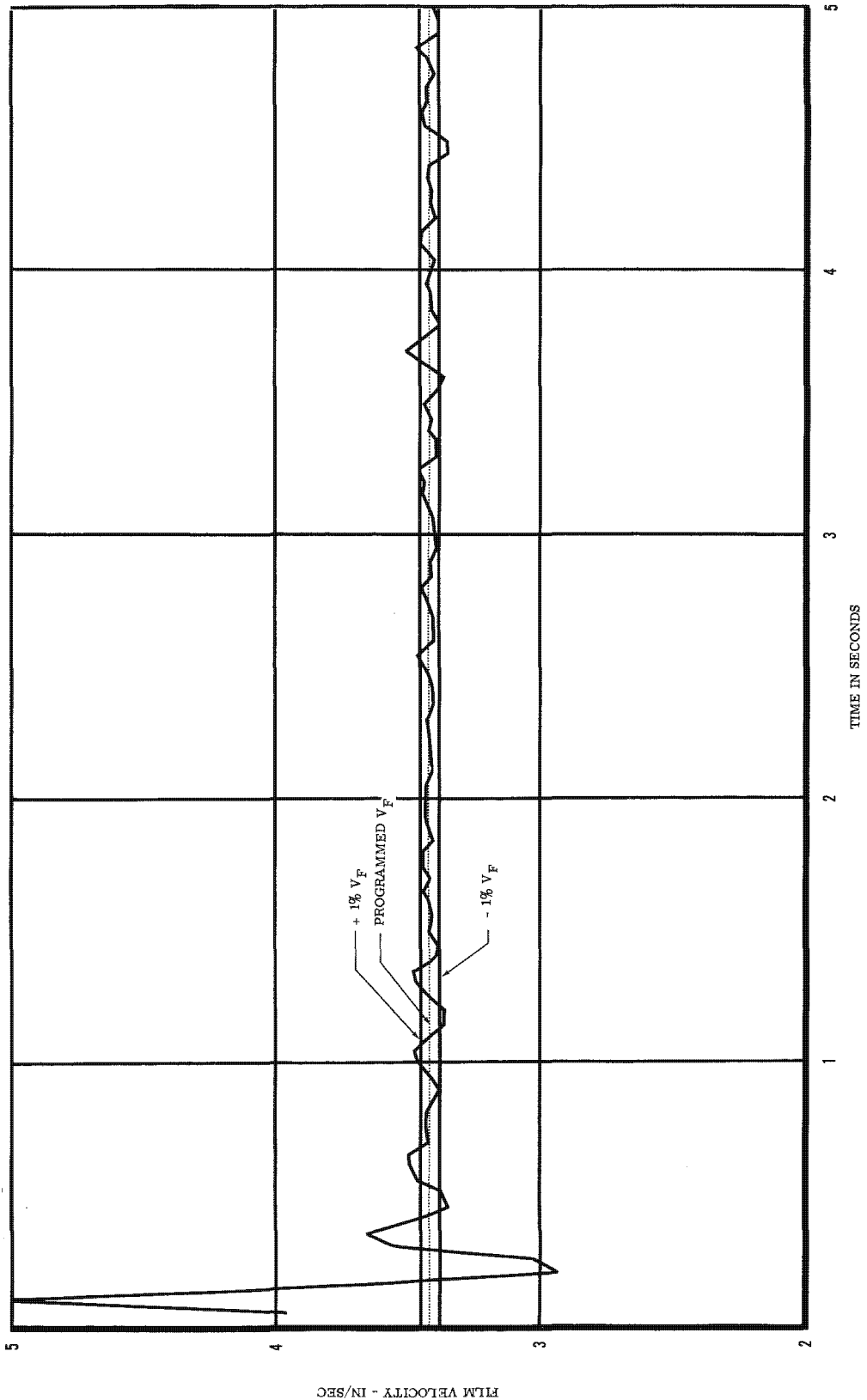
Blackbird aircraft was flown on 1, 3, and 4 October. The coverage obtained on 1 October was useful for comparison with Rev 15, frame 007, at Booneville, Missouri. On 3 October there was a camera mount and camera malfunction. On 4 October the coverage was two miles west of the CORN targets. Blackbird used KA-2, 12" focal length, and T-11, 6" focal length cameras at 16,500 feet and 8,000 feet altitudes. Bluebird aircraft was flown on 2 October. Comparison cover with Rev 32, frame 006, was obtained. Bluebird used a 12 inch focal length, KA-53 camera at 18,500 feet altitude.



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FILM VELOCITY VS TIME  
REV D10 FRAME 029



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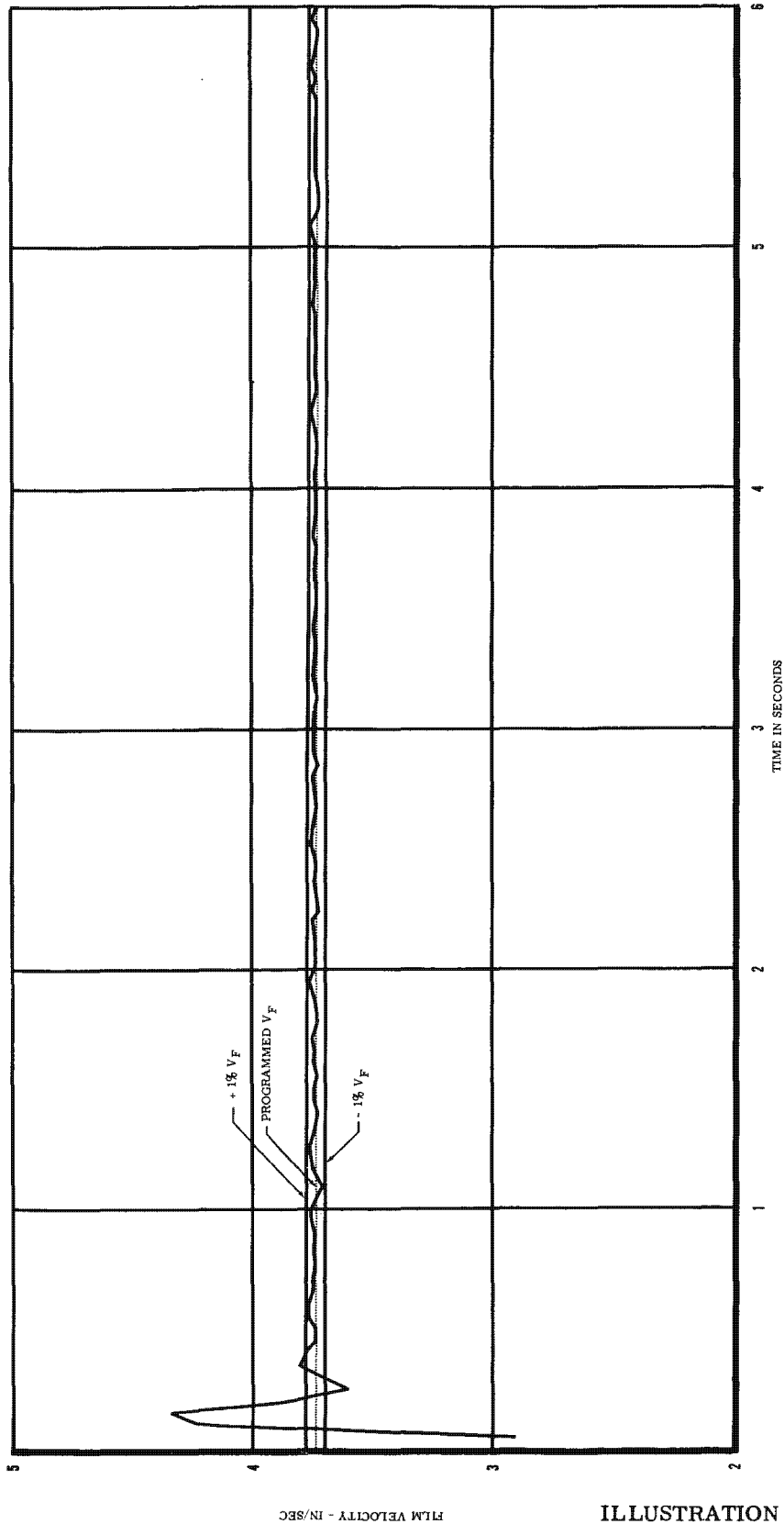
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ILLUSTRATION 15

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FILM VELOCITY VS TIME  
REV 15  
FRAME 007



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ILLUSTRATION 16

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FILM VELOCITY VS TIME  
REV D52  
PROCESSING BLANK

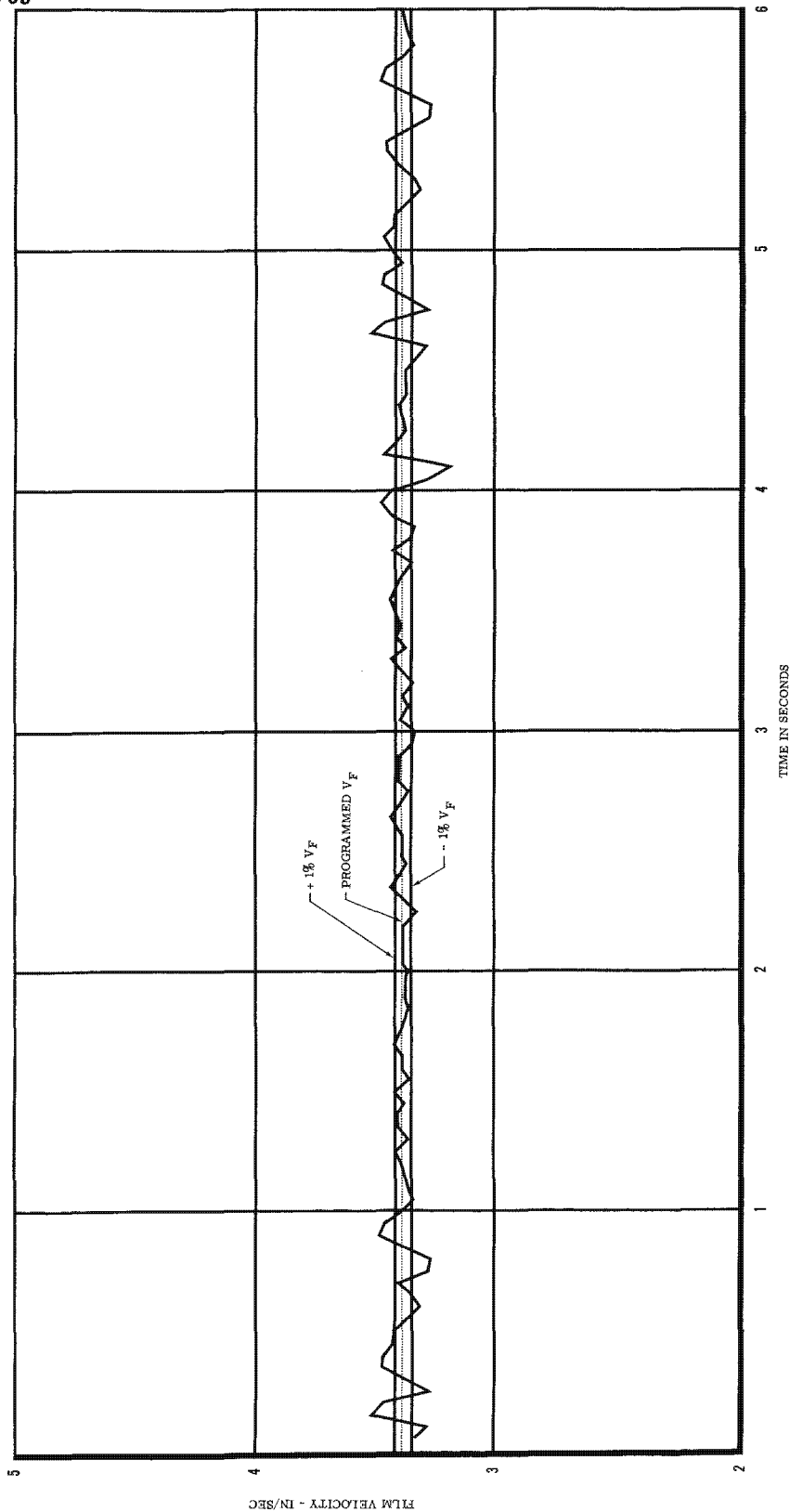


ILLUSTRATION 17

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PHOTOGRAPH NO. 1  
SFO 109-1

CORN Target at Wells, Nevada  
Door open 301 seconds, focus power on

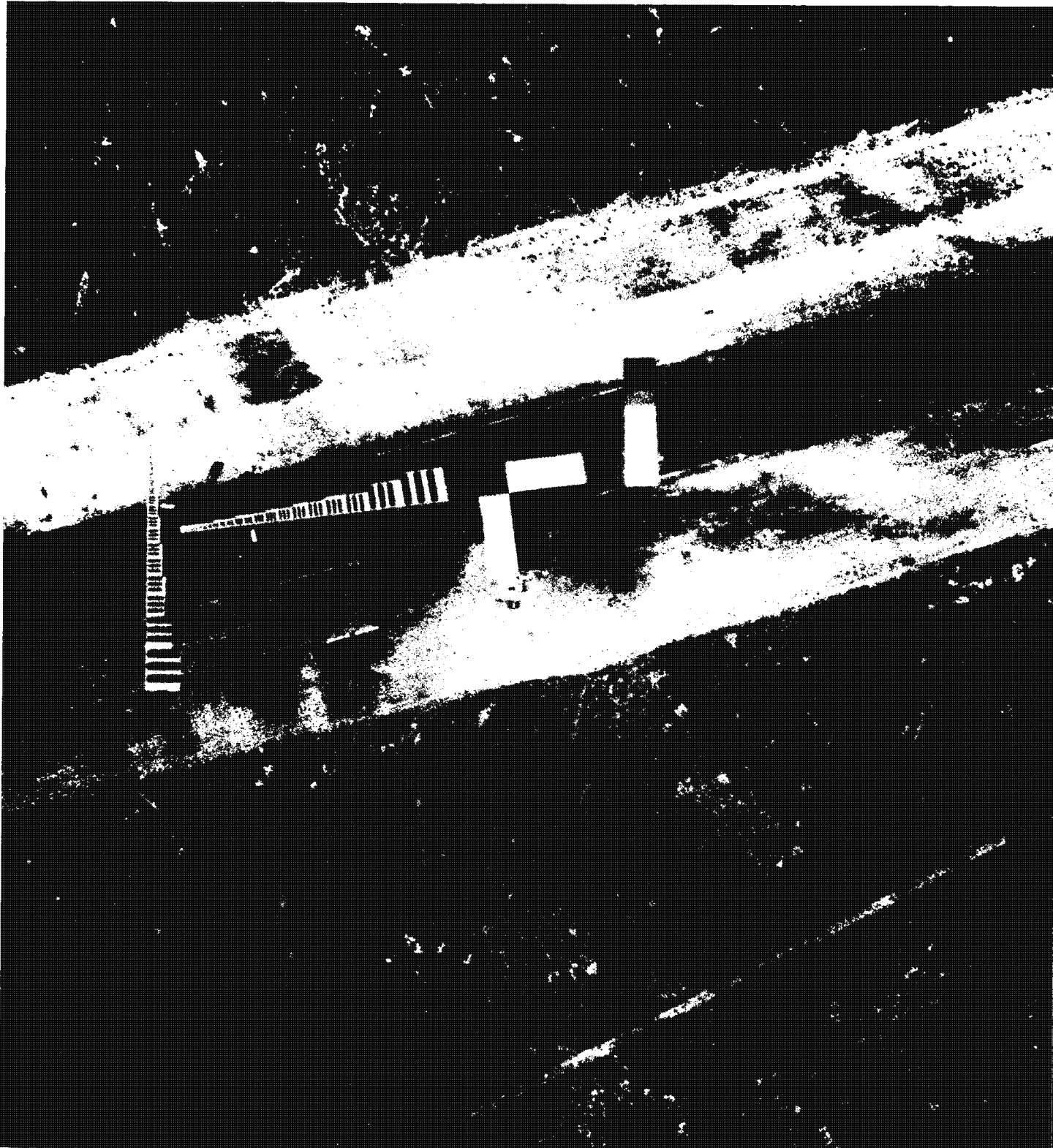
Rev D16                      Frame 003

Altitude: 85.7 NM  
Scale: 1:83,200

Obliquity: -12.3°  
Strip

40X

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PHOTOGRAPH 1  
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PHOTOGRAPH NO. 2  
SFO 109-2

CORN Target at Green River, Utah  
Door open 500 seconds, focus power on

Rev D32

Frame 003

Altitude: 84.8 NM  
Scale: 1:80,500

Obliquity: +0.4°  
Strip

40X

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PHOTOGRAPH 2

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PHOTOGRAPHS NO. 3 & NO. 4

CORN Target at Eloy, Arizona

PHOTOGRAPH NO. 3

Obtained by Bluebird aircraft  
using a KA-53, 12" focal length  
camera from 18,500 feet

20X

PHOTOGRAPH NO. 4

SFO 109-3

Door open 592 seconds, focus  
power on

Rev D32

Frame 006

Altitude: 84.4 NM  
Scale: 1:80,000

Obliquity: -1.9°  
Strip

40X

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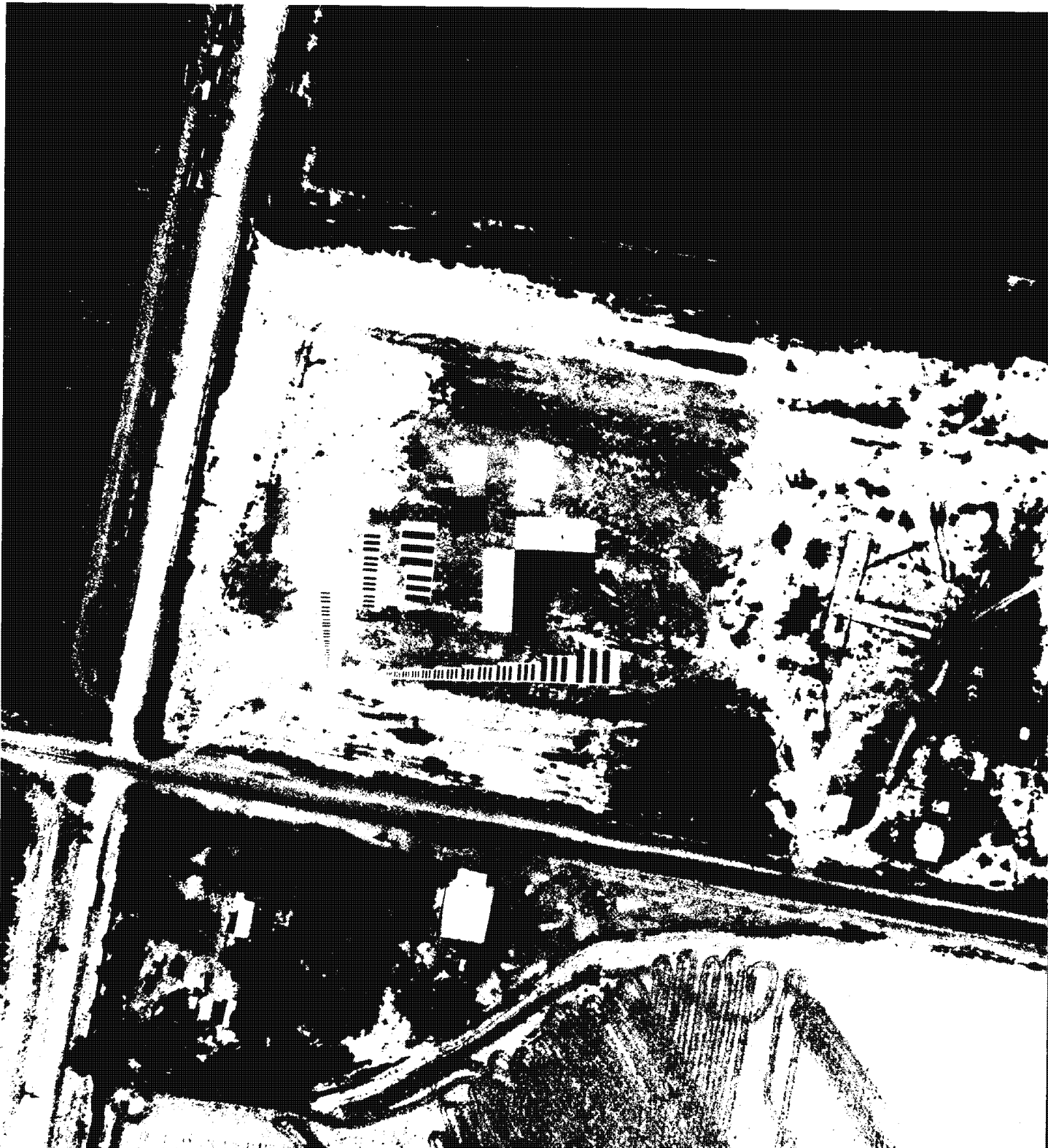


PHOTOGRAPH 3

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PHOTOGRAPH 4

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PHOTOGRAPH NO. 5  
SFO 109-4

CORN Target at Lowry AFB, Colorado  
Door open 701 seconds, focus power on

Rev D48      Frame 011

Altitude: 84.4 NM      Obliquity: + 6.0°  
Scale: 1:80,500      Strip

40X

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PHOTOGRAPH 5

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PHOTOGRAPHS NO. 6 & NO. 7

CORN Target at Booneville, Missouri

PHOTOGRAPH NO. 6

Blackbird photography obtained by KA-2/  
12" focal length camera from altitude of  
16,500 feet.

PHOTOGRAPH NO. 7

Rev D15            Frame 007

Altitude: 85.4 NM            Obliquity: -1.1°  
Scale:    1 :81,000            Strip

40X

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PHOTOGRAPH 6

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PHOTOGRAPH 7

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PHOTOGRAPH NO. 8  
CORN Target at Olney, Illinois

Rev D31            Frame 005

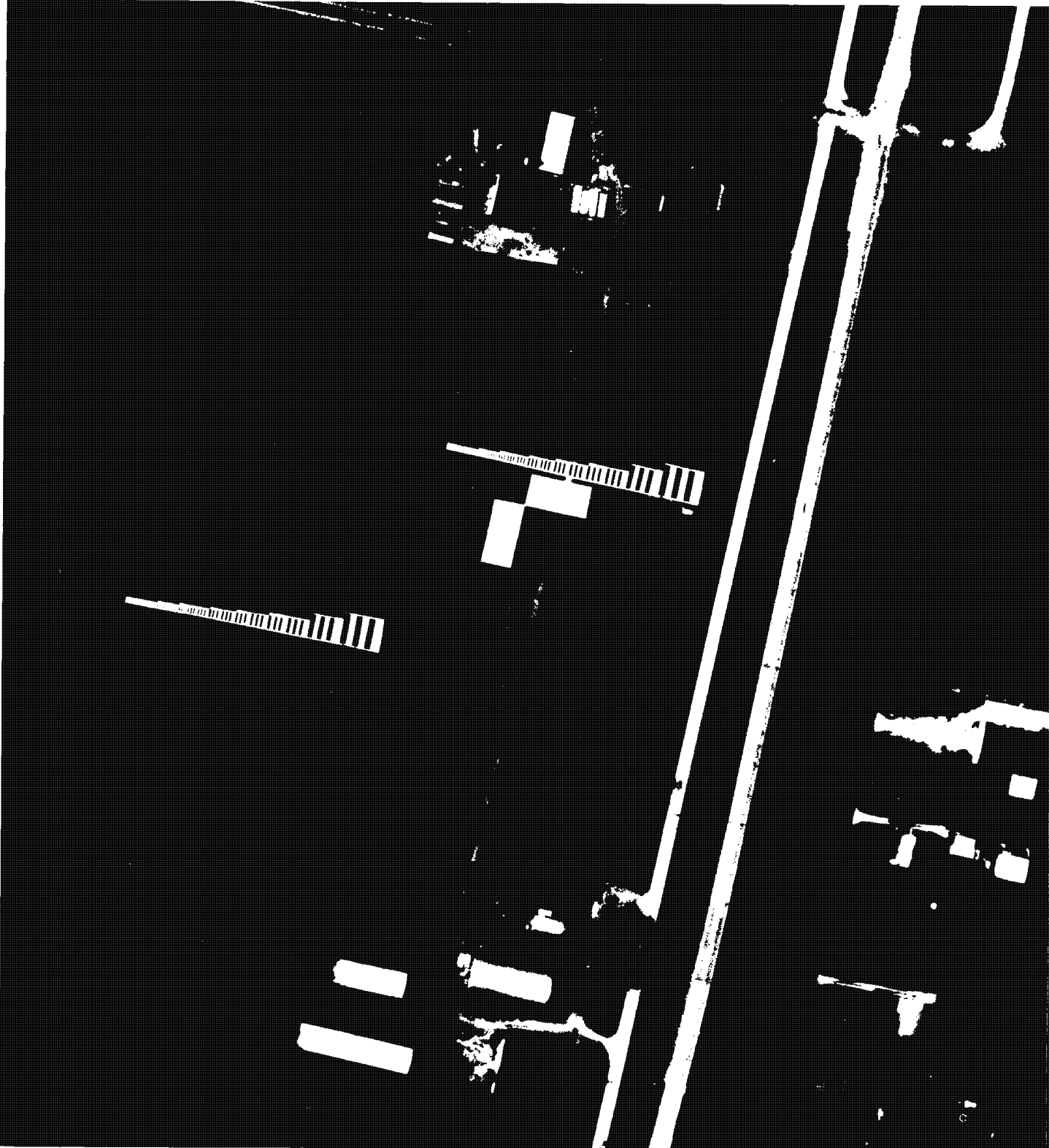
Altitude: 84.8 NM            Obliquity: +0.2°  
Scale: 1:83,200                Stereo

40X



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PHOTOGRAPH 8

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PHOTOGRAPH NO. 9

CORN Target at Mathison, Mississippi

Rev D31

Frame 006

Altitude: 84.4 NM  
Scale: 1:80,000

Obliquity: +0.9°  
Strip

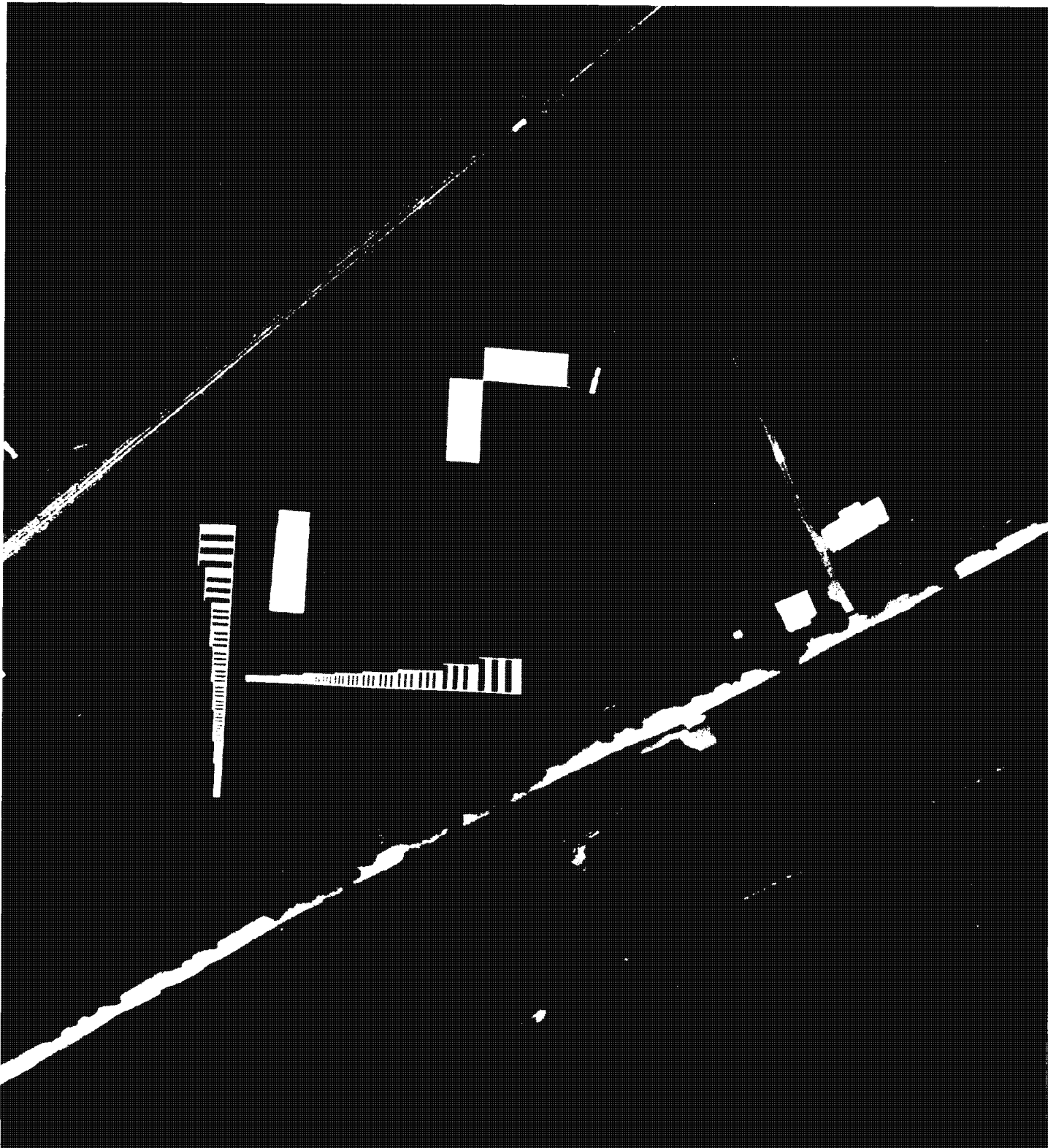
40X

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PHOTOGRAPH 9

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PHOTOGRAPH NO. 10

CORN Target at Norcross, Georgia

Rev D47

Frame 009

Altitude: 84.1 NM  
Scale: 1:79,800

Obliquity: -2.6°  
Strip

40X

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PHOTOGRAPH 10  
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PHOTOGRAPH NO. 11

CORN Target at Hill City, Kansas

This CORN target was located in the center of the format using the glass blocks test slit. Focal plane is at best plane of focus (BPF).

Rev D64

Frame 008

Altitude: 84.1 NM  
Scale: 1:83,000

Obliquity: +5.8°  
Stereo

40X

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PHOTOGRAPH 11

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PHOTOGRAPH NO. 12

CORN Target at Hill City, Kansas

This CORN target was located in glass block adjacent to center block (time track side) using glass blocks test slit. Focal plane is at BPF + .0025 in.

Rev D64

Frame 009

Altitude: 84.0 NM

Obliquity: +5.8°

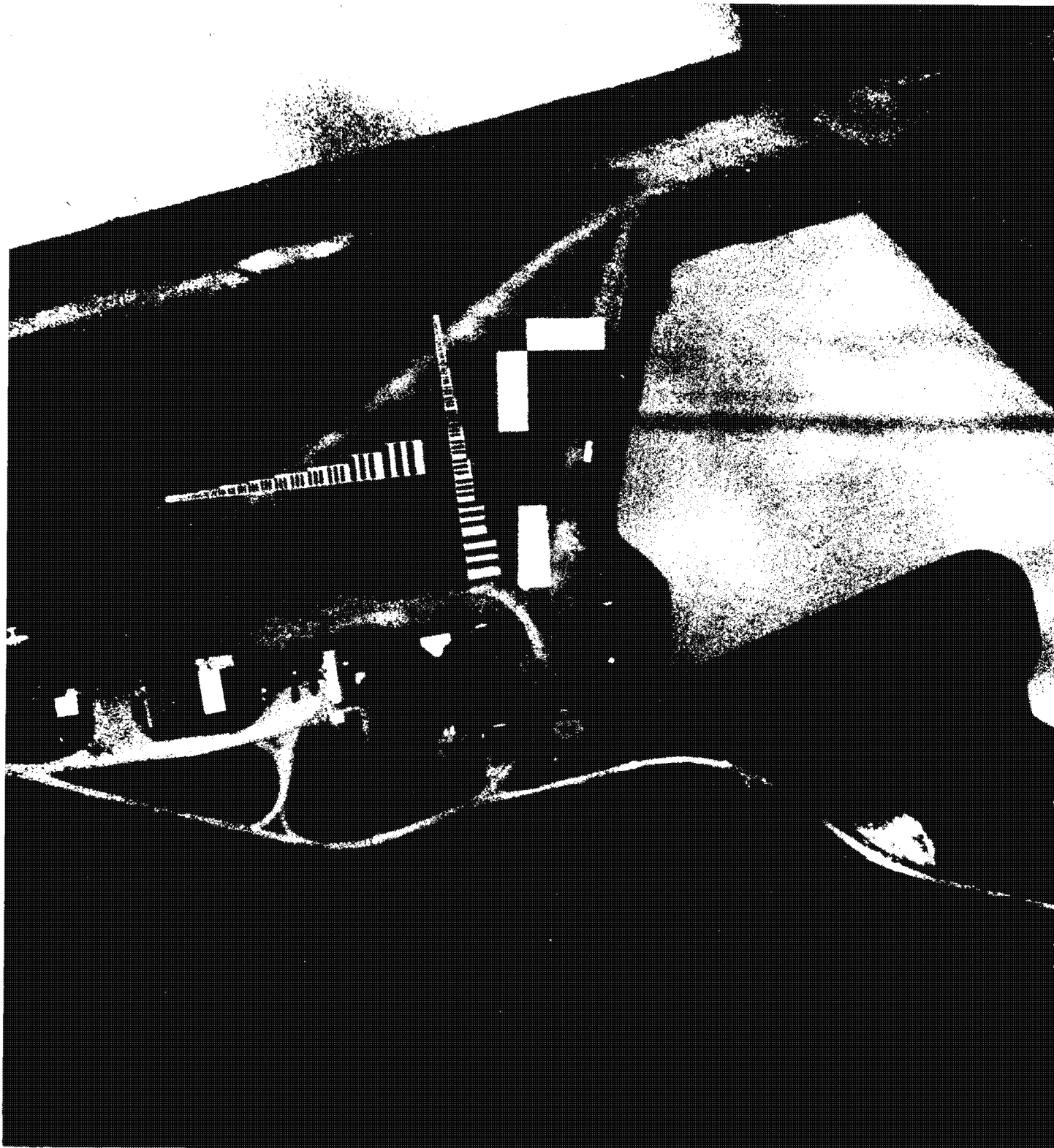
Scale: 1:82,900

Stereo

40X



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PHOTOGRAPH 12  
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PHOTOGRAPH NO. 13

CORN Target near Big Springs, Texas

This CORN Target was located two glass blocks  
from center block (time track side) using glass  
blocks test slit. Focal plane is at BPF + .002 in.

Rev D64

Frame 010

Altitude: 84.0 NM

Obliquity: +5.8°

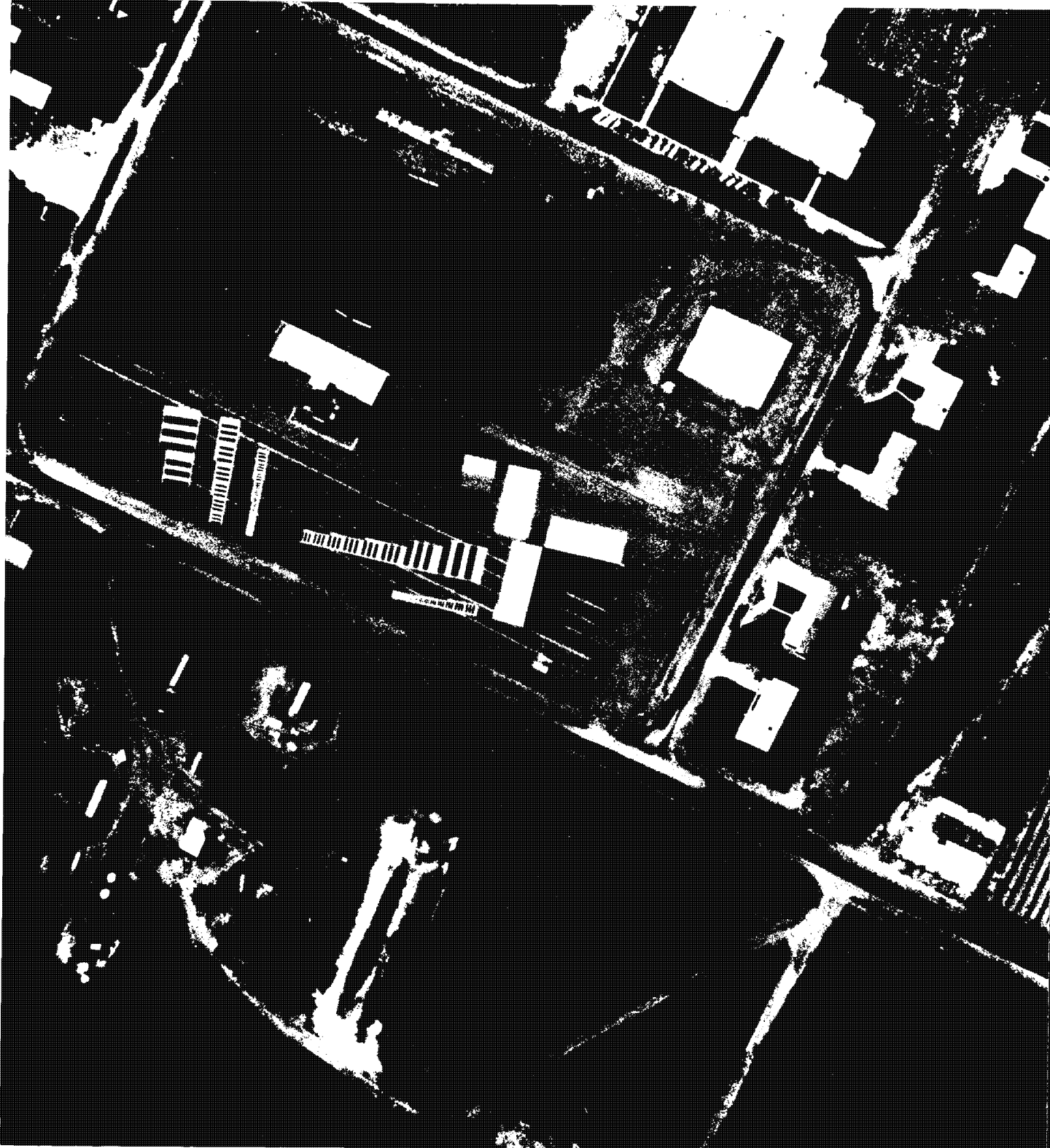
Scale: 1:80,000

Strip

40X

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PHOTOGRAPH 13  
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## SECTION V

## VEHICLE POSITIONING

## 1. Ephemerides

## a. Generation of Best Fit Ephemeris

The Best Fit Ephemeris was generated by reprocessing the tracking data in approximately one-day spans. Resulting range residuals were generally well under 1000 feet.

The following chart shows the fit spans selected for each primary BFE. Included in the chart are the weighted R. M. S. residual of the fit (a measure of quality), and the derived drag factor ( $C_D A/2m$ ). Nodal crossing time comparisons are also shown for revs common to two fits, to indicate the degree of consistency between ephemerides.

## CHARACTERISTICS OF BEST FIT EPHEMERIDES

<u>FIT SPAN</u>	<u>PAYLOAD REVS*</u>	<u>RMS RESIDUAL</u>	<u>DRAG FACTOR (FT<sup>2</sup>/SLUG)</u>	<u>NODAL CROSSING TIME COMPARISON</u>
1-18	4-17	1.22	.1628	
17-29	18-27	1.03	.1769	.048 sec @ Rev 18
26-34	28-33	0.83	.1984	.034 sec @ Rev 28
32-49	34-49	1.20	.2003	.030 sec @ Rev 34
54-64	56-64	1.17	.1889	.013 sec @ Rev 56

\*Revs used for payload analysis runs in TDATA and TEAR routines. BFE was broken into shorter fit spans due to unusually high magnetic storm activity.

## 2. Map Match

Forty-nine revs of photography, composed of 813 frames, were available for analysis. To improve the reliability of the results and reduce the map error contribution to the overall system error, no bench marks were selected on maps with positional error greater than 2000 feet on the World Geodetic System. This criteria of selecting photography in areas of less than 8/10 cloud cover provided 1711 bench marks on 305 frames of photography. After editing, 1658 bench marks were used for making final analysis. This photography was located on 39 of the 49 revs taken.

## 3. Position and Attitude Determination

a. The TEAR Program was used for the Mission 4022 evaluation. The TEAR program utilizes data derived from comparing ground bench mark positions common to the map and the photography, as well as vehicle (exposure station) positions from either the operational command or best fit ephemeris to determine in-track and cross-track miss distances. This is accomplished by using the satellite position and tracing the ray corresponding to the photo bench mark (PBM) position on the film through the camera system to the ground. The difference between the ground coordinates of the traced ray and the PBM map

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coordinates is then resolved into in-track and cross-track miss distances. Thus the miss distances can be written

$$\sigma = f(A, E, C, M)$$

where A, E, C and M respectively refer to attitude, ephemeris, camera and map quantities. Therefore, using the programmed attitude and camera orientation, the command or best fit satellite position along with the PBM position on the film and the map, the miss distances can be calculated. These miss distances ( $\sigma_x$  and  $\sigma_y$ ) are determined for each PBM in the frame and represent the total system error, the components of which are attitude, ephemeris, camera sub-system and map errors. Using TEAR with the vehicle position derived from the command ephemeris, the ephemeris error will contain a contribution due to predicting the ephemeris beyond the orbit determination (tracking data) fit span.

While the miss distances are computed for each PBM, the ultimate goal is to determine the errors which caused these miss distances. Knowing the functional relation  $\sigma = f(\ )$  the difference equations

$$\Delta\sigma = \frac{\partial\sigma}{\partial A} \Delta A + \frac{\partial\sigma}{\partial E} \Delta E + \frac{\partial\sigma}{\partial C} \Delta C + \frac{\partial\sigma}{\partial M} \Delta M = \sigma$$

can be formed where the  $\Delta$  quantities represent the errors (actual value-programmed value) which are to be determined and the coefficients of these errors can be explicitly derived. Obviously  $\Delta\sigma = \sigma$  since the programmed miss distance is zero.

In a simplified form to explicitly show the attitude and instrument errors the equations for the down range and cross range miss distances can be expressed as follows

$$\sigma_x = \sigma_{x0} + \psi \tan \Omega + \theta \left(1 + \frac{\tan^2 \Sigma}{\cos^2 \Omega}\right) + \varphi \frac{\tan \Sigma \sin \Omega}{\cos^2 \Omega} + \Delta\Sigma \frac{1}{\cos^2 \Sigma \cos \Omega} \quad (1)$$

$$\sigma_y = \sigma_{y0} + \psi \frac{\tan \Sigma}{\cos \Omega} - \theta \frac{\tan \Sigma \sin \Omega}{\cos^2 \Omega} - \varphi \frac{1}{\cos^2 \Omega} \quad (2)$$

where

- h = Altitude above horizontal plane passing through the bench mark.
- $\sigma_x$  = In-track miss distance divided by h.
- $\sigma_y$  = Cross-track miss distance divided by h.
- $\sigma_{x0}$  = Mapping, timing and ephemeris error contribution to  $\sigma_x$  divided by h.
- $\sigma_{y0}$  = Mapping, timing and ephemeris error contribution to  $\sigma_y$  divided by h.
- $\psi$  = Vehicle yaw.
- $\theta$  = Vehicle pitch.
- $\Omega$  = Vehicle obliquity angle = vehicle roll angle + mirror crab angle.
- $\varphi$  = Vehicle roll error.

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These equations have been derived for an ordered body orientation of yaw, pitch then roll and small angle approximations have been used. In particular, the errors have been regarded as small so that for example

$$F(\Omega \pm \varphi) = F(\Omega)$$

$$F(\Sigma \pm \Delta\Sigma) = F(\Sigma)$$

Where  $F(\ )$  represent the trig functions. Also the mapping error includes errors in measuring the location of the PBM on both the map and the film.

Since the vehicle yaw and pitch are always programmed to be zero,  $\psi$  and  $\theta$  are the yaw and pitch errors. Also  $\Delta\Sigma$  is the mirror stereo position error since  $\Sigma$  is the calibrated mirror position. The error in the mirror crab position has been included in the roll error as a first approximation. Thus the computed miss distances  $\sigma_x$  and  $\sigma_y$  along with the programmed obliquity angle  $\Omega$ , and the calibrated mirror stereo angle  $\Sigma$  are known. For the purpose of this analysis the bench mark height will be considered as known when computing  $\sigma_x$  and  $\sigma_y$  and will be allowed as an error included in  $\sigma_{x0}$  and  $\sigma_{y0}$ . The task then is to determine the unknown errors  $\sigma_{x0}$ ,  $\sigma_{y0}$ ,  $\Delta\Sigma$ ,  $\varphi$ ,  $\theta$  and  $\psi$  which cause the miss distances  $\sigma_x$  and  $\sigma_y$ .

#### b. Single Frame Case

Due to the nature of the GAMBIT system the program was formulated so that the programmed knowns do not vary over the length of the frame. Also the unknown errors are constant for each frame. In reality only the camera orientation errors are likely to be constant over a frame. Since the attitude and ephemeris errors are slowly varying during a typical short frame the assumption that these errors are constant is reasonable. However, while the error in the map itself may be nearly constant over a short frame the errors in measuring the PBM position on both the film and the map may vary considerably. Hopefully, these measurement errors are somewhat random with a bias nearly equal to the probable map error. Since the film and map measurement errors are included in the map error the assumption that the map error is constant during the time of a frame is likely to be poor depending upon the randomness of the measurements. Since the map errors are not truly constant during the time of the frame, the miss distances calculated from the functional relation  $\sigma = f(\ )$  will not be the same for each bench mark in the frame.

Regarding map and ephemeris errors as a single error, equations 1 and 2 are two equations in six unknowns. Since this is an undetermined and consequently unsolvable set of equations valid for each bench mark in the frame, choice of several bench marks allows the set to be expanded. Choosing three or more bench marks per frame one would expect the resulting set of equations to be exactly or overdetermined

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and solvable by the least squares technique. Since the coefficients of the unknowns as well as the unknowns are considered constant for each bench mark in a frame, equations 1 and 2 may be written

$$\sigma_{xi} = \sigma_{xo} + \psi A + \theta B + \varphi C + \Delta \Sigma D \quad (1a)$$

$$\sigma_{yi} = \sigma_{yo} + \psi E + \theta F + \varphi G \quad (2a)$$

where the subscript i refers to the i th bench mark in a frame and the trig coefficients have been replaced for simplicity. The above equations are 2i equations in six unknowns. However, the right-hand side of the 2i equations is constant for all bench marks (all values of i) so that it is not possible to uniquely determine the six unknown errors. Thus choosing any two uncorrelated errors and assuming the other errors are zero, a solution is possible. However, neglecting four of the six errors will result, in general, in erroneous values for the two determined errors. It should be noted that choosing two uncorrelated errors and applying the least squares technique to the 2i (i > 1) equations the technique simply averages over all the bench marks.

Again any two uncorrelated errors may be determined; however, the determined errors will, in general, be erroneous. By picking three sets of two uncorrelated errors and making three runs with the program all six errors may be determined. However, all six will be erroneous since they were taken two at a time. Nevertheless since only two errors can be determined at a time, the TEAR program has been used to determine each set of errors independently of the other sets in order to determine approximate values for the errors.

Since the ephemeris error is slowly varying, neglecting this error and solving for attitude error will result in a near constant attitude error bias from frame to frame. Also for the same mirror position the error is generally the same so that mirror position error will also introduce a bias into the attitude error determination. However, in this case the bias will depend upon the position of the stereo mirror. Therefore, if the map error is also constant from frame to frame, neglecting all but the attitude errors will only bias the results and the relative attitude time history will be meaningful. The map error is considered constant over a frame even though this may be an erroneous assumption. Thus the program is used, on a frame basis, to determine two attitude errors ignoring map, ephemeris and camera errors. In other words, in the current usage of the program to determine attitude errors on a frame basis,  $\Delta \Sigma$ ,  $\sigma_{xo}$  and  $\sigma_{yo}$  have been regarded as being equal to zero. The results then, at best, will be an average and only the relative attitude errors from frame to frame will be meaningful. Realizing the above limitations, then consistent with the method of using the TEAR program namely neglecting mirror, map and ephemeris errors, equations 1 and 2 may be written

$$\sigma_x = \psi \tan \Omega + \theta \left(1 + \frac{\tan^2 \Sigma}{\cos^2 \Omega}\right) + \varphi \frac{\tan \Sigma \sin \Omega}{\cos^2 \Omega} \quad (3)$$

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$$\sigma_y = \psi \frac{\tan \Sigma}{\cos \Omega} - \theta \frac{\tan \Sigma \sin \Omega}{\cos^2 \Omega} - \phi \frac{1}{\cos^2 \Omega} \tag{4}$$

to show the attitude error dependence.

Examination of equations 3 and 4 show the three attitude errors of which only two can be determined on a frame basis. Proceeding blindly and solving for the three errors, two at a time will give erroneous results since each solution ignores one of the errors. However, there are certain special combinations of the mirror and obliquity angles which allow an accurate solution for at least one of the unknown errors. Therefore, certain special cases will be examined where when solving for two of the errors, the resulting solution for at least one of the two errors will be accurate. Of course, while equations 3 and 4 may be used to determine an accurate solution for one or more of the unknowns the accuracy of the determination is still subject to the assumption that  $\Delta \Sigma = \sigma_o = 0$ .

Case A Strip Vertical ( $\Sigma = \Delta \Sigma = \sigma_o = 0$ )

Considering first the case where  $\Sigma = 0$ , equations 3 and 4 reduce to

$$\sigma_x = \psi \tan \Omega + \theta \tag{5}$$

$$\sigma_y = -\phi \frac{1}{\cos^2 \Omega} \tag{6}$$

Thus the roll error may be determined.

Case B Zero Obliquity ( $\Omega = \Delta \Sigma = \sigma_o = 0$ )

For  $\Sigma \neq 0$  and at zero obliquity angle, equations 3 and 4 reduce to

$$\sigma_x = \theta (1 + \tan^2 \Sigma) \tag{7}$$

$$\sigma_y = \psi \tan \Sigma - \phi \tag{8}$$

Thus the pitch error can be determined.

Case C Nadir ( $\Sigma = \Omega = \Delta \Sigma = \sigma_o = 0$ )

At zero obliquity and stereo angles, the equations become

$$\sigma_x = \theta \tag{9}$$

$$\sigma_y = -\phi \tag{10}$$

so that both the roll and pitch errors can be determined.

c. Multiple Frame Case

Since the program formulation assumes the unknown errors to be constant over any solution interval, equations 1a and 2a may be written

$$\sigma_{xi}^j = \sigma_{xo}^j + \psi A^j + \theta B^j + \phi C^j + \Delta \Sigma D^j \tag{1b}$$

$$\sigma_{yi}^j = \sigma_{yo}^j + \psi E^j + \theta F^j + \phi G^j \tag{2b}$$



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where the superscript  $j$  refers to the  $j$ th frame in the interval of frames where the errors are to be determined. In general, the coefficients  $A^j - - G^j$  will be different for each frame since the programmed knowns generally change from frame to frame. The assumption that the ephemeris and mirror errors are constant is reasonable over a few frames since the ephemeris error is slowly varying and the mirror is a mechanical system for which the error in position is approximately the same each time the mirror is in the same position. While it is likely that the map errors over several frames will vary even more than previously discussed, there is some probability that the measurement errors are somewhat random with a bias equal to the probable map error. Most critical is the assumption of constant attitude errors since an accurate time history (frame to frame) of these errors is desired. The assumption of constant attitude errors is poor due to attitude changes caused by the attitude control system instability, the vehicle roll maneuvers, changes in the mirror position and other mechanical devices. Considering these attitude errors constant over several frames results in the program determining an average error and the detailed time history is lost.

Equations 1b and 2b are  $2i \times j$  equations in six unknowns and solution via the least squares technique is indicated. Since for a particular frame ( $j$  constant) and for various values of  $i$  the equations are the same as section b with respect to the averaging property only  $i$  constant and various values of  $j$  will be considered in this section. In particular for  $j \geq 3$  the set of equations is exactly or overdetermined and solution is accomplished by the least squares technique. Two factors influence the solution for  $j \geq 3$ . First the solution may be ill conditioned due to the way the coefficients of equations 1 and 2 vary from frame to frame and due to correlation between the unknowns. Secondly, since the program formulation assumes the errors to be constant over the  $j \geq 3$  frames, any error which can be determined will be an average error over the  $j \geq 3$  frames.

To examine the correlation between the various unknown errors would require explicit expressions for the three map, two film measurement, three ephemeris, mirror stereo, mirror crab, three attitude and time errors. Thus including all the errors the correlation would be determined by examination of the determinant of a  $14 \times 14$  matrix. Even ignoring all errors except the attitude errors the examination of the  $3 \times 3$  matrix would involve at least the coefficients of equations 3 and 4 for two frames ( $j = 2$ ). This is an extremely difficult task and, in general, is best handled by using the program with several test cases. Also it should be noted that the program had not been used for  $j \geq 2$  and the only multiple frame case which had been used is the rather special case of stereo pairs.

The one case that can be examined in detail where only mirror stereo position and attitude errors are considered is the special case of stereo pairs. For stereo pairs the parameters corresponding to the first frame of the pair will be denoted with a plus superscript and the parameters corresponding to the second frame with a minus superscript. Letting  $\Sigma_c^+$  and  $\Sigma_c^-$  be the calibrated stereo mirror angles, then consider  $\Sigma$  to be an average value defined by

$$\Sigma = 1/2 (\Sigma_c^+ - \Sigma_c^-)$$

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Since the calibrated value is not the true mirror stereo position, the true values are then given by

$$\Sigma^+ = \Sigma + \Delta\Sigma^+$$

$$\Sigma^- = -\Sigma + \Delta\Sigma^-$$

Assuming that  $\Delta\Sigma^+$  and  $\Delta\Sigma^-$  are small so that

$$\tan \Sigma^+ = \tan \Sigma$$

$$\tan \Sigma^- = -\tan \Sigma$$

$$\cos^2 \Sigma^+ = \cos^2 \Sigma$$

$$\cos^2 \Sigma^- = \cos^2 \Sigma$$

and  $\psi$ ,  $\theta$ ,  $\varphi$  and  $\Omega$  are constant and equal for both frames of a stereo pair, then equations 1 and 2 for a stereo pair with  $\sigma_{x0} = \sigma_{y0} = 0$  may be written as

$$\frac{\sigma_x^+ + \sigma_x^-}{2} = \psi \tan \Omega + \theta \left(1 + \frac{\tan^2 \Sigma}{\cos^2 \Omega}\right) + \frac{\Delta\Sigma^+ + \Delta\Sigma^-}{2} \frac{1}{\cos^2 \Sigma \cos \Omega} \quad (11)$$

$$\frac{\sigma_y^+ - \sigma_y^-}{2} = \psi \frac{\tan \Sigma}{\cos \Omega} - \theta \frac{\tan \Sigma \sin \Omega}{\cos^2 \Omega} \quad (12)$$

$$\frac{\sigma_x^+ - \sigma_x^-}{2} = \varphi \frac{\tan \Sigma \sin \Omega}{\cos^2 \Omega} + \frac{\Delta\Sigma^+ - \Delta\Sigma^-}{2} \frac{1}{\cos^2 \Sigma \cos \Omega} \quad (13)$$

$$\frac{\sigma_y^+ + \sigma_y^-}{2} = -\varphi \frac{1}{\cos^2 \Omega} \quad (14)$$

where the  $\sigma$ 's refer to the same bench mark in each frame. In this analysis the equations for each frame were combined to represent a stereo pair. The TEAR program, for stereo pairs, combines the two sets of 2 equations per bench mark per frame into a single set which is twice as large. While this analysis differs from the program formulation, conclusions regarding permissible solutions due to the correlation of variables will be the same.

Examination of equations 11 - 14 shows four equations in five unknowns so that it is not possible to determine a unique solution for the unknown errors. However, if  $\Delta\Sigma^+$  and  $\Delta\Sigma^-$  are known or are zero then examination of the determinant of the 3 x 3 coefficient matrix shows that a unique solution can be determined. It should be noted that in the method of using the TEAR program it is assumed that

$\Delta\Sigma^+ = \Delta\Sigma^- = 0$  so that a unique solution, which is erroneous can be determined. The solution is erroneous since the effect of mirror stereo position is ignored whereas this error makes an important contribution to equations 11 and 13. Of course, neglecting  $\sigma_{x0}$  in equations 11 and  $\sigma_{y0}$  in equation 14 also

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decreases the validity of the solution.

For zero obliquity the above equations reduce further to

$$\sigma_x^+ + \sigma_x^- = 2 \theta (1 + \tan^2 \Sigma) + \frac{\Delta \Sigma^+ + \Delta \Sigma^-}{\cos^2 \Sigma} \frac{1}{\cos^2 \Sigma} \quad (15)$$

$$\sigma_y^+ - \sigma_y^- = 2 \psi \tan \Sigma \quad (16)$$

$$\sigma_x^+ - \sigma_x^- = (\Delta \Sigma^+ - \Delta \Sigma^-) \frac{1}{\cos^2 \Sigma} \quad (17)$$

$$\sigma_y^+ + \sigma_y^- = -2 \varphi \quad (18)$$

Therefore, if  $\Delta \Sigma^+$  and  $\Delta \Sigma^-$  are known, as before, a unique solution can be determined. Furthermore, even if the mirror stereo position errors are not known the yaw and roll errors can still be accurately determined. Again in the use of the program  $\Delta \Sigma^+$  and  $\Delta \Sigma^-$  are not known and closely coupled with the yaw, pitch and roll errors. The coefficient multiplying these terms is large and ignoring these terms (i. e., the error in the calibrated mirror position) will generally result in an erroneous determination of the attitude errors except yaw and roll at zero obliquity.

In summary the error in the calibrated mirror stereo position introduces two variables which are highly correlated with the attitude errors. Due to this correlation, when using the program, the mirror stereo error is ignored and the determined attitude errors are erroneous, especially for non-zero obliquity angles. For  $\Omega = 0$  the yaw and roll error determination is not affected by the mirror errors. However, at larger roll angles even the yaw and roll error determination is further degraded.

d. Correlation - Multi-Frame Case

Considering only attitude errors (i. e.,  $\sigma_{x0} = \sigma_{y0} = \Delta \Sigma = 0$ ) equations 3 and 4 may be written in matrix notation for the j th frame

$$A^j \delta = \sigma^j$$

where

$$A^j = \begin{bmatrix} A_{11}^j & A_{12}^j & A_{13}^j \\ A_{21}^j & A_{22}^j & A_{23}^j \end{bmatrix}$$

$$\delta = \begin{bmatrix} \psi \\ \theta \\ \varphi \end{bmatrix}$$

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$$\sigma^j = \begin{bmatrix} \sigma_x^j \\ \sigma_y^j \end{bmatrix}$$

and the  $i$  subscript referring to the  $i$  th PBM in the  $j$  th frame has been dropped for convenience.

For two frames  $j$  and  $k$  equations 3 and 4 may be written

$$\begin{bmatrix} \mathbf{A}^j \\ \mathbf{A}^k \end{bmatrix} \delta = \begin{bmatrix} \sigma^j \\ \sigma^k \end{bmatrix}$$

where the total matrices have been partitioned. Therefore, the normal equations in the least squares process have the coefficient matrix

$$\mathbf{B} = \begin{bmatrix} \mathbf{A}^j \\ \mathbf{A}^k \end{bmatrix}^T \begin{bmatrix} \mathbf{A}^j \\ \mathbf{A}^k \end{bmatrix} + \begin{bmatrix} \mathbf{R}^j \\ \mathbf{R}^k \end{bmatrix}^T \begin{bmatrix} \mathbf{A}^j \\ \mathbf{A}^k \end{bmatrix}$$

with elements

$$b_{mn} = \sum_j \sum_r a_{r m}^j a_{r n}^j \quad m, n = 1, 2, 3$$

where  $j$  is the frame and  $r$  is the rows of  $\mathbf{A}$ . Only the triangular matrix is given since  $\mathbf{B}^T = \mathbf{B}$  (i. e.,  $\mathbf{B}$  is symmetric). Therefore, explicitly the elements are as follows

$$b_{11} = \sum_j \left[ \tan^2 \Omega_j + \frac{\tan^2 \Sigma_j}{\cos^2 \Sigma_j} \right]$$

$$b_{12} = \sum_j \tan \Omega_j$$

$$b_{13} = \sum_j \left[ \frac{\tan \Sigma_j}{\cos \Omega_j} \right]$$

$$b_{22} = \sum_j \left[ \left( 1 + \frac{\tan^2 \Sigma_j}{\cos^2 \Omega_j} \right)^2 + \frac{\tan^2 \Sigma_j \tan^2 \Omega_j}{\cos^2 \Omega_j} \right]$$

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$$b_{23} = \sum_j \left[ \frac{\tan \Sigma_j \tan \Omega_j}{\cos \Omega_j} \left( 1 + \frac{1}{\cos^2 \Omega_j} + \frac{\tan^2 \Sigma_j}{\cos^2 \Omega_j} \right) \right]$$

$$b_{33} = \sum_j \left[ \frac{\tan^2 \Sigma_j \tan \Omega_j}{\cos^2 \Omega_j} + \frac{1}{\cos^4 \Omega_j} \right]$$

where the subscript  $j$  on the mirror stereo and obliquity angles refer to the  $j$ th frame.

The variance/covariance matrix is given by  $C = B^{-1}$  so that the correlation is given by

$$\rho_{pq} = \frac{C_{pq}}{\sqrt{C_{pp} C_{qq}}} = \frac{\Delta_{pq}}{\sqrt{\Delta_{pp} \Delta_{qq}}}$$

where  $\Delta_{pq}$  is the co-factor of  $b_{pq}$ . Explicitly

$$\rho_{\psi, \theta} = \frac{b_{23} b_{13} - b_{12} b_{33}}{\sqrt{(b_{22} b_{33} - b_{23}^2) (b_{11} b_{33} - b_{13}^2)}}$$

$$\rho_{\psi, \varphi} = \frac{b_{12} b_{23} - b_{22} b_{13}}{\sqrt{(b_{22} b_{33} - b_{23}^2) (b_{11} b_{22} - b_{12}^2)}}$$

$$\rho_{\theta, \varphi} = \frac{b_{12} b_{13} - b_{11} b_{23}}{\sqrt{(b_{11} b_{33} - b_{13}^2) (b_{11} b_{22} - b_{12}^2)}}$$

so that in general it is difficult to derive an analytical expression for the correlation.

While the correlation is a complex function of the mirror stereo and obliquity angles, results can be obtained for multiple frames of the previously discussed special cases.

Case IA  $\Sigma = 0, \quad \Omega \neq 0$

For multiple frames of  $\Sigma = 0$  photography, the correlation is given by

$$\rho_{\theta, \varphi} = \rho_{\psi, \varphi} = 0$$

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$$-\rho_{\psi, \theta} = \left[ \frac{\sum_j \tan^2 \Omega_j}{\sum_j j} \right]^{1/2}$$

Case IB  $\Sigma \neq 0, \Omega = 0$ 

Since stereo pairs will be considered separately then the only non zero values of  $\Sigma_j$  and  $\Sigma_k$  will be  $\Sigma_j = \Sigma_k$ . However, the obliquity for this case is zero and for two frames with the same obliquity and mirror stereo angle  $[A^j] = [A^k]$ . Therefore, the multiple frames case is exactly the same as the single frame case with multiple bench marks.

Case IC  $\Sigma = 0, \Omega = 0$ 

It is obvious (Case IB) that multiple frames of nadir photography are exactly the same as the single frame case with multiple bench marks.

Case ID Stereo Pairs

For a single stereo pair, the correlation is given by

$$\rho_{\theta, \varphi} = \rho_{\psi, \varphi} = 0$$

$$\rho_{\psi, \theta} = \frac{-\sin \Omega \cos^2 \Omega}{[\tan^2 \Sigma (1 + \tan^2 \Sigma)^2 + \sin^2 \Omega \cos^4 \Omega]^{1/2}}$$

Therefore, allowing a correlation of about  $\rho_{\psi, \theta} = .2$  the solution for the yaw and pitch errors will be reasonable for  $-3.5^\circ \geq \Omega \geq 3.5^\circ$ . The yaw-pitch correlation is plotted in Figure 1.

Case IE  $\Sigma_1 = \Sigma_2, \Omega_1 = -\Omega_2$ 

For equal mirror stereo angles along with equal and opposite obliquity angles, the correlation is given by

$$\rho_{\psi, \theta} = \rho_{\theta, \varphi} = 0$$

$$\rho_{\psi, \varphi} = \frac{-\tan \Sigma}{\sqrt{b_1 b_3}}$$

where

$$b_1 = \tan^2 \Omega + \frac{\tan^2 \Sigma}{\cos^2 \Omega}$$

$$b_3 = \tan^2 \Sigma \tan^2 \Omega + \frac{1}{\cos^2 \Omega}$$

From the yaw-roll correlation plotted in Illustration 19, the correlation is small ( $\rho_{\psi, \varphi} \geq .2$ ) for  $\Omega \geq 40^\circ$ .

Results can also be derived for various combinations of the above cases.

Case IIA  $\Sigma = 0, \Omega \neq 0$  plus  $\Sigma = 0, \Omega = 0$ 

$$\rho_{\theta, \varphi} = \rho_{\varphi, \psi} = 0$$

$$\rho_{\psi, \theta} = \frac{-1}{\sqrt{2}}$$

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$$\rho_{\theta, \varphi} = \rho_{\psi, \theta} = 0$$

$$\rho_{\psi, \varphi} = \frac{1}{\sqrt{2}}$$

Case IIC  $\Sigma \neq 0, \Omega = 0$  plus  $\Sigma = 0, \Omega \neq 0$ 

$$\rho_{\psi, \theta} = \frac{-P \tan \Omega}{\sqrt{AC}}$$

$$\rho_{\psi, \varphi} = \frac{(1 + Q^2) \tan \Sigma}{\sqrt{BC}}$$

$$\rho_{\theta, \varphi} = \frac{-\tan \Sigma \tan \Omega}{\sqrt{AB}}$$

where

$$P = 1 + \frac{1}{\cos^4 \Omega}$$

$$Q = (1 + \tan^2 \Sigma)$$

$$A = P \tan^2 \Omega + \frac{\tan^2 \Sigma}{\cos^4 \Omega}$$

$$B = Q^2 (\tan \Sigma + \tan^2 \Omega) + \tan^2 \Sigma$$

$$C = (1 + Q^2) P$$

Case IID  $\Sigma = 0, \Omega = 0$  plus  $\Sigma \neq 0$  plus  $\Sigma = 0, \Omega \neq 0$ 

For the combination of the three special cases, the correlations are given by the same formulas as Case IIC where

$$P' = P + 1$$

$$Q' = Q + 1$$

Plots of the correlations are presented in Illustration 20 for Case IID. Case IIC will have the same general shape with slightly larger correlations. Note that it is not possible to choose an obliquity angle or range of angles for which all three correlations are small.

Consideration of other combinations of frames is very algebraically complex. However, a sufficient but not necessary condition for the correlation to be zero is that the co-factor  $\Delta_{pq} = 0$ . Several results can be presented utilizing the above condition.

Case IIIA Multiple Stereo Pairs

$$\rho_{\psi, \varphi} = \rho_{\theta, \varphi} = 0$$

$$\Delta_{\psi, \theta} = 0 \text{ if } \sum_j \tan \Omega_j = 0$$

Case IIIB Stereo Pairs Plus All Other Combinations

The one exception not included in this case is  $\pm \Sigma_j = \Sigma_k, \Omega_j \neq \Omega_k$  and non-zero. For

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$$\sum_j \tan \Omega_j = 0$$

the correlations are given by

$$\rho_{\psi, \theta} = 0$$

$$\rho_{\theta, \varphi} = 0$$

where

$$\rho_{\psi, \varphi} = \frac{\tan \Sigma}{\sqrt{b_{11} b_{33}}}$$

$$b_{11} = 3 \tan^2 \Omega + 2 \frac{\tan^2 \Sigma}{\cos^2 \Omega} + \tan^2 \Sigma$$

$$b_{33} = 2 \frac{\tan^2 \Sigma \tan^2 \Omega}{\cos^2 \Omega} + \frac{3}{\cos^4 \Omega} + 2$$

It can be seen from the plot of  $\rho_{\psi, \varphi}$  in Illustration 21 that the roll-yaw correlation is small ( $\rho_{\psi, \varphi} \leq .2$ ) for  $\Omega \geq 10^\circ$

While including all combinations is too algebraically complex for analytical analysis, it is interesting to investigate what combinations are permissible when requiring zero correlation. Since  $b_{11}$ ,  $b_{22}$  and  $b_{33}$  cannot be zero except for  $b_{11}$  which is zero for the trivial case of  $\Sigma = \Omega = 0$  then for zero correlation it is necessary that  $b_{12} = b_{13} = b_{23} = 0$  or

$$\sum_j \tan \Omega_j = 0$$

$$\sum_j \frac{\tan \Sigma_j}{\cos \Omega_j} = 0$$

$$\sum_j \frac{\tan \Sigma_j \tan \Omega_j}{\cos \Omega_j} \left[ 1 + \frac{\tan^2 \Sigma_j}{\cos^2 \Omega_j} + \frac{1}{\cos^2 \Omega_j} \right] = 0$$

It does not appear likely that for several randomly chosen frames that the above three conditions would be satisfied. Also for several frames of photography the number of variables  $\Omega_j$  and  $\Sigma_j$  greatly exceeds the number of constraining equations. However, from the following table

	$b_{12}$	$b_{13}$	$b_{23}$
$\Sigma = 0 \Omega = 0$	0	0	0
$\Sigma = 0 \Omega \neq 0$	$\tan \Omega$	0	0
$\Sigma \neq 0 \Omega = 0$	0	$\tan \Sigma$	0
$\Sigma_1 = -\Sigma_2, \Omega_1 = \Omega_2$	$2 \tan \Omega$	0	0
$\Sigma_1 = \Sigma_2, \Omega_1 = -\Omega_2$	0	$2 \frac{\tan \Sigma}{\cos \Omega}$	$2 \frac{\tan \Sigma \tan \Omega}{\cos \Omega} \left[ 1 + \frac{\tan^2 \Sigma}{\cos^2 \Omega} + \frac{1}{\cos^2 \Omega} \right]$



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it is obvious that considering multiple frames of stereo and  $\Sigma = 0$  photography such that  $\Sigma \tan \Omega_j = 0$  then all correlations will be zero.

e. Effect of Non-Zero Mirror, Map and Ephemeris Errors

The effect of an error in the calibrated mirror stereo position is shown in the following chart where the indicated attitude errors are coupled to the mirror stereo position error.

Error	$\Sigma = 0$		$\Sigma \neq 0$		Pairs	
	$\Omega \neq 0$	$\Omega = 0$	$\Omega \neq 0$	$\Omega = 0$	$\Omega \neq 0$	$\Omega = 0$
$\psi$	$\Delta \Sigma$	*	$\Delta \Sigma$	none	$\Delta \Sigma^+ + \Delta \Sigma^-$	none
$\theta$	$\Delta \Sigma$	$\Delta \Sigma$	$\Delta \Sigma$	$\Delta \Sigma$	$\Delta \Sigma^+ + \Delta \Sigma^-$	$\Delta \Sigma^+ + \Delta \Sigma^-$
$\varphi$	none	none	$\Delta \Sigma$	none	$\Delta \Sigma^+ - \Delta \Sigma^-$	none

Therefore, some errors are not affected by a mirror stereo position error for different combinations of  $\Sigma$  and  $\Omega$ .

The effect of non-zero map and ephemeris errors can be determined by replacing  $\sigma$  by  $\sigma - \sigma_0$ . Examination of equations 5 - 10 shows the coupling between the attitude and ephemeris errors. The components which are coupled depend upon the obliquity angle as shown by the following chart.

Error	$\Sigma = 0$		$\Sigma \neq 0$		Pairs	
	$\Omega \neq 0$	$\Omega = 0$	$\Omega \neq 0$	$\Omega = 0$	$\Omega \neq 0$	$\Omega = 0$
$\psi$	$\sigma_{x0}$	*	both	$\sigma_{y0}$	both	none
$\theta$	$\sigma_{x0}$	$\sigma_{x0}$	both	$\sigma_{x0}$	both	$\sigma_{x0}$
$\varphi$	$\sigma_{y0}$	$\sigma_{y0}$	both	$\sigma_{y0}$	both	$\sigma_{y0}$

For stereo pairs it has been assumed that  $\sigma_0^+ = \sigma_0^-$  and of course the bench mark is common to both frames. Note from the chart that using stereo pairs at zero obliquity the yaw error can be determined even for non-zero map and ephemeris errors. However, in actuality  $\sigma_0^+ \neq \sigma_0^-$  and even the yaw error determination will be degraded.

f. Test Case Results

Prior to the appropriate equations being available for analysis, several test cases were designed to help determine how to use the program. Initially, these test cases were designed with zero ephemeris error and such that for each bench mark  $\sigma_x = \sigma_y = 0$ . Then inputting attitude errors the question was whether or not the solution for attitude errors would be identical to the errors that were input. Without exception the results of the previous analysis were confirmed by the test cases.

\* Yaw cannot be determined.

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Since the actual data is carefully edited to remove obvious errors, data noise (random map reading error) was simulated by introducing a small error such that on an individual frame and bench mark basis  $\sigma_x$  and  $\sigma_y$  were not zero while the average  $\sigma_x$  and  $\sigma_y$  per frame were zero. The addition of the data noise did not seriously affect the determination of the attitude errors.

The effect of non-zero values for  $\sigma_{x0}$  and  $\sigma_{y0}$  was simulated by introducing a data bias, i. e., for each bench mark  $\sigma_x$  and  $\sigma_y$  on a frame basis are non-zero and equal. The test case results, as anticipated, showed that generally the effect of ephemeris and/or map errors degraded the determination of the attitude errors. In fact these errors combined to either increase or decrease the attitude errors depending upon the sign of the data bias, i. e.,  $\sigma_{x0}$  and  $\sigma_{y0}$  and the magnitude of the obliquity angle. In particular, the correlations between  $\sigma_{x0}$  and  $\theta$  along with  $\sigma_{y0}$  and  $\phi$  were very high.

g. Summary

(1) The TEAR program can be used post flight, prior to the availability of stellar photography, to attempt to determine gross sub-system malfunctions. Any attempt to utilize the program without inputs from the stellar reduction to detect small, nominal, within spec subsystem errors may be misleading. In particular, the program should be used first to determine the in-track and cross-track ephemeris errors independent of attitude errors. Secondly, the attitude errors should be determined independent of ephemeris errors. When solving for the attitude errors on a frame basis, the special cases should be used to determine the errors. The attitude errors should only be used to examine the relative behavior between adjacent frames or pairs. However, even this use of the program may be misleading due to the possible compensating effect of the various errors.

(2) The following table indicates what combinations are permissible when solving for attitude errors in the presence of other system errors and when the other system errors are not so large so as to be overriding.

	$\Omega = 0$			$\Omega \neq 0$		
	Y	P	R	Y	P	R
$\Sigma =$	No	Yes	Yes	No	Yes*	Yes
$\Sigma \neq$	No	Yes	No	No	Yes*	No
Pairs	Yes	Yes	Yes	No	Yes*	Yes

The asterisk indicates serious degradation of accuracy at large obliquity angles.

(3) An error in the calibrated stereo mirror position (i. e., true position not equal to calibrated position) will add an additional unknown  $\Delta \Sigma$  term to the  $\sigma_x$  equation for  $\Sigma = 0$  and  $\Sigma \neq 0$ . For stereo pairs an additional unknown term involving  $\Delta \Sigma^+$  and  $\Delta \Sigma^-$  will be added to the two equations in  $\sigma_x^+$  and  $\sigma_x^-$ . The necessity of neglecting these terms due to the usage of the program will degrade any attitude

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error determination resulting from a  $\sigma_x$  equation depending upon the obliquity angle.

(4) Since  $\sigma_{x0}$  should appear in the  $\sigma_x$  equation and  $\sigma_{y0}$  in the  $\sigma_y$  equation, all results except  $\psi$  from stereo pairs with  $\Omega = 0$  will be seriously degraded by the necessity of neglecting these terms. The degree that the results will be degraded is dependent upon the value of the obliquity angle.

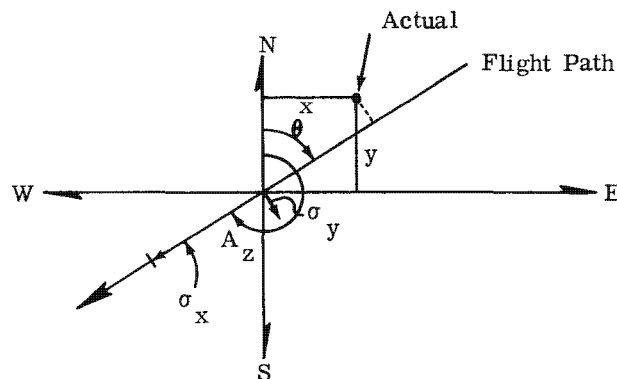
(5) Since pitch and roll error are so closely correlated with mirror position error (crab and stereo) solving for attitude error while neglecting mirror position error will also indicate gross malfunctions in the mirror and/or attitude control system. One way of deciding which system is malfunctioning is to observe some pattern which shows the mirror malfunction in a particular position.

(6) Utilizing the attitude orientation information obtained from the reduction of stellar photography, the TEAR program can be used to attempt to determine the ephemeris errors both in the best fit and command ephemerides. Since the BFE errors are slowly varying over a payload operating span, a very limited amount of attitude orientation data derived from the stellar photography will serve to determine the ephemeris errors. Inputting the ephemeris errors will then allow for solution of the attitude errors on a per frame and/or stereo pair basis with the same limitations as discussed in the analysis. Possible problems may occur due to map and instrument errors along with the correlation between the radial, cross-track and in-track ephemeris errors.

#### 4. Accuracy

The accuracy of the miss distance calculation is largely dependent upon the accuracy of the photo map match, i. e., the accuracy of determining the map coordinates of the PBM. In order to determine the influence of map errors and human errors in reading the map, the following analysis was conducted.

Denoting the map coordinates of the intersection of the traced ray with the earth as the computed position and the map coordinates of the PBM as the actual position, the coordinate system as shown below can be established with the origin at the computed position.

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Thus  $x$  is the longitude of the actual point relative (actual-computed) to the computed point and  $y$  is the latitude of the actual point relative to the computed point.  $A_z$  is the azimuth given by

$$A_z = \sin^{-1} \left[ \frac{\cos i}{\cos \varphi} \right]$$

where  $i$  is the orbit inclination and  $\varphi$  is the geocentric latitude of the computed point. Also, for convenience

$$\theta = A_z - 180^\circ$$

Therefore the latitude and longitude of the actual point are related to  $\sigma_x$  and  $\sigma_y$  by

$$\begin{bmatrix} -y \\ x \end{bmatrix} = \begin{bmatrix} \cos \theta & \sin \theta \\ -\sin \theta & \cos \theta \end{bmatrix} \begin{bmatrix} \sigma_x \\ \sigma_y \end{bmatrix}$$

so that

$$\begin{bmatrix} \sigma_x \\ \sigma_y \end{bmatrix} = \begin{bmatrix} \cos \theta & -\sin \theta \\ \sin \theta & \cos \theta \end{bmatrix} \begin{bmatrix} -y \\ x \end{bmatrix}$$

Therefore, an error in the map or in measuring  $x$  and  $y$  will be reflected in  $\sigma_x$  and  $\sigma_y$  according to

$$\begin{bmatrix} \delta \sigma_x \\ \delta \sigma_y \end{bmatrix} = \begin{bmatrix} \cos \theta & -\sin \theta \\ \sin \theta & \cos \theta \end{bmatrix} \begin{bmatrix} -\delta y \\ \delta x \end{bmatrix}$$

which shows the dependence upon the azimuth and consequently the inclination of the orbit and the latitude of the computed point.

For polar orbits, and north to south camera operation

$$\delta \sigma_x = -\delta y = -.15 \text{ NM}$$

$$\delta \sigma_y = \delta x = .15 \text{ NM}$$

for typical map and measurement errors of .15 NM. For an orbit inclination of  $105^\circ$  and at approximately  $60^\circ$  north latitude,  $\theta \cong 26.8^\circ$  so that

$$\delta \sigma_x \cong -.2 \text{ NM}$$

$$\delta \sigma_y \cong .07 \text{ NM}$$

which shows that for this particular orbit and latitude the error in the  $\sigma_x$  and/or  $\sigma_y$  determination can be larger or smaller than the error in the latitude and/or longitude measurement.

##### 5. Method of Making TEAR Runs

For the PBM's that were chosen the corresponding latitude, longitude, time and film measurement along with the appropriate satellite vehicle positions were input to the TEAR Computer Program.

a. Using the BFE satellite position, the  $\sigma_x$  and  $\sigma_y$  for each individual PBM was determined and then edited by comparison with the average  $\sigma_x$  and  $\sigma_y$  for the entire frame. A comparison was also made of the  $\sigma_x$  and  $\sigma_y$  of neighboring PBM's. All PBM's whose  $\sigma_x$  and  $\sigma_y$  were not in close agreement

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were identified and rechecks of all the PBM parameters were then made. A second computer run was then made with the corrected PBM's to check for agreement with surrounding points. At this point all bench marks that did not agree within  $\pm 0.3$  nautical miles with surrounding points were eliminated from further analysis. Also, all frames with only one bench mark per frame were excluded from further consideration.

b. The miss distances ( $\sigma_x$  and  $\sigma_y$ ) shown on Geo-Positioning Data Sheets, Appendix A, were computed for each individual bench mark using the BFE satellite position, and then averaged for the entire photographic frame. Regarding the PBM map position coordinates as the actual position and the coordinates of the traced ray as the predicted position, then  $\sigma_x$  is positive if the actual position is south of the predicted position and  $\sigma_y$  is positive if the actual position is east of the computed position. The Mean Error (ME) shown is also computed for each frame.

c. Due to limitations posed by the correlation between the yaw, pitch and roll error sub-components the following procedure was used when determining the attitude errors:

(1) For photography where  $\Sigma = 0$  and the roll angle is zero, the total system error was considered to be due to attitude error and the attitude error was resolved into pitch and roll components without any loss of accuracy even in the presence of yaw. The accuracy of the determined pitch error is degraded at greater roll angles.

(2) For frames where  $\Sigma \neq 0$ , and for zero roll angle again the total system error was considered to be caused by attitude error. However, only the pitch component of the attitude error could be determined with reasonable accuracy. The pitch error determination is even less accurate for greater roll angles.

(3) For stereo pairs, again the total system error was considered to be caused by attitude error and the attitude error was resolved into pitch and roll components. For zero roll angle the yaw pitch and roll errors are accurate. The accuracy of the yaw and pitch error determination is degraded for greater roll angles.

d. In Case I the vehicle attitude errors were computed using the BFE satellite position. The results of this run are shown on the Data Sheets. In this case the system error is considered to be due entirely to attitude errors and is determined as outlined in Section 4. c.

e. In Case II the BFE errors were computed using the BFE satellite position. Errors shown on the Geo-Positioning Data Sheets under BFE Ephemeris Error columns are the in-track and cross-track errors that would result if all of the system error was caused entirely by the BFE errors.

f. In Case III the command ephemeris errors were computed using the command ephemeris satellite position. The errors in the command ephemeris are shown on the Geo-Positioning Data Sheets in terms of in-track and cross-track errors. This TEAR run used the commanded satellite vehicle position for comparison with the bench mark positions. The results shown here give the error that would result if

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all the system error was caused entirely by command ephemeris errors. Also shown is the difference between the command and best fit ephemerides where the sign convention of section g was utilized to show which difference is larger in absolute value.

g. The command and best fit ephemeris errors along with the mean error were averaged over each payload operating span on each rev and are displayed on the summation chart (page 87). This chart shows the in-track and cross-track error history for each ephemeris. In theory, the errors in the BFE should be less than the errors in the command ephemeris since no extrapolation is involved, the length of the fit span (tracking data interval) is longer, and the orbit determination is accomplished post flight in a more precise manner. These BFE and command ephemeris in-track and cross-track errors, along with the mean error, were differenced on the same summary chart. Since both the BFE and command ephemeris error determination used the same BM's and the same values for attitude, then the differences in the errors must be a slowly varying function. Examination of the differences show the trend to be the expected slowly varying function with some dispersion about this trend. This dispersion or noise is due to inaccuracy in the  $\sigma_x$  and  $\sigma_y$ , the in-track and the cross-track error determination. In other words, the difference between the true best fit and command ephemeris errors is a slowly varying function and an error in the determined ephemeris errors will result in the observed noise. The sign following the error number in the difference columns shows which ephemeris had the bigger error in accordance with the following sign convention.

$$\begin{aligned} - \epsilon \left[ \epsilon_{\text{CMD}} \right] &> \left[ \epsilon_{\text{BFE}} \right] \\ + \epsilon \left[ \epsilon_{\text{BFE}} \right] &> \left[ \epsilon_{\text{CMD}} \right] \end{aligned}$$

h. The last several GAMBIT missions indicated in some instances that the miss distances and consequently the ephemeris errors were, in magnitude, larger for the best fit than the command ephemeris. It has previously been pointed out that the ephemeris errors for either the command or best fit ephemeris result from resolving the total miss distances into only ephemeris errors. Also the attitude errors result from resolving the BFE total miss distances into only attitude errors. Thus, neither the attitude or ephemeris errors are correct. The following example will show that due to this method of using the TEAR program, it is not possible to determine which ephemeris has the larger errors. Consider the BFE to be absolutely correct (i. e., no ephemeris errors) and no map errors so that the miss distances are due to attitude or mirror position errors. More specifically assume the vehicle to have only a pitch error corresponding to a finite  $\sigma_x$  and a negligible  $\sigma_y$ . Now the TEAR program normally is used to determine the ephemeris error due to  $\sigma_x$ , resulting in an in-track  $\rho$  approximately equal to  $\sigma_x$  even though the BFE was not in error. Now consider the command ephemeris position to be behind the BFE position by a distance  $-\rho$ . Then solving for the command in-track ephemeris error would give a  $\sigma_x = 0$  and a zero ephemeris error. Thus, the TEAR results would show the BFE to be in error by  $\rho$  and the command ephemeris to have zero error. Also if the command ephemeris were ahead of the BFE by a distance  $\rho$

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then solving for the command in-track ephemeris error would give  $2\sigma_x$  and  $2\rho$  ephemeris error. Due to the method of producing the BFE and since the command ephemeris is predicted, the conclusion may be drawn that the BFE errors must be smaller than the command ephemeris errors. Also the ephemeris errors as well as the difference between the command and best fit ephemeris errors is a slowly varying function. Thus, if the TEAR produced results show the BFE ephemeris errors to be larger than the command ephemeris errors, the true error must be an attitude, mirror or map error which, in fact, compensated for part of the command ephemeris error. Since accurate maps are used the contribution due to map errors should be small. Also, since the BFE is more accurate, then true attitude errors will tend to increase  $\sigma_x$  and  $\sigma_y$  when using the BFE satellite position. Conversely, since the command ephemeris is less accurate, then the attitude errors will tend to be more compensating, thus sometimes reducing the  $\sigma_x$  and  $\sigma_y$  when using the commanded vehicle position. Therefore, if the vehicle has attitude errors it is not surprising the BFE ephemeris errors will sometimes appear to be larger than the command ephemeris errors.

i. As mentioned previously for stereo pairs with the absolute value of obliquity less than about 3.5 deg the yaw, pitch and roll errors may be accurately determined.

Rev	Frames	Yaw	Pitch	Roll
7	18 & 19	.636	.599	-.069
10	11 & 12	.540	.395	-.410
21	5 & 6	.721	.486	.071
26	10 & 11	.047	.042	-.343
26	16 & 17	.369	.244	-.149
31	4 & 5	-.138	.325	-.153
42	28 & 29	.612	.194	-.307
48	8 & 9	.516	.105	-.569
54	20 & 21	.046	.025	-.460
58	4 & 5	.356	.382	-.102

j. For certain revs the various correlations between the attitude subcomponent errors were small so that the yaw pitch and roll errors could be determined on a rev basis as shown below.

Rev	Yaw	Pitch	Roll
5	.400	.294	-.272
24	.497	.143	-.356
27	.086	-.244	-.387
32	.720	-.009	-.344
37	.499	.403	-.236
58	-.042	.387	-.368

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k. Based on the data for the entire flight the average errors were

- $\sigma_x = .302 \text{ NM}$
- $\sigma_y = -.710 \text{ NM}$
- Yaw = .286 deg
- Pitch = .193 deg
- Roll = -.364 deg

5. Statistical Analysis

The in-track and cross-track miss distances were analyzed on a statistical basis. The procedure used was to make the usual frequency of occurrence plots, calculate the probability and plot the results on "probability paper" (Illustrations 22-27). Referring to Illustrations 22-27 the probability that the error  $\sigma$  will be greater than  $\sigma_0$  can be found by first determining the intersection of  $\bar{\sigma} - \sigma_0$  and  $\bar{\sigma} + \sigma_0$ , where  $\bar{\sigma}$  is the mean error, with the plotted line. Secondly the probability  $P_L$  (in percent) corresponding to  $\bar{\sigma} \pm \sigma_0$  to the left of  $\bar{\sigma}$  is read off the left hand scale. The corresponding probability  $P_R$  to the right of  $\bar{\sigma}$  is read off the right hand scale. The desired probability is then the positive difference of  $P_L$  and  $P_R$ .

Since the data bias and distribution is a function of the mirror stereo position and the vehicle roll angle the analysis was conducted using several combinations of  $\Sigma$  and roll. The analysis of the cross-track errors was accomplished for  $\Sigma = 0$  (Illustration 22) and  $\Sigma \neq 0$  (Illustration 23). For the in-track errors the break down was the same ( $\Sigma = 0$ , Illustration 24); however, it was necessary to consider sub cases for  $\Sigma \neq 0$ . Illustration 25 shows  $\pm \Sigma \neq 0$  for roll angles between  $\pm 15$  degrees. Illustration 26 is the same except the roll angle varies between 15 and 45 degrees. Illustration 27 shows the probability plot for the roll angle between -15 and -45 degrees. The results (mean and standard deviation) are summarized below where the units are NM.

For  $\sigma_y$

	$\Sigma = 0$	$\Sigma \neq 0$
Mean	-.6	-.7
SD	.4	.45

For  $\sigma_x$

		$\Sigma \neq 0$						
		$-15^\circ \rightarrow 15^\circ$		$15^\circ \rightarrow 45^\circ$		$-15^\circ \rightarrow -45^\circ$		
		$+\Sigma$	$-\Sigma$	$+\Sigma$	$-\Sigma$	$+\Sigma$	$-\Sigma$	
Mean	$\Sigma = 0$	.12	.2	.45	-.03	.37	.4	1.0
SD		.49	.6	.6	.5	.6	.8	.9



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## 6. Conclusions

a. The pitch (and stereo mirror position) error is closely correlated with the in-track ephemeris error and consequently  $\sigma_x$ . Therefore, the pitch error determined independently from ephemeris errors is not the actual pitch error but is highly biased by the presence of an actual in-track ephemeris error. In other words, while  $\sigma_x$  is caused principally by an in-track ephemeris error, then solving for pitch independently results in a pitch error equivalent to  $\sigma_x$  and consequently equivalent to the in-track ephemeris error. However, over a payload operating span the in-track ephemeris error is a slowly varying function so that the difference between the determined pitch error on succeeding frames is a meaningful number within the accuracy bounds of the  $\sigma_x$  and  $\sigma_y$  determination. The principle inaccuracy in the  $\sigma_x$  and  $\sigma_y$  determination is due to map errors.

The pitch error difference between succeeding frames is caused by either a pitch attitude error, an error in the stereo mirror position, or latitude errors in reading the maps. Without use of attitude data from the stellar camera it is not possible to determine whether this pitch difference is caused by the vehicle attitude control system, the map or the position of the stereo mirror. For this mission the vehicle telemetry indicated that the attitude control system (pitch) did not experience any large variations. Examining the difference between the determined pitch error on succeeding frames tends to confirm a relatively quiet attitude control system. The following exhibited unusual pitch behavior when compared to adjacent frames.

Rev	Frame	Stereo	Crab	Roll
8	12	V	2.0	39.70
9	22	F	2.0	-24.82
11	8	F	2.5	36.87
24	17	V	2.0	-40.41
40	14	F	2.0	-26.94
41	31	F	3.0	-21.27
43	20	F	2.5	-8.51

b. Since the orbit plane is usually accurately determined, the major contribution to  $\sigma_y$ , at least for  $\Sigma = 0$  photography, is due to roll and/or crab errors. For  $\Sigma \neq 0$ , the major contribution to  $\sigma_y$  is due to roll and/or crab errors. Since the best independent estimate of the BFE cross-track ephemeris error is about 300 feet, any  $\sigma_y$  in excess of 300 feet must be due to roll and/or crab errors. Therefore, the TEAR results (roll and/or crab errors) for  $\Sigma = 0$  photography give an accurate indication of these errors. For  $\Sigma \neq 0$ , the accuracy of the errors is slightly degraded. However, it is not possible to determine which sub-component is in error without additional inputs. Nevertheless,

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since the cross-track error is small and it is, again, a slowly varying function of time, the relative change of  $\sigma_y$  between succeeding frames gives an indication of the roll (or crab) and yaw errors. While it is very difficult to separate  $\sigma_y$  into a yaw and roll component, a large symmetrical change in  $\sigma_y$  between two adjacent frames is indicative of a predominant yaw error.

c. Comparing the in-track and cross-track miss distances on adjacent frames showed that the change was very large on the following frames:

$\Delta\sigma_x$		$\Delta\sigma_y$	
<u>Rev</u>	<u>Frame</u>	<u>Rev</u>	<u>Frame</u>
5	11	8	10
6	5	8	11
9	21	27	14
11	4	27	15
11	7		
24	17		
41	30		
43	18		

Note that all the above frames where  $\Delta\sigma_x$  is very large are the second frame of a stereo pair.

d. The in-track and cross-track miss distances for each Bench Mark on the long (100 sec) frames were examined in detail. Several plots of  $\sigma_x$  and or  $\sigma_y$  versus latitude appear as Illustrations 28 and 29. Since the mirror system remains fixed, the ephemeris errors are slowly varying and high accuracy maps were used, therefore, the miss distances reflect variations in the vehicle attitude during the frame. Illustration 28 shows a change in pitch near the center of the frame. Illustration 28 shows the effect of calculating the crab correction at the frame center. Illustration 29 shows a change in the vehicle roll attitude during the frame.

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CASE I D  
Stereo Pairs  
Yaw - Pitch

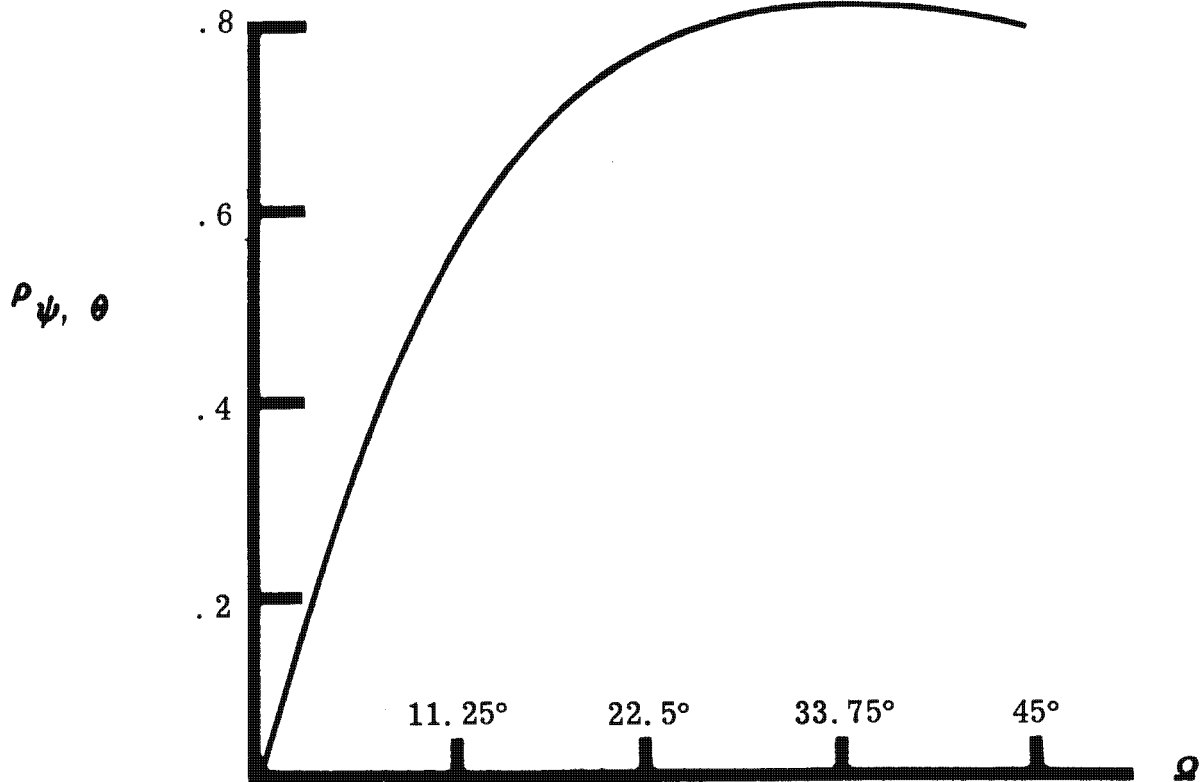
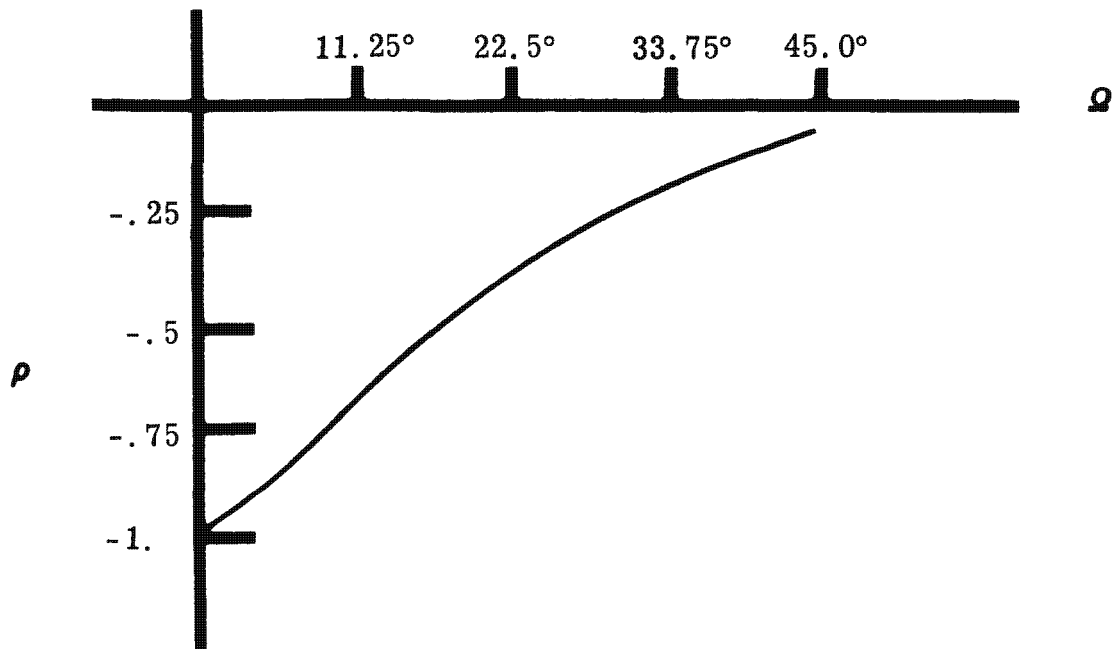


ILLUSTRATION 18

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CASE IE  
 $\Sigma_1 = \Sigma_2, \Omega_1 = -\Omega_2$   
Yaw - Roll



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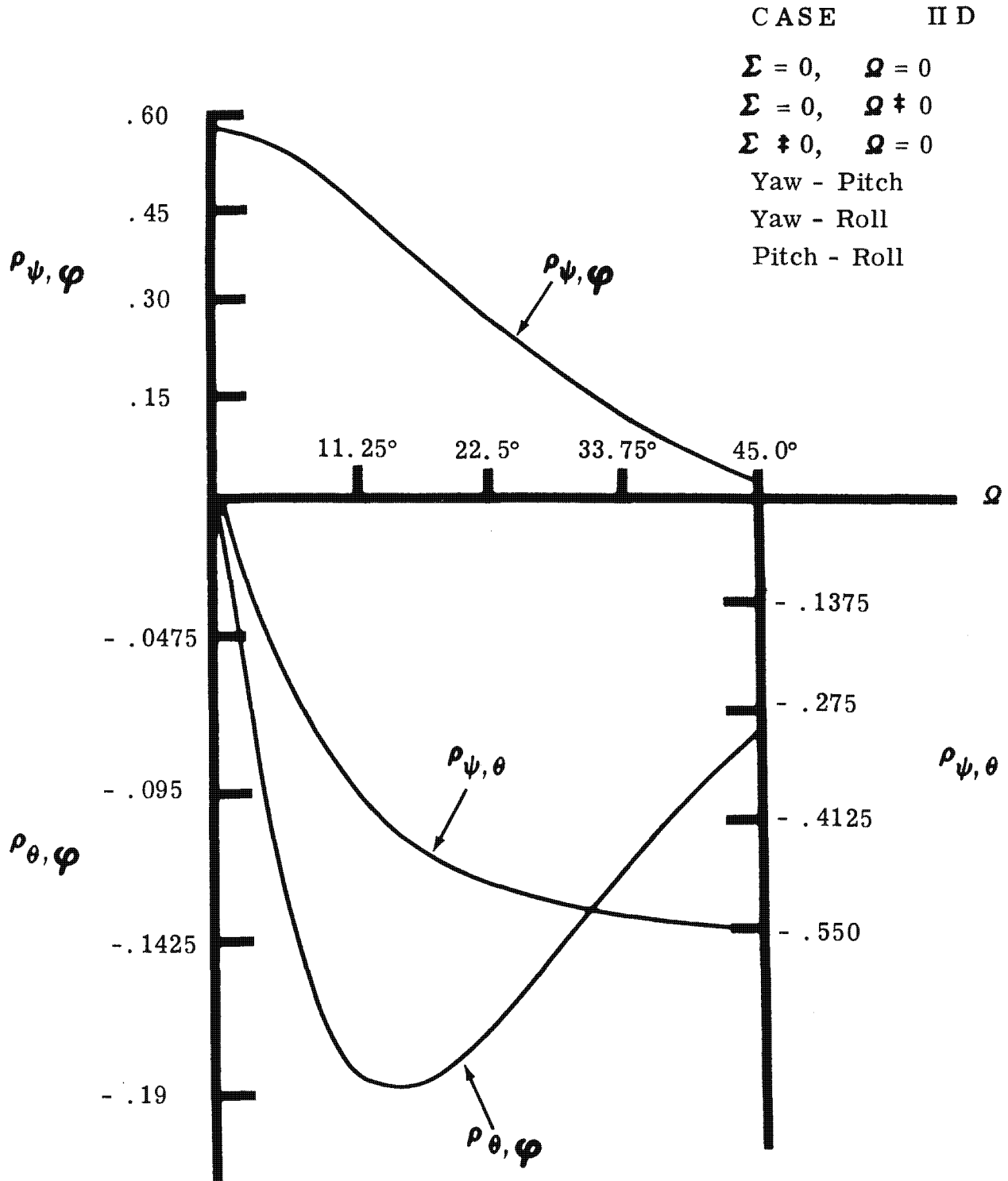


ILLUSTRATION 20

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CASE III B

Stereo Pairs

$$\Sigma = 0, \quad \Omega = 0$$

$$\Sigma \neq 0, \quad \Omega = 0$$

$$\Sigma = 0, \quad \Omega \neq 0$$

Yaw - Roll

$$\sum_j \Omega_j = 0$$

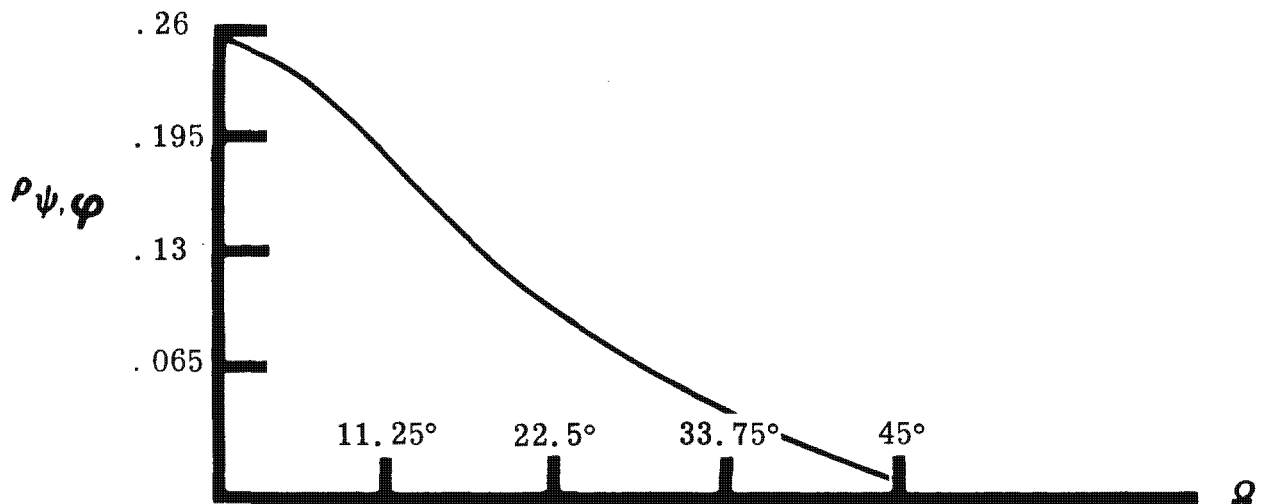


ILLUSTRATION 21

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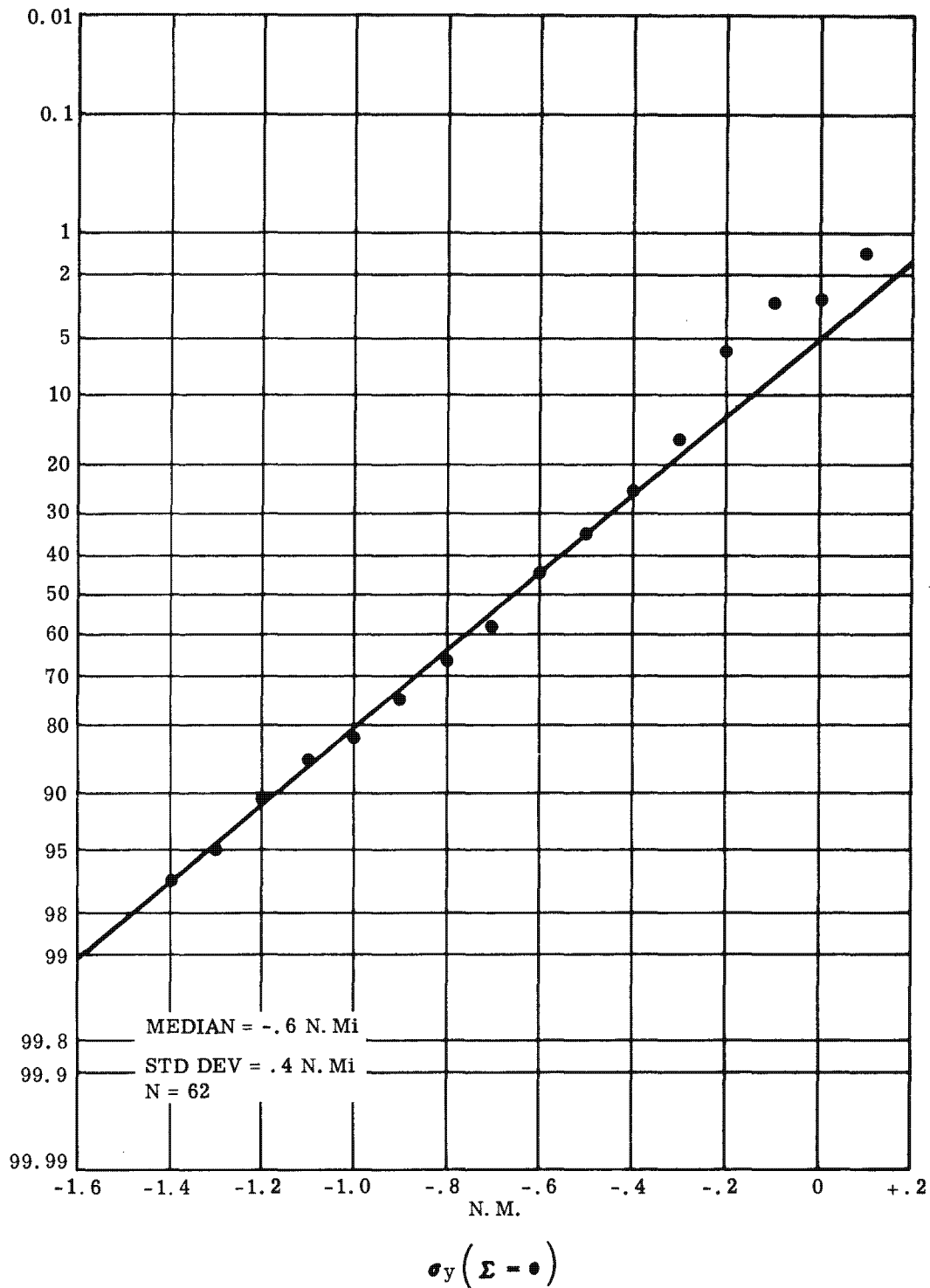


ILLUSTRATION 22

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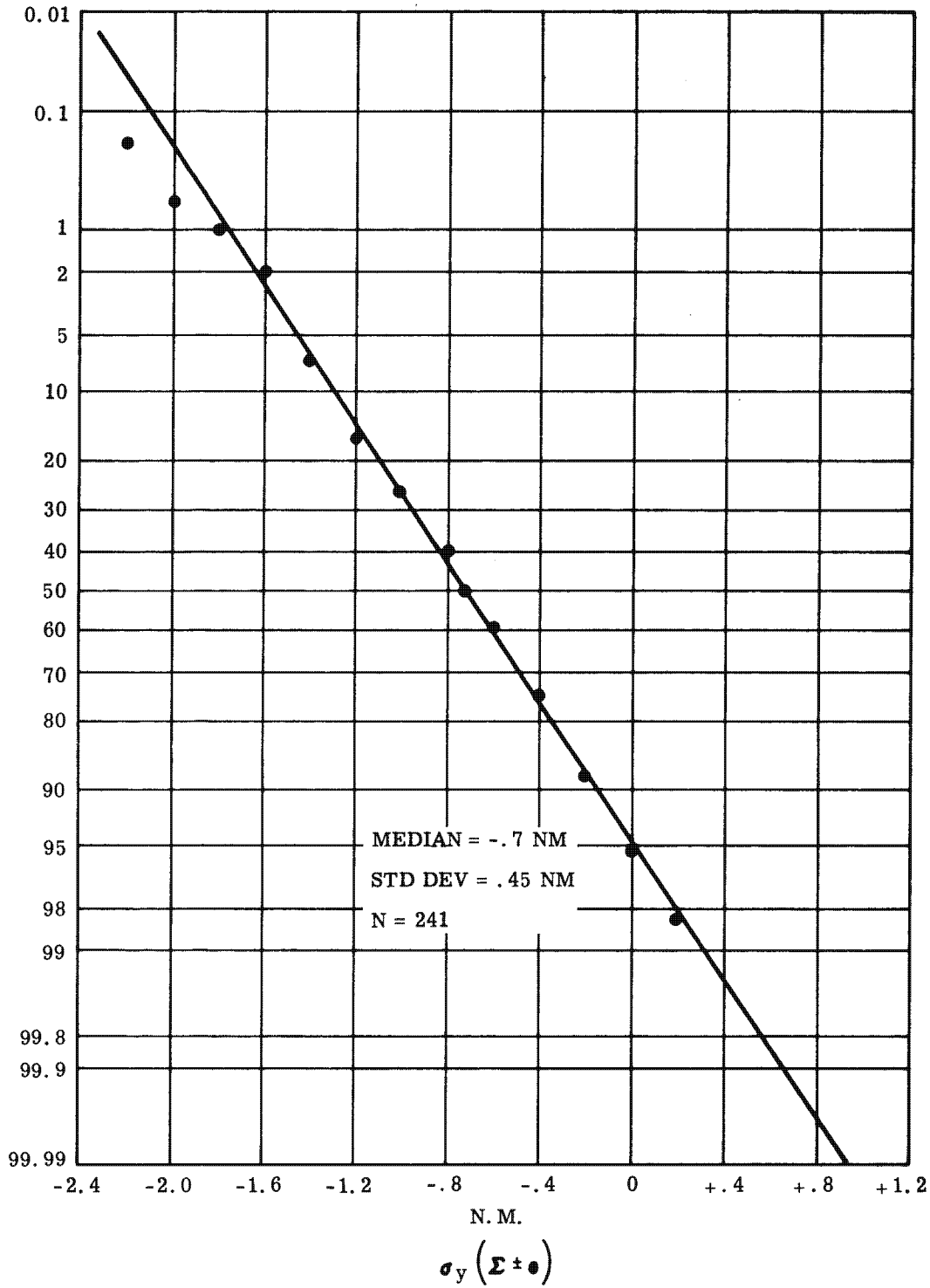


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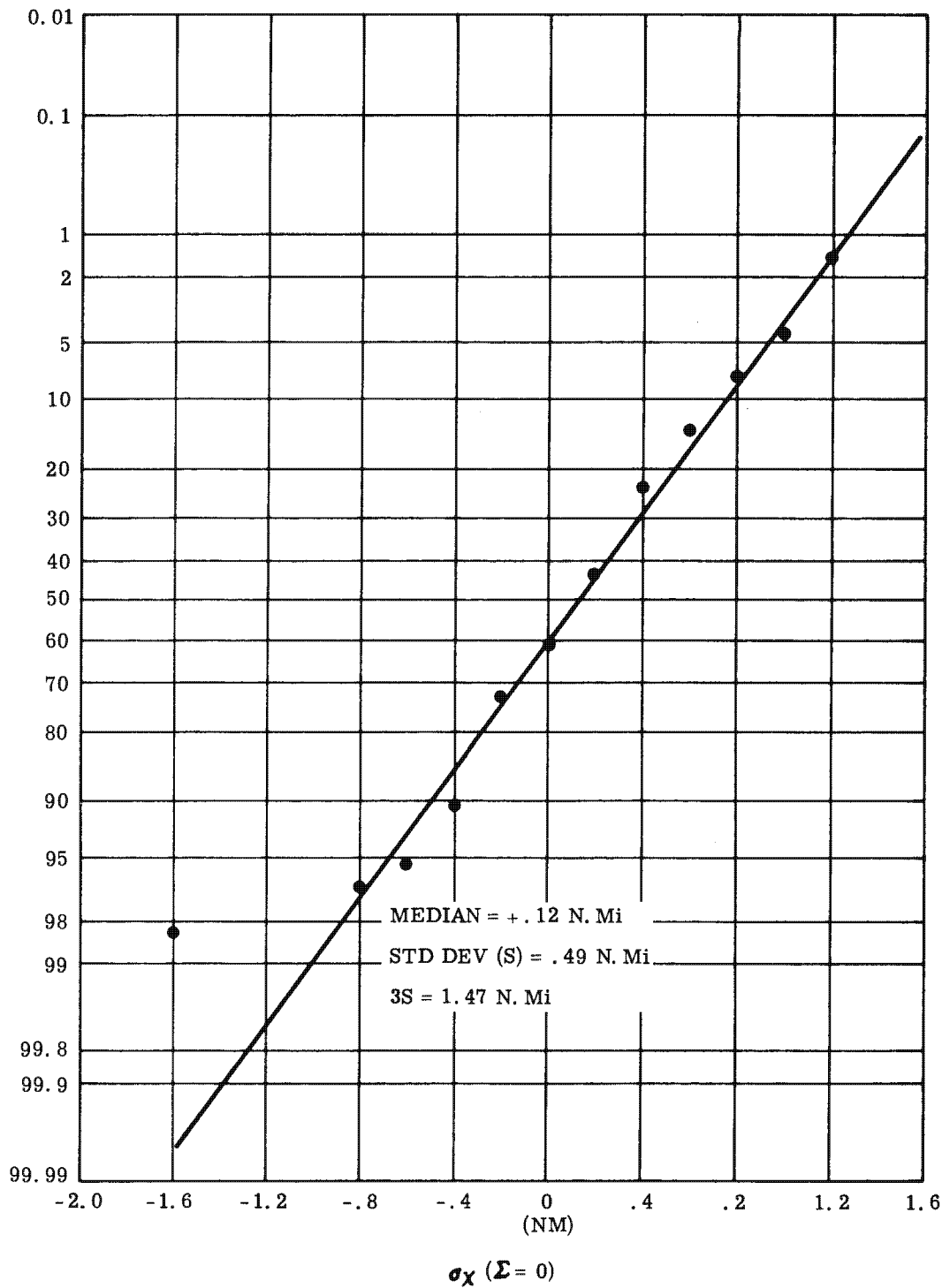


ILLUSTRATION 24

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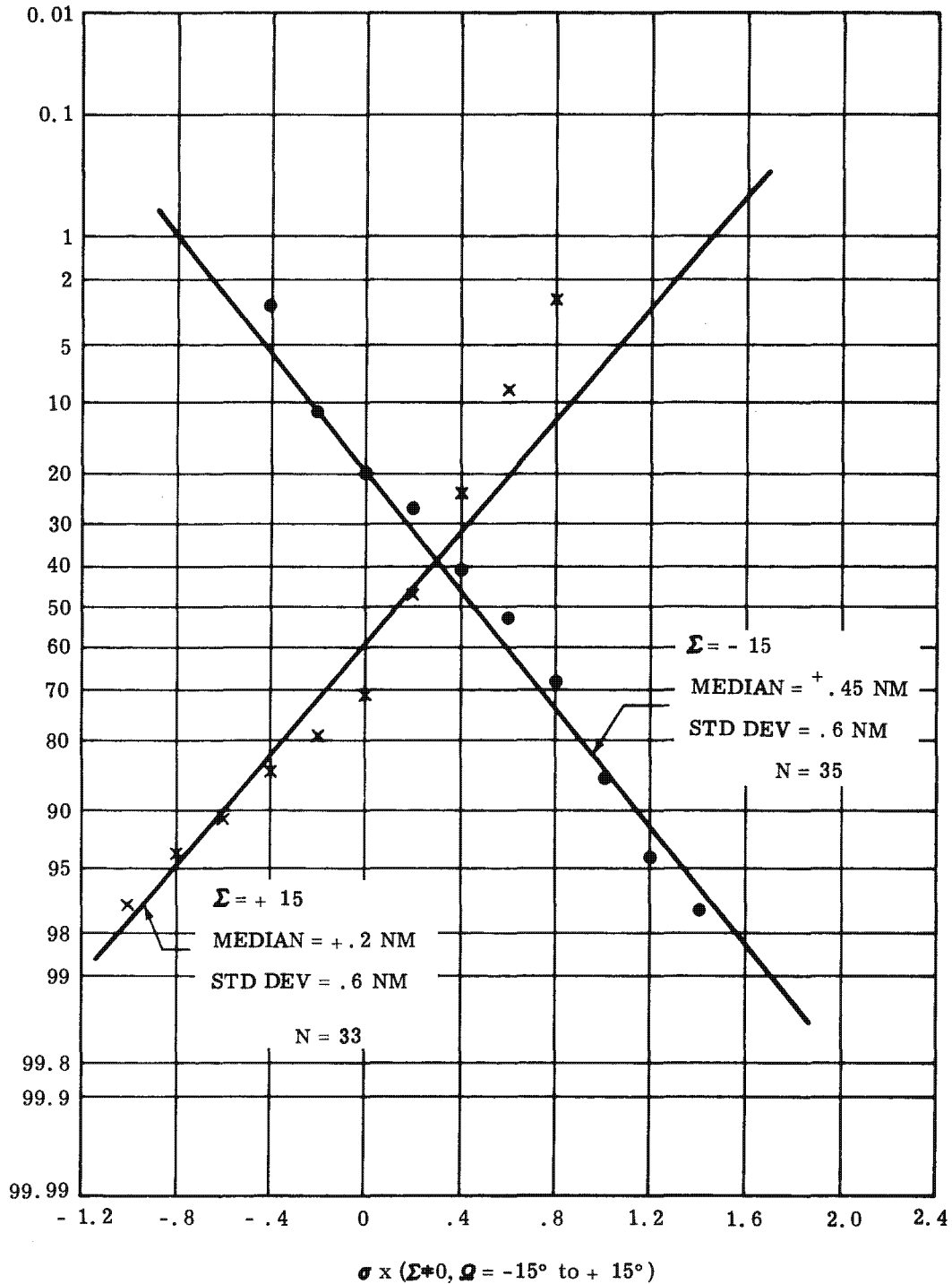


ILLUSTRATION 25

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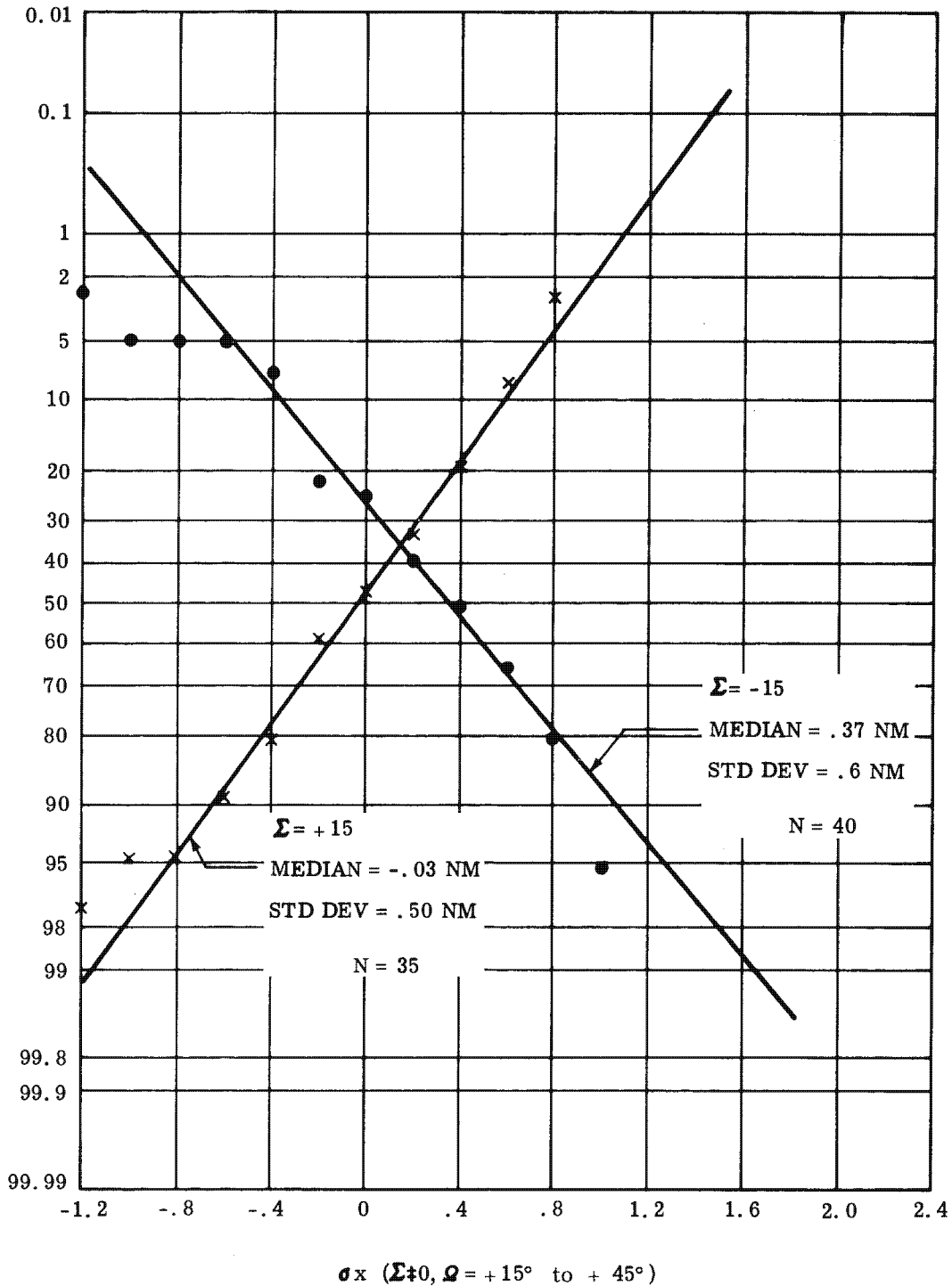


ILLUSTRATION 26

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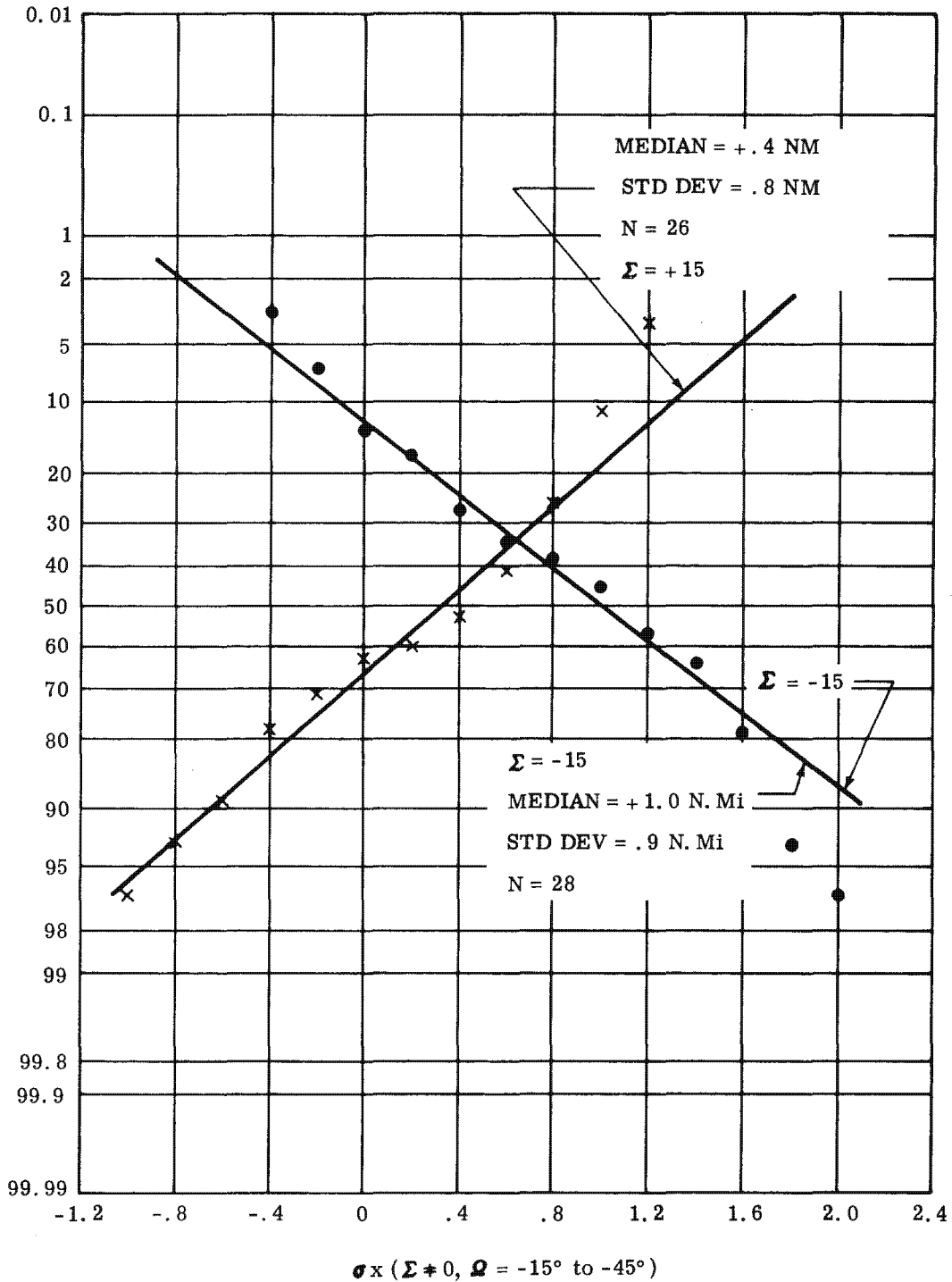


ILLUSTRATION 27

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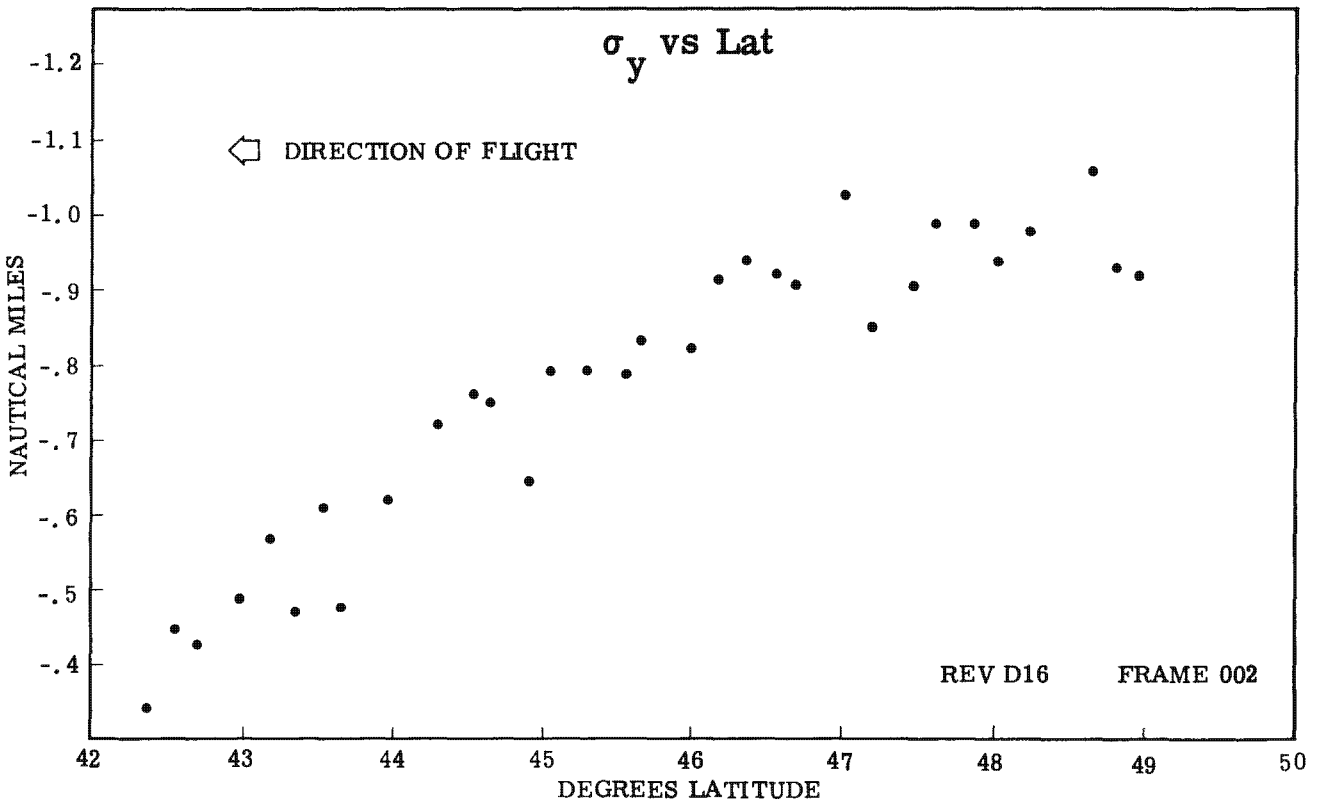
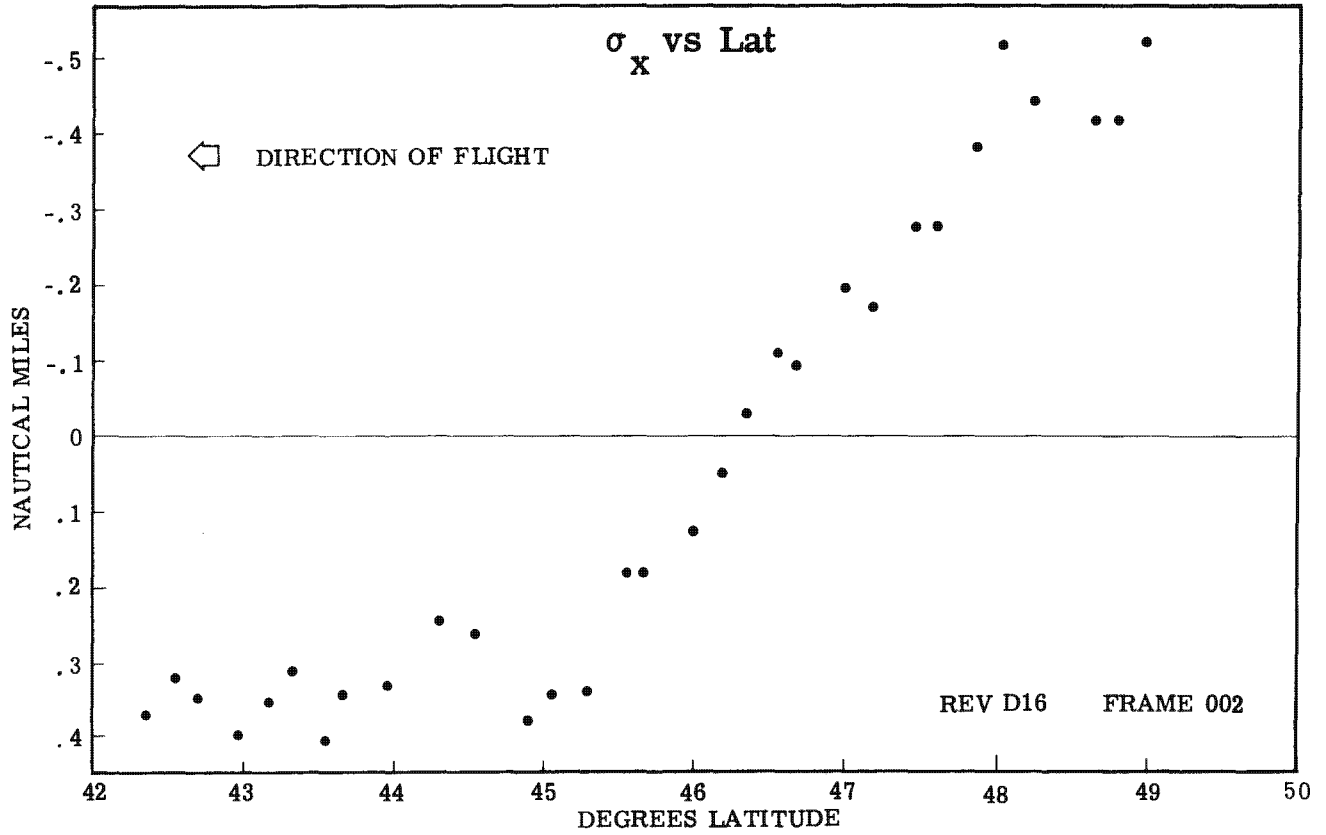


ILLUSTRATION 28

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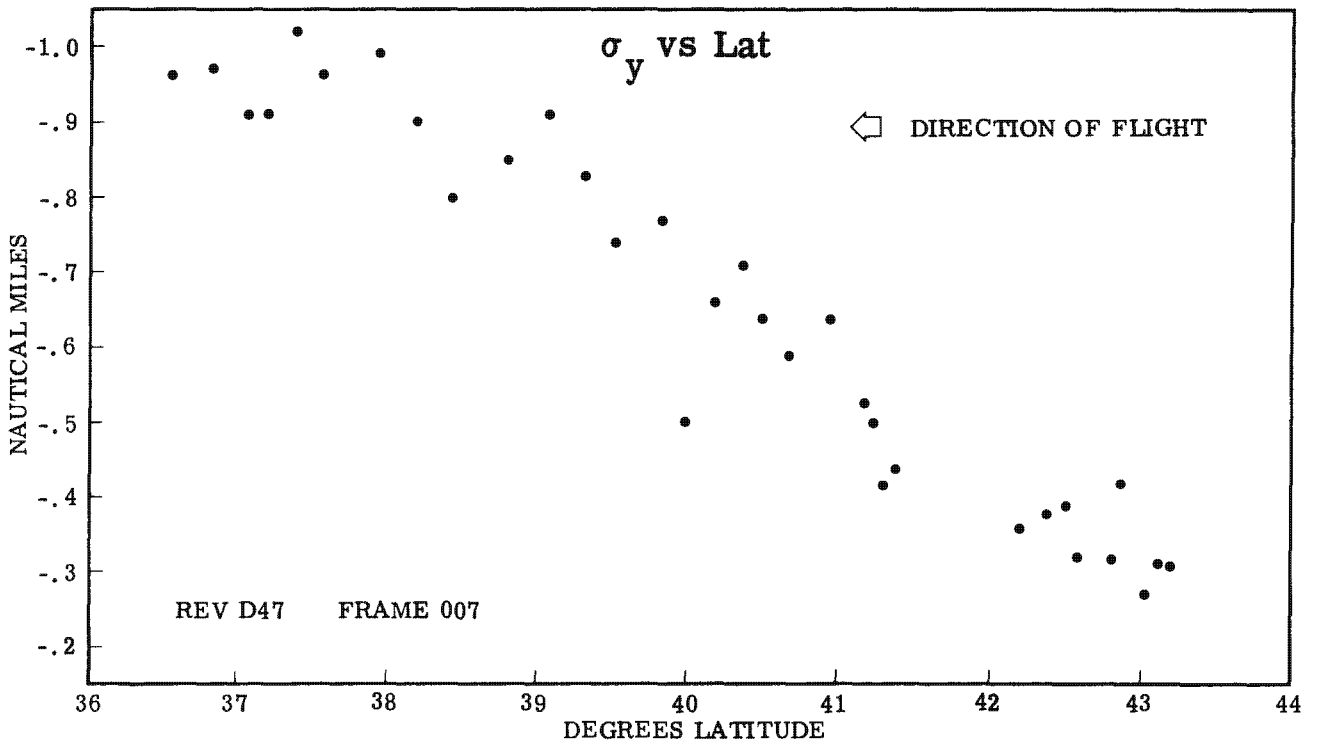
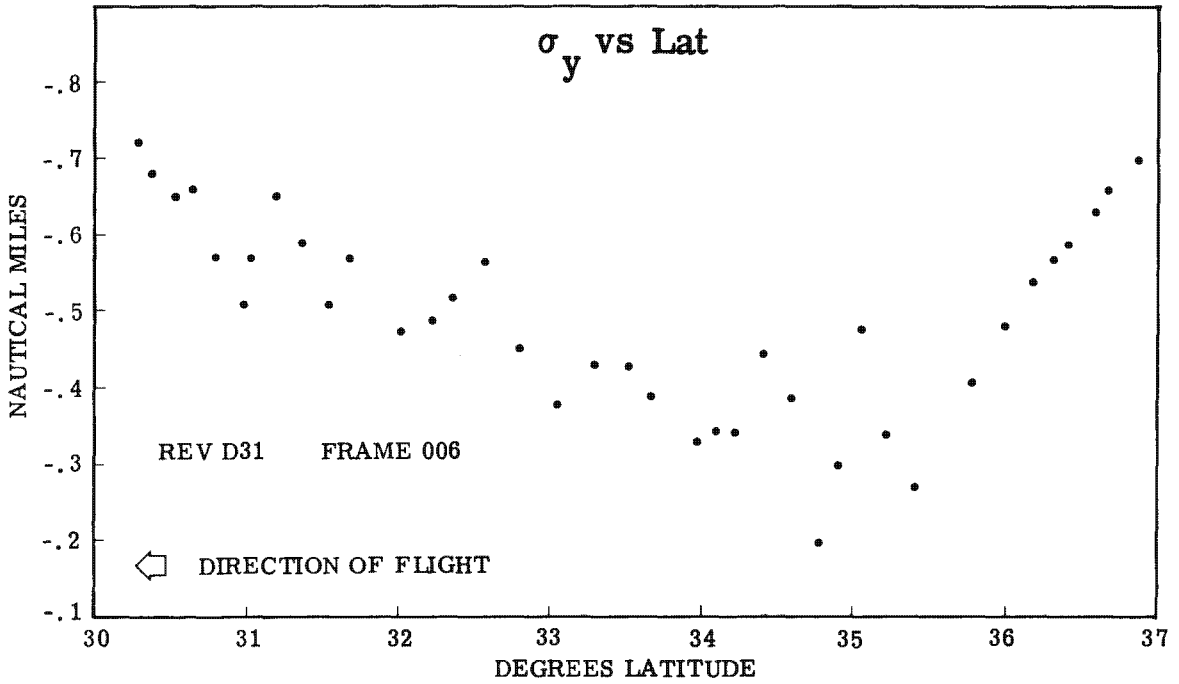


ILLUSTRATION 29

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E P H E M E R I S   C O M P A R I S O N

S U M M A R Y   B Y   R E V S

MSG NR.	FIT SPAN	EPOCH REV	LOAD REV	OPS REV	BEST FIT EPHEMERIS			COMMAND EPHEMERIS			EPHEMERIS ERROR (COMMAND - BFE)		
					IN-TRACK	X-TRACK	ME	IN-TRACK	X-TRACK	ME	IN-TRACK	X-TRACK	ME
103	01-02	02	04	05	.462	-.662	.712	.038	-.632		-.424+	.030+	-.095
105	01-04	04	06	06	1.155	-1.051	1.189	.265	-.974		-.890+	.077+	-.357
				07	1.295	-.058	.928	-.030	-.059		-1.325+	-.001-	-.658
108	01-06	06	08	08	.735	-.900	.947	.056	-.825		-.679+	.075+	-.173
				09	.458	-.889	.846	-.459	-.804		-.917-	.085+	-.045
109	01-08	08	10	10	.538	-.622	.728	.252	-.599		-.286+	.023+	-.089
				11	.865	-.985	1.108	.535	-.869		-.330+	.116+	-.159
110	01-10	10	14	14	-1.097	-1.378	1.218	-1.506	-1.377		-.409-	.001+	.192
				15	-.137	-.532	.597	-.529	-.521		-.392-	.011+	.091
				16	.276	-.739	.637	.399	-.831		.123-	-.092-	.079
112	01-14	14	16	16	.276	-.739	.637	.399	-.831		.123-	-.092-	.079
				17	.164	-.752	.577	.418	-.809		.254-	-.057-	.088
				21	.599	-.481	.686	1.127	-.533		.528-	-.052-	.282
114	09-20	20	22	22	.445	-.870	.802	1.145	-.863		.700-	.007+	.279
				23	.003	-.987	.686	1.071	-.985		1.068-	.002+	.321
				24	.218	-.744	.875	1.735	-.729		1.517-	.015+	.596
116	13-23	23	25	25	-.058	-.576	.455	.225	-.538		.283-	.038+	-.001
				26	.272	-.499	.523	.685	-.462		.413-	.037+	.143
				27	-.428	-.782	.770	.114	-.740		.542+	.042+	-.076
117	15-26	26	30	31	.139	-.462	.451	1.331	-.470		1.192-	-.008-	.570
				32	-.014	-.505	.509	1.926	-.506		1.940-	-.001-	.914
118	21-32	32	33	34	-.289	-.692	.539	.444	-.721		.733-	-.029-	.065
121	23-34	34	36	37	.637	-.511	.751	1.313	-.544		.676-	-.033-	.350
122	25-36	36	38	38	.423	-.963	.835	.671	-.995		.248-	-.032-	.091
				39	.355	-.865	.715	.666	-.910		.311-	-.045-	.122
				40	.769	-.873	.955	1.160	-.893		.391-	-.020-	.180
123	29-39	39	41	41	-.059	-.834	.784	-.773	-.856		-.714-	-.022-	.175
				42	.329	-.624	.660	-.609	-.651		-.938-	-.027-	.102
				43	.505	-.993	1.058	-.698	-1.002		-1.203-	-.009-	.048
125	32-42	42	46	47	-.067	-.695	.524	.169	-.714		.236-	-.019-	.022
				48	.036	-.699	.585	.305	-.725		.269-	-.026-	.050
127	39-50	50	52	53	.592	-1.114	.890	.293	-1.073		-.299+	.041+	-.099
128	41-52	52	54	54	.113	-.838	.741	.062	-.833		-.051+	.005+	-.010
				55	-.064	-.608	.539	-.144	-.617		-.080-	-.009-	.014
				56	.474	-1.535	1.233	.075	-1.587		-.399+	-.052-	-.009
129	46-55	55	58	58	.643	-.581	.676	.109	-.618		-.534+	-.037-	-.146
				59	.642	-.628	.701	-.154	-.682		-.796+	-.054-	-.118
132	48-59	59	63	63	-.005	-.959	.744	-1.609	-.987		-1.604-	-.028-	.603
				64	-.146	-.745	.685	-2.153	-.768		-2.007-	-.023-	.956

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BCS 39739-65

PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65

APPENDIX A

GEO - POSITIONING DATA

PHOTO I.O.	COMMANDED			MAP ACCURACY	MISS DISTANCE			ATTITUDE ERRORS			EPHEMERIS ERRORS						
	POSITION				NAUTICAL MILES			DEGREES			NAUTICAL MILES						
	REV	FRAME	ROLL CRAB STEREO		+/- FEET	SIG X	SIG Y	ME	PITCH	ROLL	BEST FIT		COMMAND		COMMAND - BFE		
										IN-TRACK	X-TRACK	IN-TRACK	X-TRACK	IN-TRACK	X-TRACK		
D 5	8	-24.81	2.0	2F	1600.	.296	-.878	.658	.116		.293	-.903	-.189	-.841	-.482+	.062+	
D 5	9	19.85	2.5	2F	1500.	.098	-.622	.454	.104		.107	-.636	-.367	-.690	-.474-	-.054-	
D 5	10	36.15	2.5	2F	1280.	.864	-.677	.780	.602		.899	-.678	.412	-.792	-.487+	-.114-	
D 5	11	36.15	2.5	2A	1280.	1.710	-.002	1.210	1.050		1.752	.031	1.381	-.091	-.371+	-.122-	
D 5	10-11	STEREO PAIR			1280.	1.287	-.339	1.018	.671	-.148		1.325	-.324	.896	-.442	-.429+	-.118-
D 5	12	-15.59	2.5	2F	1300.	.195	-.769	.563	.085		.195	-.791	-.267	-.749	-.462-	.042+	
D 5	13	-15.59	2.5	2A	1300.	.602	-.425	.524	.389		.614	-.439	.247	-.399	-.367+	.040+	
D 5	12-13	STEREO PAIR			1300.	.398	-.597	.544	.248	-.364		.405	-.615	-.010	-.574	-.415+	.041+
D 5	14	-43.95	1.5	2F	1200.	-.165	-.909	.660	-.216		-.187	-.931	-.642	-.763	-.455+	.168+	
D 5	15	-43.95	1.5	2A	1200.	.160	-.845	.613	.184		.149	-.869	-.182	-.701	-.331-	.168+	
D 5	14-15	STEREO PAIR			1200.	.025	-.872	.633	.015	-.289		.009	-.895	-.373	-.727	-.382-	.168+
D 6	4	45.37	2.0	2F	900.	-.368	-1.758	1.276	.462		.442	-1.779	-.415	-1.619	-.857+	.160+	
D 6	5	45.37	2.0	2A	900.	1.898	-1.347	1.536	.886		1.791	-1.313	.860	-1.157	-.931+	.156+	
D 6	4-5	STEREO PAIR			900.	1.033	-1.553	1.412	.470	-.444		1.116	-1.547	.222	-1.389	-.894+	.158+
D 6	15	31.90	2.5	2V	1300.	1.162	-.462	.887	.787	-.207	1.194	-.468	.297	-.470	-.897+	-.002-	
D 6	16	31.90	2.5	2V	1120.	1.159	-.612	.938	.787	-.274	1.191	-.623	.316	-.633	-.875+	-.010-	
D 7	18	.00	3.5	2F	900.	.796	-.328	.610	.501		.816	-.339	-.553	-.306	-1.369+	.033+	
D 7	19	.00	3.5	2A	900.	1.251	.116	.890	.785		1.282	.120	-.026	.150	-1.308+	.030-	
D 7	18-19	STEREO PAIR			900.	1.023	-.106	.763	.637	-.071		1.049	-.108	-.289	-.076	-1.338+	.032+
D 7	20	20.56	3.5	1F	900.	1.329	.013	.941	.815		1.363	.015	.003	-.015	-1.360+	-.030+	
D 7	21	20.56	3.5	1A	900.	1.666	-.046	1.179	1.022		1.709	-.045	.448	-.079	-1.261+	-.034-	
D 7	20-21	STEREO PAIR			900.	1.482	-.014	1.056	.884	-.023		1.520	-.013	.205	-.044	-1.315+	-.031-
D 8	10	-29.06	2.0	2F	1375.	.918	-1.551	1.276	.432		.920	-1.601	.310	-1.599	-.610+	.002+	
D 8	11	-29.06	2.0	2A	1380.	.607	-1.293	1.013	.455		.605	-1.333	-.076	-1.329	-.681+	.004+	
D 8	10-11	STEREO PAIR			1378.	.745	-1.408	1.137	.448	-.750		.745	-1.452	.095	-1.449	-.650+	.003+
D 8	12	39.70	2.0	2V	600.	-.471	-.217	.377	-.310	-.086	-.479	-.232	-1.156	-.079	-.677-	.153+	
D 8	13	22.68	2.5	2A	1900.	-.508	-.602	.559	.279		-.526	-.613	-.195	-.504	-.721+	.109+	
D 8	16	11.34	3.0	1F	1100.	.998	-.812	.910	.667		1.026	-.831	.383	-.742	-.643+	.089+	
D 8	17	11.34	3.0	1A	1100.	1.507	-.790	1.203	.920		1.548	-.803	.813	-.711	-.735+	.092+	
D 8	16-17	STEREO PAIR			1100.	1.229	-.802	1.053	.772	-.510		1.263	-.818	.578	-.728	-.685+	.090+
D 9	17	24.81	2.0	1F	880.	-.470	-.917	.730	-.208		-.467	-.944	-1.326	-.836	-.859-	.108+	
D 9	18	24.81	2.0	1A	840.	.344	-1.189	.877	.132		.373	-1.210	.553	-1.098	-.926-	.112+	
D 9	17-18	STEREO PAIR			860.	-.063	-1.053	.807	-.020	-.561		-.047	-1.077	.939	-.967	-.892-	.110+
D 9	19	36.86	1.5	2V	780.	.172	-1.231	.881	.131	-.476	.208	-1.254	-.689	-1.108	-.897-	.146+	
D 9	20	32.61	2.0	2F	680.	.437	-1.112	.853	.379		.471	-1.128	.401	-.995	-.872+	.133+	
D 9	21	32.61	2.0	2A	667.	1.426	-1.570	1.504	.738		1.495	-1.577	.547	-1.440	-.948+	.137+	
D 9	20-21	STEREO PAIR			673.	.976	-1.362	1.251	.572	-.589		1.029	-1.373	.116	-1.238	-.913+	.135+
D 9	22	-24.81	2.0	2F	783.	-.009	-1.128	.804	-.086		-.022	-1.156	-.900	-1.146	-.878-	.039-	
D 9	23	-41.83	2.0	2V	700.	.784	-1.264	1.054	.496	-.476	.776	-1.313	.159	-1.352	-.935+	.035+	
D 9	24	-14.88	2.5	2F	900.	.680	-.522	.612	.400		.695	-.539	-.207	-.504	-.902+	.034+	
D 9	25	-14.88	2.5	2A	900.	.681	-.519	.610	.439		.696	-.535	.252	-.501	-.948+	.035+	
D 9	24-25	STEREO PAIR			900.	.681	-.520	.611	.417	-.330		.695	-.537	.229	-.502	-.924+	.035+
D 9	26	14.18	2.5	2A	1300.	.975	-.562	.804	.580		1.003	-.570	.047	-.476	-.956+	.094+	
D 9	27	28.36	2.5	1F	700.	-.010	.040	.039	-.010		-.011	.041	.923	.165	-.912-	.124-	
D 9	28	28.36	2.5	1A	700.	.274	-.206	.244	.155		.283	-.208	.707	-.082	-.990-	.126+	
D 9	27-28	STEREO PAIR			700.	.132	-.083	.174	.091	-.049		.136	-.084	-.815	.041	-.951-	.125+

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GEO - POSITIONING DATA

PHOTO I.D.	COMMANDED POSITION	MAP ACCURACY	MISS DISTANCE NAUTICAL MILES	ATTITUDE ERRORS DEGREES	EPHEMERIS ERRORS											
					BEST FIT		COMMAND		COMMAND - BFE							
REV	FRAME	ROLL	CRAB	STEREO	+/- FEET	SIG X	SIG Y	ME	PITCH	ROLL	IN-TRACK	X-TRACK	IN-TRACK	X-TRACK	IN-TRACK	X-TRACK
D 10	3	19.14	1.5	2F	760.	.447	-1.533	1.130	.359		.506	-1.533	.298	-1.467	-.208+	.066+
D 10	4	19.14	1.0	2A	760.	1.224	-.620	.972	.663		1.274	-.590	1.017	-.523	-.257+	.067+
D 10	3-4	STEREO PAIR			760.	.855	-1.077	1.054	.446	-.557	.890	-1.063	.657	-.997	-.233+	.066+
D 10	7	21.27	2.0	1F	800.	1.045	-1.506	1.298	.713		1.109	-1.501	.881	-1.425	-.228+	.076+
D 10	8	21.27	2.0	1A	800.	1.842	-.781	1.417	1.027		1.909	-.749	1.641	-.671	-.268+	.078+
D 10	7-8	STEREO PAIR			800.	1.444	-1.144	1.359	.796	-.583	1.509	-1.126	1.261	-1.049	-.248+	.077+
D 10	9	5.67	2.0	1F	700.	.816	-.811	.815	.490		.842	-.819	.601	-.770	-.241+	.049+
D 10	10	5.67	2.0	1A	700.	1.231	-.628	.978	.708		1.267	-.632	.985	-.582	-.282+	.050+
D 10	9-10	STEREO PAIR			700.	1.023	-.720	.900	.594	-.441	1.054	-.726	.793	-.676	-.261+	.050+
D 10	11	-5.67	1.5	1F	600.	.430	-.897	.705	.239		.438	-.913	.219	-.879	-.219+	.034+
D 10	12	-5.67	1.5	1A	600.	.814	-.428	.655	.481		.834	-.439	.541	-.404	-.293+	.035+
D 10	11-12	STEREO PAIR			600.	.622	-.663	.680	.364	-.412	.636	-.677	.380	-.642	-.256+	.035+
D 10	13	-36.15	1.5	2A	700.	.127	-.492	.262	.110		.115	-.506	-.175	-.527	-.290-	-.021-
D 10	14	-29.06	1.5	2F	620.	.024	-.647	.460	-.043		.010	-.659	-.251	-.665	-.261-	-.006-
D 10	15	-29.06	2.0	2A	620.	.260	-.061	.209	.158		.265	-.068	-.033	-.072	-.298+	-.004-
D 10	14-15	STEREO PAIR			620.	-.142	-.354	.357	.099	-.165	.138	-.365	-.142	-.369	-.280-	-.004-
D 10	16	-19.14	1.5	2F	600.	-.217	-.729	.542	.169		-.231	-.741	-.492	-.725	-.261-	.016+
D 10	17	-19.14	2.0	2A	525.	.006	-.149	.115	.010		.005	-.152	-.302	-.136	-.307-	.016+
D 10	16-17	STEREO PAIR			554.	-.080	-.372	.348	-.032	-.208	-.086	-.379	-.375	-.363	-.289-	.016+
D 10	18	-21.27	2.0	2V	600.	-.202	-.619	.463	-.135	-.357	-.215	-.630	-.502	-.615	-.287-	.015+
D 10	19	-24.10	2.0	2V	700.	.038	-.646	.460	.039	-.357	.030	-.662	-.256	-.652	-.286-	.010+
D 10	20	-29.77	2.0	2V	700.	.129	-.673	.492	.076	-.340	.121	-.691	-.161	-.692	-.282-	-.001-
D 10	21	-42.54	2.0	2A	600.	.596	-.474	.542	.403		.599	-.501	.262	-.530	-.337+	-.029-
D 10	22	-36.86	1.5	2V	700.	.301	-.384	.352	.191	-.169	.301	-.400	.001	-.414	-.300+	-.014-
D 10	23	-33.32	2.0	2V	600.	.072	-.382	.280	.043	-.179	.068	-.393	-.232	-.400	-.300-	-.007-
D 10	24	-42.54	2.0	2A	822.	.435	-.912	.727	.350		.425	-.945	.085	-.972	-.340+	-.027-
D 10	25	-29.06	2.0	1F	800.	.476	-.827	.677	.220		.477	-.855	.184	-.852	-.293+	.003+
D 10	26	-29.06	2.0	1A	800.	.649	-.001	.461	.402		.665	-.008	.324	-.003	-.341+	.005+
D 10	25-26	STEREO PAIR			800.	.562	-.413	.579	.365	-.207	-.571	-.431	.254	-.428	-.317+	.003+
D 10	27	14.88	2.5	1F	600.	.983	-.510	.785	.636		1.011	-.516	.729	-.438	-.282+	.078+
D 10	28	14.88	2.5	1A	600.	1.266	-.035	.897	.786		1.298	-.026	.946	.053	-.352+	.079+
D 10	27-28	STEREO PAIR			600.	1.111	-.294	.838	.669	-.180	1.141	-.293	.828	-.215	-.313+	.078+
D 11	3	32.61	2.0	1F	900.	.197	-1.733	1.239	.302		.267	-1.747	-.003	-1.648	-.270+	.099+
D 11	4	32.61	2.0	1A	900.	1.099	-1.573	1.360	.528		1.188	-1.557	.848	-1.454	-.340+	.103+
D 11	3-4	STEREO PAIR			900.	.648	-1.653	1.301	.360	-.672	.727	-1.653	.422	-1.552	-.305+	.101+
D 11	5	36.86	2.0	1F	700.	.533	-1.275	.979	.461		.599	-1.271	.323	-1.157	-.276+	.114+
D 11	6	36.86	2.0	1A	700.	1.611	-.501	1.194	.900		1.672	-.444	1.330	-.329	-.342+	.115+
D 11	5-6	STEREO PAIR			700.	1.072	-.888	1.092	.512	-.326	1.136	-.444	.826	-.745	-.310+	.114+
D 11	7	43.24	1.0	2A	1100.	1.838	-.660	1.382	1.016		1.915	-.583	1.559	-.450	-.356+	.133+
D 11	8	36.86	2.5	1F	1000.	-.295	-1.138	.832	-.041		-.277	-1.172	-.601	-1.048	-.324-	.124+
D 11	9	36.86	2.5	1A	1000.	.450	-.145	.345	.263		.465	-.138	.063	-.013	-.402+	.125+
D 11	8-9	STEREO PAIR			1000.	.119	-.586	.611	-.011	-.233	.135	-.598	-.232	-.473	-.367-	.125+
D 14	1	-39.70	2.5	1F	500.	-1.104	-1.470	1.301	-.848		-1.142	-1.512	-1.532	-1.511	-.390-	.001+
D 14	2	-39.70	2.5	1A	500.	-1.019	-1.225	1.129	-.522		-1.052	-1.247	-1.481	-1.245	-.429-	.002+
D 14	1-2	STEREO PAIR			500.	-1.061	-1.347	1.218	-.598	-.554	-1.097	-1.378	-1.506	-1.377	-.409-	.001+
D 15	1	-32.61	2.0	1F	700.	-.377	-1.200	.891	-.341		-.406	-1.222	-.769	-1.223	-.363-	.001-
D 15	2	-32.61	2.0	1A	700.	-.249	-.772	.576	-.099		-.267	-.787	-.649	-.787	-.382-	.000+
D 15	1-2	STEREO PAIR			700.	-.313	-.986	.750	-.176	-.474	-.337	-1.005	-.709	-1.005	-.372-	.000+
D 15	3	-43.24	1.5	1F	500.	-.114	-.676	.486	-.156		-.134	-.690	-.488	-.719	-.354-	.029-
D 15	4	-43.24	1.5	1A	500.	.270	-.992	.728	.262		.254	-1.023	-.167	-1.049	-.421+	.026-
D 15	3-4	STEREO PAIR			500.	.078	-.834	.619	-.001	-.271	.060	-.856	-.328	-.884	-.388-	.028-
D 15	5	-38.99	2.0	1F	760.	.151	-.421	.320	.045		.147	-.435	-.221	-.448	-.368-	.013-
D 15	6	-38.99	2.0	1A	760.	.502	.336	.428	.282		.520	.336	.099	.324	-.421+	.012+
D 15	5-6	STEREO PAIR			760.	.327	-.043	.378	.233	-.008	.334	-.049	-.061	-.062	-.395+	.013-
D 15	7	-1.41	2.5	2V	400.	-.244	-.306	.282	-.164	-.206	-.249	-.315	-.649	-.268	-.400-	.047+
D 15	8	22.68	2.5	2V	500.	-.253	-.502	.400	-.166	-.272	-.254	-.518	-.652	-.435	-.398-	.083+
D 15	9	-.70	3.0	2V	400.	-.940	-.788	.887	-.635	-.532	-.962	-.812	-1.366	-.761	-.404-	.051+
D 16	1	-6.38	.0	2V	400.	.445	-.806	.692	.289	-.520	.454	-.825	.592	-.911	-.138-	.086-
D 16	2	-11.34	2.5	2V	400.	.055	-.864	.665	.033	-.563	.053	-.885	.181	-.976	-.128-	.091-
D 16	3	-14.88	2.5	2V	600.	.286	-.754	.572	.191	-.484	.289	-.775	.398	-.871	-.109-	.096-
D 16	4	-24.81	2.5	2F	600.	.703	-.952	.842	.377		.713	-.984	.847	-1.088	-.134-	.104-
D 16	5	-25.52	2.5	2A	600.	.849	-.200	.620	.550		.869	-.211	.974	-.314	-.105-	.103-
D 16	4-5	STEREO PAIR			600.	.768	-.618	.752	.490	-.350	.782	-.640	.904	-.744	-.122-	.104-
D 16	6	-19.85	2.5	2V	400.	.311	-.177	.287	.210	-.111	.318	-.184	.416	-.284	-.098-	.100-
D 16	7	-.70	3.0	2V	400.	.475	-.355	.420	.322	-.240	.487	-.365	.591	-.450	-.104-	.085-
D 20	9	-31.90	1.5	2F	1850.	.021	-.932	.666	-.072		.004	-.951	.280	-1.009	.276-	.058-
D 20	10	-31.90	1.5	2A	1933.	.376	-.467	.432	.259		.377	-.484	.602	-.542	-.225-	.058-
D 20	9-10	STEREO PAIR			1886.	.173	-.732	.577	.102	-.347	.164	-.752	.418	-.809	.254-	.057-

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PERFORMANCE EVALUATION TEAM REPORT NO. 4022/65

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GEO-POSITIONING DATA

PHOTO		COMMANDED			MAP	MISS DISTANCE			ATTITUDE ERRORS			EPHEMERIS ERRORS				
I.D.		POSITION			ACCURACY	NAUTICAL MILES			DEGREES			NAUTICAL MILES				
REV	FRAME	ROLL	CRA8	STERED	+/- FEET	SIG X	SIG Y	ME	PITCH	ROLL	BEST FIT		COMMAND		COMMAND - BFE	
											IN-TRACK	X-TRACK	IN-TRACK	X-TRACK	IN-TRACK	X-TRACK
O 21	5	-1.41	2.5	2F	1800.	.510	-1.76	.398	.311		.523	-.180	1.069	-.201	-.546-	-.021-
O 21	6	-1.41	2.5	2A	1800.	1.068	.390	.823	.656		1.094	.400	1.599	.379	-.505-	-.021+
D 21	5-6			STEREO PAIR	1800.	.829	.147	.675	.505	.096	.849	.151	1.372	.130	-.523-	-.021+
O 21	7	-33.32	2.0	2V	1800.	.252	-1.003	.737	.154	-.486	.240	-1.031	.765	-1.092	-.525-	-.061-
D 21	8	-29.77	2.5	2A	1800.	.742	-.311	.579	.477		.756	-.330	1.251	-.386	-.495-	-.056-
D 21	9	-41.12	1.5	2F	1900.	.687	-.627	.670	.341		.689	-.658	1.242	-.738	-.553-	-.080-
D 22	9	17.72	2.5	1F	1100.	.654	-.772	.718	.448		.679	-.781	1.395	-.762	-.716-	.019+
D 22	10	17.72	2.5	1A	1100.	1.413	-.707	1.119	.831		1.456	-.707	2.153	-.688	-.697-	.019+
D 22	9-10			STEREO PAIR	1100.	1.033	-.740	.940	.623	-.437	1.068	-.744	1.774	-.725	-.706-	.019+
O 22	18	35.45	2.5	2V	800.	.036	-.509	.362	.028	-.207	.042	-.524	.736	-.524	-.694-	.000+
D 22	19	12.76	3.0	2A	900.	.385	-.636	.530	.217		.396	-.652	1.081	-.646	-.685-	.006+
D 22	20	-20.56	3.0	2V	800.	-.403	-.809	.641	-.275	-.489	-.416	-.831	.287	-.817	-.703+	.014+
D 22	21	20.56	3.0	2A	1640.	.276	-1.067	.784	.104		.288	-1.094	.988	-1.093	-.700-	.001+
O 22	23	43.24	2.0	2V	1000.	.642	-1.468	1.139	.437	-.449	.672	-1.509	1.373	-1.521	-.701-	-.012-
D 23	11	31.19	2.5	2V	1450.	-.005	-.958	.686	.003	-.434	.003	-.987	1.071	-.985	1.068-	.002+
D 24	9	-38.99	1.5	2F	1100.	-.273	-1.552	1.116	-.342		-.318	-1.579	1.218	-1.567	1.536-	.012+
D 24	10	-39.70	2.0	2A	1100.	-.216	-.807	.596	-.066		-.240	-.821	1.278	-.808	1.518-	.013+
D 24	9-10			STEREO PAIR	1100.	-.249	-1.233	.930	-.129	-.499	-.284	-1.255	1.244	-1.242	1.528-	.013+
D 24	12	-9.21	2.5	1F	1100.	.944	-.582	.787	.570		.966	-.599	2.486	-.584	1.520-	.015+
D 24	13	-9.21	2.5	1A	1100.	.996	-.054	.709	.619		1.021	-.059	2.518	-.044	1.497-	.015+
D 24	12-13			STEREO PAIR	1100.	.970	-.318	.749	.603	-.208	.993	-.329	2.502	-.314	1.509-	.015+
O 24	14	32.61	2.5	2F	1700.	.374	-.842	.654	.318		.397	-.857	1.916	-.843	1.519-	.014+
O 24	15	33.32	2.0	2A	1600.	.852	-1.236	1.070	.424		.893	-1.253	2.409	-1.239	1.516-	.014+
D 24	14-15			STEREO PAIR	1642.	.653	-1.072	.920	.386	-.466	.687	-1.088	2.203	-1.074	1.516-	.014+
D 24	16	32.61	2.0	2V	1350.	.096	-.519	.377	.069		.106	-.531	1.626	-.518	1.520-	.013+
D 24	17	-40.41	2.0	2V	1400.	-1.620	-.215	1.157	-1.090	-.084	-1.663	-.196	-1.152	-1.181	1.511+	.015+
D 25	12	12.76	2.0	2F	900.	.255	-.420	.351	.174		.266	-.426	.548	-.386	-.282-	.040+
D 25	15	-33.32	1.5	2F	1000.	-.316	-.447	.393	-.238		-.332	-.452	-.028	-.427	-.304-	.025+
O 25	16	-33.32	1.5	2A	1000.	.163	-.269	.228	.120		.162	-.278	.442	-.253	-.280-	.025+
O 25	15-16			STEREO PAIR	1000.	-.037	-.343	.308	-.014	-.142	-.044	-.350	.246	-.326	-.290-	.024+
O 25	19	-40.41	2.5	2A	980.	.053	-.573	.410	.086		.044	-.589	.321	-.562	-.277-	.027+
O 25	20	7.09	2.5	2V	1400.	-.090	-.690	.512	-.058	-.454	-.089	-.718	.200	-.665	-.289-	.053+
O 25	21	1.41	3.0	1F	980.	-.360	-.857	.658	-.214		-.367	-.881	-.086	-.829	-.281-	.052+
D 25	22	1.41	3.0	1A	980.	-.056	-.763	.542	-.047		-.055	-.783	.214	-.730	-.269-	.053+
D 25	21-22			STEREO PAIR	980.	-.208	-.810	.602	-.133	-.546	-.211	-.832	.064	-.780	-.275+	.052+
D 26	6	-16.30	1.0	2V	800.	-.272	-.186	.248	-.170	-.104	-.282	-.184	.146	-.173	-.428+	.011+
O 26	7	32.61	1.5	1F	600.	.227	-.600	.460	.200		.259	-.596	.688	-.563	-.429-	.033+
O 26	8	32.61	1.5	1A	600.	.681	-.250	.518	.381		.709	-.224	1.132	-.189	-.423-	.035+
D 26	7-8			STEREO PAIR	600.	.454	-.425	.490	.228	-.170	.484	-.411	.910	-.377	-.426-	.034+
O 26	9	5.67	1.5	2V	600.	-.342	-.758	.594	-.104	-.473	-.346	-.772	.078	-.746	-.424+	.026+
D 26	10	-1.41	2.0	1F	600.	-.177	-.550	.412	-.104		-.182	-.558	.250	-.534	-.432-	.024+
D 26	11	-1.41	2.0	1A	600.	.324	-.532	.444	.192		.333	-.542	.759	-.516	-.426-	.026+
O 26	10-11			STEREO PAIR	600.	.074	-.541	.428	.043	-.344	.076	-.550	.504	-.525	-.428-	.025+
D 26	12	-35.45	1.0	2F	960.	-.311	-.989	.736	-.293		-.350	-.996	.071	-.984	-.421-	.012+
O 26	13	-35.45	1.0	2A	960.	.132	-.768	.554	.135		.113	-.788	.530	-.776	-.417-	.012+
O 26	12-13			STEREO PAIR	960.	-.089	-.879	.651	-.055	-.349	-.118	-.892	.301	-.880	-.419-	.012+
O 26	14	-4.183	1.5	2A	680.	.158	-.671	.489	.150		.138	-.691	.548	-.681	-.410-	.010+
D 26	15	-38.99	1.5	2V	1100.	-.001	-.649	.467	-.013	-.259	-.021	-.661	.391	-.648	-.412-	.013+
D 26	16	-4.25	2.0	2F	700.	.310	-.387	.355	.185		.317	-.395	.740	-.362	-.423-	.033+
D 26	17	-4.25	2.0	2A	700.	.472	-.076	.343	.288		.483	-.078	.888	-.045	-.405-	.033+
O 26	16-17			STEREO PAIR	700.	.383	-.245	.350	.233	-.159	.392	-.251	.807	-.218	-.415-	.033+
D 26	18	5.67	2.5	2F	700.	-.387	-.280	.495	-.222		-.394	-.594	.018	-.555	-.412+	.039+
O 26	19	21.27	2.5	2F	700.	-.242	-.514	.405	-.109		-.241	-.529	.183	-.482	-.424+	.047+
O 26	20	21.27	2.5	2A	700.	.134	-.086	.120	.077		-.139	-.086	.527	-.039	-.388-	.047+
D 26	19-20			STEREO PAIR	700.	-.054	-.300	.298	-.046	-.173	-.051	-.308	.355	-.260	-.406-	.048+
O 26	21	26.23	2.0	2A	700.	.357	-.215	.318	.205		.369	-.214	.764	-.163	-.395-	.051+
O 26	22	41.12	1.5	2V	900.	.488	-.679	.594	.332	-.230	.518	-.682	.931	-.619	-.413-	.063+
O 26	23	26.94	2.0	2A	700.	1.047	-.566	.845	.611		1.082	-.565	1.475	-.510	-.396-	.055+
O 26	24	23.39	2.0	2A	960.	1.054	-.560	.844	.620		1.087	-.560	1.497	-.506	-.410-	.054+
O 26	25	13.47	2.5	2V	1200.	.414	-.849	.697	.281	-.523	.430	-.868	.852	-.816	-.422-	.052+
O 26	26	18.43	2.5	2A	1100.	.812	-.373	.635	.487		.836	-.375	1.242	-.320	-.406-	.055+
O 26	27	24.10	2.0	2A	1160.	.505	-.339	.441	.293		.521	-.342	.919	-.283	-.398-	.059+
D 26	28	27.65	2.0	2A	1033.	.648	-.645	.564	.373		.670	-.447	1.061	-.386	-.391-	.061+

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TOP SECRET - GAMBIT

BCS 39739-65  
PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65

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GEO-POSITIONING DATA

PHOTO I.D.	COMMANDED			MAP ACCURACY	MISS DISTANCE			ATTITUDE ERRORS			EPHEMERIS ERRORS							
	REV	FRAME	ROLL		CRAB	STERED	+/- FEET	SIG X	SIG Y	ME	PITCH	ROLL	BEST FIT		COMMAND		COMMAND - BFE	
													IN-TRACK	X-TRACK	IN-TRACK	X-TRACK	IN-TRACK	X-TRACK
O 27	3		33.32	1.5	1F	1000.	-7.48	-1.031	.904									
O 27	4		33.32	1.5	1A	1000.	.369	-1.088	.813									
O 27	3+ 4		STEREO PAIR			1000.	-.190	-1.059	.860									
O 27	10		-24.81	2.0	1F	880.	-.540	-.268	.430		-4.46							
D 27	11		-24.81	2.0	1A	880.	-.248	-.081	.194									
O 27	10-11		STEREO PAIR			880.	-.394	-1.174	.334									
D 27	12		26.23	2.5	1F	1500.	-.977	-.534	.788									
D 27	13		26.23	2.5	1A	1500.	-.257	-.398	.338									
D 27	12-13		STEREO PAIR			1500.	-.617	-.466	.606									
O 27	14		-37.57	2.5	1F	2000.	-.605	-1.477	1.136									
D 27	15		-37.57	2.5	1A	2000.	-.464	-1.526	1.137									
O 27	14-15		STEREO PAIR			2000.	-.535	-1.501	1.137									
D 31	1		-26.94	2.5	2V	700.	.274	-1.115	.815									
O 31	2		-15.59	2.5	1F	700.	.158	-.301	.248									
O 31	3		-15.59	2.5	1A	700.	.684	-.337	.545									
O 31	2+ 3		STEREO PAIR			700.	.433	-.320	.429									
O 31	4		-2.83	3.0	2F	400.	.329	-1.176	.269									
D 31	5		-2.83	3.0	2A	400.	.709	-.277	.539									
D 31	4+ 5		STEREO PAIR			400.	.519	-.227	.426									
O 31	6		-2.12	3.0	2V	400.	-.160	-.499	.400									
D 32	2		-.70	2.5	2V	400.	-.033	-.284	.247									
D 32	3		-2.12	2.5	2V	400.	-.179	-.954	.688									
D 32	4		-20.56	2.5	1F	400.	-.754	-1.005	.891									
D 32	5		-20.56	2.5	1A	400.	-.207	-1.175	.847									
D 32	4+ 5		STEREO PAIR			400.	-.480	-1.090	.869									
D 32	6		-4.96	3.0	2V	400.	.291	-.266	.286									
D 32	7		19.85	2.5	2A	400.	.915	-.405	.710									
O 34	2		2.12	3.5	2F	1500.	-.498	-.725	.622									
D 34	3		2.12	3.5	2A	1500.	-.068	-.619	.441									
D 34	2+ 3		STEREO PAIR			1500.	-.283	-.672	.539									
O 37	5		-39.70	1.5	2V	1680.	-.354	-1.050	.788									
O 37	6		4.25	2.0	2V	1500.	.557	-.439	.503									
O 37	7		-29.06	2.0	2F	1600.	.375	-.958	.731									
O 37	8		-29.06	2.0	2A	1600.	.872	-.261	.647									
D 37	7+ 8		STEREO PAIR			1600.	.623	-.609	.690									
D 37	9		34.03	2.0	2F	1400.	.787	-.370	.618									
D 37	10		34.03	2.0	2A	1400.	1.529	.046	1.084									
O 37	9+10		STEREO PAIR			1400.	1.158	-.162	.882									
D 38	13		-24.10	2.5	2F	1740.	.228	-1.177	.853									
D 38	14		-24.10	2.5	2A	1600.	.583	-1.349	1.042									
D 38	13+14		STEREO PAIR			1654.	.446	-1.283	.974									
O 38	15		-44.66	2.0	2V	700.	-.266	-1.367	.989									
O 38	21		-11.34	3.0	2V	1460.	.045	-.635	.466									
O 38	27		-38.28	3.0	2F	1700.	.297	-1.023	.755									
O 38	28		-38.99	3.5	2A	1700.	.754	-.560	.665									
O 38	27+28		STEREO PAIR			1700.	.526	-.792	.712									
D 38	29		-31.90	3.5	2A	1800.	1.144	-.305	.839									
D 39	13		-38.99	2.5	2V	1600.	.652	-.482	.583									
D 39	14		-44.66	2.0	2V	1400.	.120	-1.122	.805									
D 40	3		36.15	1.5	2F	1600.	.603	-.360	.505									
O 40	4		36.86	1.0	2A	1600.	1.525	-1.380	1.457									
D 40	3+ 4		STEREO PAIR			1600.	1.141	-.955	1.160									
D 40	7		-19.14	2.0	1F	900.	.297	-1.443	1.044									
O 40	12		17.72	2.5	1F	2000.	.756	-.800	.788									
O 40	13		17.72	2.5	1A	2000.	1.070	-.330	.801									
D 40	12-13		STEREO PAIR			2000.	.913	-.565	.795									
D 40	14		-26.94	2.0	2F	1025.	-.313	-.969	.732									
O 40	15		-26.94	2.5	2A	800.	.049	-.333	.241									
D 40	14+15		STEREO PAIR			929.	-.158	-.696	.576									
D 40	17		26.94	2.5	2V	1375.	1.002	-.883	.968									

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G E D - P O S I T I O N I N G   D A T A

PHOTO I.D.		COMMANDED POSITION			MAP ACCURACY	MISS DISTANCE NAUTICAL MILES		ATTITUDE ERRORS DEGREES		EPHEMERIS ERRORS NAUTICAL MILES						
REV	FRAME	ROLL	CRAB	STEREO	+/- FEET	SIG X	SIG Y	ME	PITCH	BEST FIT		COMMAND		COMMAND - BFF		
										IN-TRACK	X-TRACK	IN-TRACK	X-TRACK	IN-TRACK	X-TRACK	
0 41	24	-12.05	2.5	1F	1000.	-504	-1.265	.969	-.360	-.521	-1.297	-1.222	-1.319	-.701-	-.022-	
0 41	25	-12.05	2.5	1A	1000.	-.281	-1.066	.787	-.149	-.292	-1.093	-1.005	-1.114	-.713-	-.021-	
0 41	24-25	STEREO PAIR			1000.	-.382	-1.157	.874	-.239	-.751	-.396	-1.185	-1.207	-.707-	-.022-	
0 41	26	-27.65	2.5	2A	900.	-.622	-1.153	.928	-.316	-.649	-1.176	-1.379	-1.205	-.730-	-.029-	
0 41	27	-34.74	2.5	2A	1000.	-.586	-1.202	.949	-.271	-.615	-1.226	-1.333	-1.258	-.718-	-.032-	
0 41	27	-25.52	2.5	2F	1500.	.838	-.964	.905	.461	.853	-.999	.151	-1.019	-.702+	-.020-	
0 41	30	-25.52	3.0	2A	1500.	1.075	-.373	.805	.705	1.100	-.390	.388	-4.409	-.712+	-.019-	
0 41	29-30	STEREO PAIR			1500.	.957	-.669	.856	.607	-.376	.976	-.693	.269	-.713	-.707+	-.020-
0 41	31	-21.27	3.0	3F	2000.	-.306	-.027	.224	-.195	-.313	-.027	-1.027	-.041	-.714-	-.014-	
0 41	32	-24.10	3.0	3A	2000.	.016	-.395	.286	.033	.015	-.406	-.709	-.420	-.724-	-.014-	
0 41	31-32	STEREO PAIR			2000.	-.145	-.211	.257	-.101	-.117	-.149	-.217	-.868	-.232	-.719-	-.015-
0 42	5	38.99	1.5	2F	700.	1.161	-.953	1.065	.799	1.238	-.904	.337	-.880	-.901+	.024+	
0 42	6	38.99	1.5	2A	725.	1.721	-.318	1.239	1.043	1.746	.412	.806	.435	-.940+	.023+	
0 42	5-6	STEREO PAIR			711.	1.410	-.388	1.146	.644	-.131	1.463	-.322	.545	-.299	-.918+	.023+
0 42	8	-20.56	1.5	2V	600.	-.216	-1.360	.975	-.154	-.774	-.245	-1.378	-1.169	-1.419	-.924-	-.041-
0 42	9	-34.03	1.0	1F	600.	-.059	-1.363	.952	-.174	-.100	-1.362	-1.002	-1.416	-.902-	-.054-	
0 42	10	-34.03	1.0	1A	600.	.238	-1.174	.849	.222	.210	-1.203	-.727	-1.256	-.937-	-.053+	
0 42	9-10	STEREO PAIR			600.	.073	-1.268	.907	.015	-.564	.038	-1.292	-.880	-1.345	-.918-	-.053+
0 42	11	-9.92	2.0	2F	600.	.020	-.665	.473	-.007	.016	-.676	-.884	-.706	-.900-	-.030-	
0 42	12	-9.92	2.0	2A	600.	.319	-.273	.301	.199	.325	-.280	-.619	-.309	-.944-	-.029-	
0 42	11-12	STEREO PAIR			600.	1.169	-.469	.396	.104	-.295	.171	-.479	-.751	-.508	-.922-	-.029-
0 42	13	-14.88	2.0	2V	700.	-.375	-.530	.460	-.248	-.326	-.536	-1.319	-.569	-.930-	-.033+	
0 42	14	-41.12	1.5	2V	600.	-.559	-.795	.691	-.380	-.293	-.598	-.793	-1.542	-.852	-.944-	-.059-
0 42	15	-35.45	2.0	2A	700.	.563	-.237	.437	.363	.571	-.256	-.381	-.307	-.952+	-.051-	
0 42	16	-15.59	2.0	2F	600.	.586	-.503	.549	.330	.597	-.518	-.326	-.548	-.923+	-.030-	
0 42	17	-15.59	2.0	2A	600.	.859	.055	.609	.530	.881	.050	-.065	.020	-.946+	-.030+	
0 42	16-17	STEREO PAIR			600.	.708	-.255	.576	.440	-.154	.723	-.266	.210	-.296	-.933+	-.030-
0 42	18	-6.38	2.0	2A	600.	1.053	-.368	.795	.655	1.078	-.268	.135	-.401	-.943+	-.023+	
0 42	19	-7.79	2.0	2F	600.	.442	-.754	.623	.259	.451	-.772	-.481	-.797	-.932-	-.025+	
0 42	20	-7.79	2.0	2A	600.	.684	-.569	.635	.433	.700	.584	-.246	-.607	-.946+	-.023+	
0 42	19-20	STEREO PAIR			600.	.563	-.661	.629	.348	-.434	.575	-.678	-.464	-.702	-.939+	-.024+
0 42	21	19.85	2.0	2V	660.	.411	-.428	.427	.277	-.244	.426	-.433	-.505	-.439	-.931-	-.006-
0 42	22	-8.50	2.5	2F	600.	-.188	-.760	.555	-.134	-.195	-.777	-1.143	-.800	-.948-	-.023+	
0 42	23	-8.50	2.5	2A	600.	.286	-.607	.476	.188	.292	-.623	-.665	-.645	-.957-	-.022-	
0 42	22-23	STEREO PAIR			600.	.049	-.683	.517	.029	-.449	-.048	-.700	-.904	-.723	-.952-	-.023+
0 42	24	-19.85	2.5	2V	700.	.003	-.749	.547	-.002	-.460	-.003	-.768	-.954	-.796	-.951-	-.023+
0 42	25	-26.94	2.0	1F	1000.	.030	-.746	.530	-.041	.022	-.765	-.903	-.798	-.925-	-.033+	
0 42	26	-26.94	2.0	1A	1000.	.563	-.418	.498	.380	.572	-.434	-.391	-.466	-.963+	-.032-	
0 42	25-26	STEREO PAIR			1000.	.296	-.582	.514	.186	-.301	.297	-.600	-.647	-.632	-.944-	-.032-
0 42	27	-34.03	2.5	2A	800.	.520	-.588	.560	.375	.525	-.611	-.435	-.647	-.960+	-.036-	
0 42	28	-4.25	3.0	1F	600.	.120	-.697	.502	.071	.123	-.716	-.824	-.733	-.947+	-.017-	
0 42	29	-4.25	3.0	1A	600.	.442	-.219	.351	.281	.453	-.225	-.509	-.241	-.962-	-.016-	
0 42	28-29	STEREO PAIR			600.	.281	-.458	.433	.178	-.309	.288	-.470	-.666	-.487	-.954+	-.017-
0 43	4	-13.47	1.0	1F	740.	-.745	-1.233	1.020	-.487	-.783	-1.234	-1.951	-1.272	-1.168-	-.038-	
0 43	5	-13.47	1.0	1A	740.	-.214	-.706	.524	-.109	-.231	-.712	-1.419	-.750	-1.188-	-.038-	
0 43	4-5	STEREO PAIR			740.	-.480	-.969	.810	-.280	-.563	-.507	-.974	-1.685	-1.012	-1.178-	-.038-
0 43	6	-26.94	1.0	2F	900.	-.769	-1.162	.986	-.549	-.825	-1.151	-2.007	-1.204	-1.182-	-.053+	
0 43	9	13.47	1.5	1F	1920.	-.366	-1.132	.843	-.162	-.359	-1.152	-1.526	-1.162	-1.167-	-.010-	
0 43	10	13.47	1.5	1A	1933.	.477	-1.088	.844	.248	.506	-1.098	-.715	-1.108	-1.221-	-.010-	
0 43	9-10	STEREO PAIR			1927.	.094	-1.108	.843	.057	-.671	-.113	-1.123	-1.083	-1.133	-1.196-	-.010-
0 43	13	36.86	2.0	1F	420.	.321	-.630	.502	.268	.348	-.630	-.848	-.616	-1.196-	-.014+	
0 43	14	36.86	2.0	1A	420.	1.166	.261	.849	.731	1.186	.303	-.039	.317	-1.225+	-.014+	
0 43	13-14	STEREO PAIR			420.	.743	-.184	.697	.346	-.086	.767	-.165	-.443	-.151	-1.210+	-.014+
0 43	17	39.70	2.0	2F	600.	.671	-1.497	1.161	.603	.734	-1.505	-.437	-1.486	-1.171+	-.019+	
0 43	18	39.70	2.0	2A	600.	2.185	-1.482	1.868	1.205	2.285	-1.445	1.052	-1.426	-1.233+	-.019+	
0 43	17-18	STEREO PAIR			600.	1.428	-1.490	1.555	.761	-.520	1.510	-1.475	.307	-1.457	-1.203+	-.018+
0 43	19	39.70	1.5	2A	700.	1.800	-1.411	1.620	.982	1.884	-1.391	.665	-1.377	-1.219+	-.014+	
0 43	20	-8.50	2.5	1F	1000.	.483	-1.239	.942	.281	.492	-1.271	-.729	-1.292	-1.221-	-.021-	
0 43	21	-8.50	2.5	1A	1000.	.797	-.586	.701	.509	.815	-.602	-.418	-.623	-1.233+	-.021-	
0 43	20-21	STEREO PAIR			1000.	.640	-.912	.830	.404	-.607	.654	-.937	-.574	-.958	-1.228+	-.021-
0 47	7	-7.70	2.5	2V	927.	.022	-.646	.491	.015	-.437	.023	-.663	.255	-.686	.232-	-.023-
0 47	8	-20.56	3.0	2A	900.	.054	-.658	.470	.066	.052	-.676	.294	-.689	.242-	-.013-	
0 47	9	-5.67	3.0	2V	900.	-.321	-.360	.347	-.219	-.245	-.329	-.370	-.088	-.391	.241+	-.021-
0 47	10	-2.68	3.0	2F	900.	-.394	-.925	.715	-.308	-.409	-.951	-.180	-.962	.229+	-.011-	
0 47	11	-2.68	3.0	2A	900.	.053	-.703	.503	.070	-.409	.050	-.722	.733	.252-	-.011-	
0 47	10-11	STEREO PAIR			900.	-.171	-.814	.618	-.106	-.474	-.179	-.836	.061	-.847	.240+	-.011-

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GEO-POSITIONING DATA

PHOTO		COMMANDED			MAP	MISS DISTANCE			ATTITUDE ERRORS			EPHEMERIS ERRORS				
I.D.		POSITION			ACCURACY	NAUTICAL MILES			DEGREES			NAUTICAL MILES				
REV	FRAME	ROLL	CRAB	STEREO	+/- FEET	SIG X	SIG Y	ME	PITCH	ROLL	BEST FIT		COMMAND		COMMAND - BFE	
											IN-TRACK	X-TRACK	IN-TRACK	X-TRACK	IN-TRACK	X-TRACK
0 48	6	7.79	2.5	2F	1000.	.208	-1.104	.795	.170		.218	-1.130	.482	-1.158	.264-	-.028-
0 48	7	7.79	2.5	2A	1000.	.658	-.646	.654	.397		.678	-.659	.954	-.688	.276-	-.029-
0 48	6-7	STEREO PAIR			1000.	.433	-.875	.728	.266	-.573	.448	-.894	.718	-.923	.270-	-.029-
0 48	8	-1.41	2.5	1F	1000.	.094	-1.032	.733	.061		.097	-1.059	.353	-1.083	.256-	-.024-
0 48	9	-1.41	2.5	1A	1000.	.256	-.637	.488	.162		.263	-.653	.533	-.677	.270-	-.024-
0 48	8-9	STEREO PAIR			1000.	.175	-.834	.623	.110	-.571	.180	-.856	.443	-.880	.263-	-.024-
0 48	10	1.41	2.5	2A	1000.	-.117	-.315	.248	-.077		-.120	-.323	.156	-.348	.276-	-.025-
0 48	11	3.54	2.5	2V	1000.	-.643	-.568	.608	-.440	-.385	-.658	-.584	-.384	-.610	.274+	-.026-
0 48	12	.70	3.0	2V	1000.	-.143	-.434	.328	-.098	-.296	-.146	-.446	.125	-.470	.271+	-.024-
0 53	8	-29.06	3.0	1F	1600.	.524	-1.313	1.002	.216		.538	-1.356	.218	-1.315	-.320+	.041+
0 53	9	-29.06	3.0	1A	1600.	.628	-.865	.762	.422		.646	-.879	.368	-.838	-.278+	.041+
0 53	8-9	STEREO PAIR			1600.	.576	-1.089	.890	.341	-.541	.592	-1.114	.293	-1.073	-.299+	.041+
0 54	11	24.10	2.0	1F	1200.	-.031	-.763	.548	.042		-.021	-.780	-.076	-.547	-.055-	.003+
0 54	12	24.10	2.0	1A	1200.	.888	-.391	.689	.532		.915	-.387	.850	-.385	-.065+	.002+
0 54	11-12	STEREO PAIR			1200.	.470	-.560	.629	.270	-.323	.489	-.566	.429	-.564	-.060+	.002+
0 54	13	4.25	2.5	2V	1020.	.603	-.440	.529	.407	-.293	.619	-.449	.550	-.447	-.069+	.002+
0 54	14	-19.85	2.5	2F	1200.	.359	-1.249	.923	.158		.358	-1.282	.299	-1.278	-.059+	.004+
0 54	15	-19.85	2.5	2A	1200.	.678	-.849	.774	.464		.688	-.875	.627	-.871	-.061+	.004+
0 54	14-15	STEREO PAIR			1200.	.501	-1.072	.860	.309	-.653	.505	-1.101	.445	-1.098	-.060+	.003+
0 54	16	20.56	2.0	2V	1300.	.908	.087	.645	.614	.056	.929	.098	.863	.098	-.066+	.000+
0 54	20	-3.54	3.0	1F	1500.	-.159	-.690	.506	-.102		-.163	-.708	-.213	-.707	-.050-	.001+
0 54	21	-3.54	3.0	1A	1467.	.208	-.663	.498	.131		.213	-.681	.160	-.679	-.053+	.002+
0 54	20-21	STEREO PAIR			1482.	.041	-.675	.501	.025	-.459	.042	-.693	-.009	-.692	-.051+	.001+
0 54	22	-.70	2.5	2V	900.	-.335	-.702	.556	-.228	-.479	-.343	-.722	-.395	-.721	-.052-	.001+
0 54	23	-26.23	2.5	2F	880.	-.528	-1.121	.878	-.417		-.550	-1.149	-.617	-1.142	-.067-	.007+
0 54	24	-26.23	2.5	2A	880.	-.297	-1.305	.947	-.113		-.314	-1.337	-.367	-1.331	-.053-	.006+
0 54	23-24	STEREO PAIR			880.	-.413	-1.213	.913	-.273	-.685	-.432	-1.243	-.492	-1.237	-.060+	.006+
0 54	39	-34.03	3.0	2F	1900.	-.423	-1.306	.974	-.338		-.432	-1.349	-.466	-1.331	-.034-	.018+
0 54	40	-34.03	3.0	2A	1900.	-.227	-1.227	.886	-.037		-.231	-1.242	-.229	-1.226	.002+	.016+
0 54	39-40	STEREO PAIR			1900.	-.325	-1.266	.931	-.175	-.557	-.332	-1.294	-.348	-1.278	-.016-	.016+
0 55	17	17.01	3.0	1F	1467.	-.455	-.627	.557	-.248		-.464	-.650	-.552	-.658	-.088-	-.008-
0 55	18	17.01	3.0	1A	1520.	-.104	-.035	.114	-.068		-.106	-.036	-.184	-.045	-.078-	-.009-
0 55	17-18	STEREO PAIR			1491.	-.296	-.358	.418	-.199	-.219	-.301	-.369	-.384	-.378	-.083-	-.009-
0 55	19	-8.50	3.5	1F	1300.	.258	-.735	.552	.152		.264	-.760	.175	-.760	-.089+	.000+
0 55	20	-8.50	3.5	1A	1300.	.665	-.669	.668	.428		.681	-.687	.616	-.687	-.065+	.000+
0 55	19-20	STEREO PAIR			1300.	.462	-.702	.613	.291	-.467	.473	-.723	.396	-.723	-.077+	.000+
0 55	21	24.81	3.0	2F	957.	-.475	-.663	.582	-.239		-.484	-.689	-.572	-.706	-.088-	-.017-
0 55	22	24.81	3.0	2A	940.	-.030	-.771	.551	-.078		-.027	-.791	-.095	-.809	-.068-	-.018-
0 55	21-22	STEREO PAIR			950.	-.290	-.708	.569	-.157	-.367	-.294	-.732	-.373	-.749	-.079-	-.017-
0 56	9	23.39	2.0	2F	700.	.832	-1.287	1.085	.620		.877	-1.294	.540	-1.267	-.337+	.027+
0 56	10	23.39	2.0	2A	700.	1.597	-1.362	1.485	.905		1.661	-1.361	1.258	-1.333	-.403+	.028+
0 56	9-10	STEREO PAIR			700.	1.215	-1.324	1.301	.741	-.705	1.269	-1.328	.899	-1.300	-.370+	.028+
0 56	12	-27.65	2.0	2F	1333.	-.171	-1.977	1.407	-.272		-.205	-2.016	-.559	-2.093	-.354-	-.077-
0 56	13	-27.65	2.0	2A	1340.	-.013	-2.139	1.516	.126		-.045	-2.187	-.458	-2.263	-.413-	-.076-
0 56	12-13	STEREO PAIR			1336.	-.099	-2.050	1.458	-.103	-1.087	-.133	-2.094	-.513	-2.171	-.380-	-.077-
0 56	24	-42.54	2.5	2V	1200.	.246	-1.246	.904	.155	-.472	.235	-1.284	-.209	-1.391	-.444+	-.107-
0 56	25	-41.83	2.5	2V	1300.	.458	-.933	.740	.302	-.375	.457	-.965	.008	-1.065	-.449+	-.100-
0 57	21	-31.19	3.0	1F	1100.	.470	-.729	.619	.234		.477	-.756	-.153	-.825	-.630+	-.069-
0 57	22	-31.19	3.0	1A	1100.	.935	-.400	.723	.618		.956	-.416	.268	-.481	-.688+	-.065-
0 57	21-22	STEREO PAIR			1100.	.703	-.565	.673	.435	-.281	.717	-.585	.058	-.652	-.659+	-.067-
0 58	4	-1.41	1.5	1F	900.	.387	-.303	.355	.233		.397	-.307	-.070	-.347	-.467+	-.040-
0 58	5	-1.41	1.5	1A	900.	.852	-.016	.606	.518		.873	-.017	.348	-.056	-.525+	-.039-
0 58	4-5	STEREO PAIR			900.	.619	-.160	.496	.375	-.103	.635	-.163	.139	-.202	-.496+	-.039-
0 58	6	.00	1.5	2V	1900.	.315	-.257	.306	.206	-.167	.324	-.260	-.177	-.295	-.501+	-.035-
0 58	9	9.92	2.0	1F	800.	.224	-.461	.366	.156		.233	-.468	-.272	-.480	-.505-	-.012-
0 58	10	9.92	2.0	1A	800.	.874	-.513	.719	.526		.899	-.516	.344	-.528	-.555+	-.012-
0 58	9-10	STEREO PAIR			800.	.564	-.488	.578	.348	-.315	.582	-.493	.051	-.505	-.531+	-.012-

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~~TOP SECRET~~ - GAMBIT

BCS 39739-65  
PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65

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G E O - P O S I T I O N I N G O A T A

REV	PHOTO I.D.	COMMANDED POSITION			MAP ACCURACY	MISS DISTANCE NAUTICAL MILES			ATTITUDE ERRORS DEGREES			EPHEMERIS ERRORS NAUTICAL MILES						
		FRAME	ROLL	CRAB		STERED	+/- FEET	SIG X	SIG Y	ME	PITCH	ROLL	BEST FIT		COMMAND		COMMAND - BFE	
													IN-TRACK	X-TRACK	IN-TRACK	X-TRACK	IN-TRACK	X-TRACK
D 58	11	-10.63	2.0	1F	600.	.315	-.735	.567	.173			.319	-.751	-.193	-.801	-.512+	-.050-	
D 58	12	-10.63	2.0	1A	600.	.671	-.937	.816	.437			.683	-.960	-.114	-1.008	-.569+	-.048-	
D 58	11-12	STEREO PAIR			600.	.493	-.836	.702	.299	-.539		.501	-.855	-.039	-.904	-.540+	-.049-	
D 58	13	-7.09	2.0	1F	700.	.419	-.420	.421	.254			.429	-.430	-.084	-.472	-.513+	-.042-	
D 58	14	-7.09	2.0	1A	700.	.918	-.022	.650	.576			.940	-.025	.368	-.065	-.572+	-.040-	
D 58	13-14	STEREO PAIR			700.	.669	-.221	.548	.420	-.143		.685	-.228	.142	-.269	-.543+	-.041-	
D 58	15	-11.34	2.5	1F	700.	.553	-.867	.734	.322			.562	-.889	.036	-.935	-.526+	-.046-	
D 58	16	-11.34	2.5	1A	700.	1.093	-.918	1.014	.707			1.116	-.944	.545	-.989	-.571+	-.045-	
D 58	15-16	STEREO PAIR			700.	.798	-.890	.872	.493	-.578		.814	-.914	.267	-.960	-.547+	-.046-	
D 58	17	-17.01	2.5	1F	1020.	.711	-.974	.858	.401			.722	-1.002	.189	-1.058	-.533+	-.056-	
D 58	18	-17.01	2.5	1A	1020.	1.018	-1.000	1.014	.680			1.037	-1.031	.441	-1.085	-.596+	-.054-	
D 58	17-18	STEREO PAIR			1020.	.864	-.987	.939	.533	-.618		.879	-1.016	.315	-1.072	-.566+	-.056-	
D 59	3	-26.23	2.0	2V	620.	.089	-.899	.644	.050	-.498		.077	-.918	-.705	-.999	-.782-	-.081-	
D 59	4	-27.65	2.0	2V	600.	.386	-.681	.558	.250	-.368		.384	-.702	-.406	-.787	-.790-	-.085-	
D 59	5	.00	2.0	2V	600.	.706	-.232	.530	.473	-.155		.723	-.236	-.060	-.265	-.783+	-.029-	
D 59	6	-2.83	2.0	2A	600.	1.100	-.669	.912	.691			1.126	-.685	.309	-.719	-.817+	-.034-	
D 63	3	17.01	2.0	2V	1000.	-.106	-.263	.209	-.070	-.161		-.106	-.270	-1.682	-.263	-1.576-	.007+	
D 63	4	16.30	2.5	2V	600.	-.339	-1.186	.875	-.225	-.725		-.339	-1.219	-1.929	-1.213	-1.590-	.006+	
D 63	5	-36.86	2.5	2F	400.	.091	-1.283	.910	-.075			.084	-1.326	-1.519	-1.395	-1.603-	-.069-	
D 63	6	-36.86	2.5	2A	400.	.406	-.948	.731	.339			.409	-.975	-1.241	-1.041	-1.650-	-.066-	
D 63	5-6	STEREO PAIR			400.	.249	-1.116	.825	.153	-.478		.246	-1.149	-1.380	-1.217	-1.626-	-.068-	
D 64	2	-8.50	1.5	2F	1000.	.196	-1.381	.989	.083			.190	-1.398	-1.747	-1.424	-1.937-	-.026-	
D 64	3	-9.21	2.0	2A	2000.	.009	-.729	.517	.017			.006	-.743	-1.998	-.769	-2.004-	-.026-	
D 64	2-3	STEREO PAIR			1500.	1.103	-1.055	.790	.063	-.683		.098	-1.073	-1.872	-1.099	-1.970-	-.026-	
D 64	4	-17.72	2.0	1F	1200.	.039	-1.227	.871	-.037			.031	-1.255	-1.969	-1.295	-2.000-	-.040-	
D 64	5	-17.72	2.0	1A	1200.	.367	-.868	.668	.265			.370	-.891	-1.667	-.929	-2.037-	-.038-	
D 64	4-5	STEREO PAIR			1200.	.203	-1.047	.776	.127	-.651		.200	-1.073	-1.818	-1.112	-2.018-	-.039-	
D 64	6	2.83	3.0	2F	1000.	-.848	-.292	.635	-.535			-.868	-.301	-2.869	-.313	-2.001-	-.012-	
D 64	7	2.83	3.0	2A	1000.	-.261	.121	.208	-.165			-.268	.124	-2.313	.113	-2.045-	-.011+	
D 64	6-7	STEREO PAIR			1000.	-.555	-.085	.473	-.357	-.064		-.568	-.088	-2.591	-.099	-2.023-	-.011-	
D 64	8-9	STEREO PAIR			600.	-.570	-.570	.570	-.387	-.389		-.580	-.591	-2.617	-.603	-2.037-	-.112-	

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BCS 39739-65

PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65

COMMAND INFORMATION

ACC. NO.	SYSTEM TIME	BURST TIME	COMMAND		POSITION		BEST EPHEMERIS				FILM VELOCITY		MIR-ROR	CRAB	ROLL	
			LAT	LONG	LAT	LONG	LAT	LONG	LAT	LONG	COMMAND	ACTUAL				PDS.
	SEC	SEC	DEG	MIN	DEG	MIN	DEG	MIN	DEG	MIN	DEG	MIN				
REV D13																
1	53931.28	7.5	30	55S	60	23W	30	54.81S	60	23.00W	2.7440	2.742	1	3.0	-8.508	
2	53944.18	7.5					31	47.28S	60	33.19W	2.7168	2.715	1	3.0	-8.508	
3	53957.58	6.3	32	42S	60	44W	32	41.75S	60	43.92W	2.9128	2.911	2	3.0	0.000	
4	54035.98	7.1	38	00S	61	50W	37	59.75S	61	49.91W	2.5850	2.584	1	3.0	-10.635	
5	54049.68	7.1					38	55.19S	62	2.10W	2.5850	2.586	1	3.0	-10.635	
REV D14																
1	58504.37	5.5	20	50N	74	01W	20	50.57N	74	1.16W	2.7714	2.770	1	2.5	-39.704	
2	58517.67	5.5					19	55.65N	74	10.53W	2.8271	2.828	1	2.5	-39.704	
REV D15																
1	63446.67	5.1	47	17N	90	43W	47	17.12N	90	43.34W	2.9128	2.910	1	2.0	-32.614	
2	63459.57	5.1					46	24.36N	90	57.52W	2.9713	2.972	1	2.0	-32.614	
3	63472.37	5.5	45	32N	91	11W	45	31.97N	91	11.25W	2.5089	2.509	1	1.5	-43.249	
4	63486.87	5.5					44	32.60N	91	26.42W	2.5594	2.556	1	1.5	-43.249	
5	63522.17	5.5	42	08N	92	01W	42	7.91N	92	1.80W	2.7168	2.713	1	2.0	-38.995	
6	63535.47	5.5					41	13.36N	92	14.61W	2.7714	2.772	1	2.0	-38.995	
7	63564.67	7.1	39	13N	92	42W	39	13.46N	92	41.85W	3.7355	3.734	2	2.5	-1.418	
8	63581.57	7.0	38	04N	92	57W	38	4.02N	92	57.11W	3.4155	3.415	2	2.5	22.688	
9	63612.47	100.1	35	57N	93	24W	35	56.98N	93	24.14W	3.7728	3.770	2	3.0	-7.709	
REV D16																
1	68691.67	24.8	52	56N	111	14W	52	55.06N	111	14.75W	3.3817	3.382	2	0.0	-6.381	
2	68747.47	100.1	49	08N	112	24W	49	7.53N	112	24.79W	3.6985	3.698	2	2.5	-11.344	
3	68861.27	7.0	41	22N	114	25W	41	21.61N	114	25.08W	3.6985	3.697	2	2.5	-14.889	
4	68912.87	7.5	37	50N	115	12W	37	49.70N	115	12.61W	3.2497	3.248	2	2.5	-28.815	
5	68923.37	8.6	37	07N	115	21W	37	6.54N	115	21.88W	3.2822	3.282	2	2.5	-25.524	
6	68954.97	6.3	34	57N	115	49W	34	56.57N	115	49.04W	3.6256	3.624	2	2.5	-19.852	
7	68969.07	7.1	33	59N	116	00W	33	58.54N	116	.83W	3.7728	3.770	2	3.0	-7.709	
REV D17																
1	75031.67	5.5	16	32S	146	40W	16	32.04S	146	39.71W	2.4841	2.486	1	2.5	34.032	
2	75047.37	5.5					17	36.41S	146	50.45W	2.4841	2.486	1	2.5	34.032	
REV D19																
1	84494.67	5.5	64	56N	172	32W	64	57.51N	172	30.85W	3.0920	3.093	1	1.5	-17.016	
2	84506.77	5.5					64	9.06N	172	59.80W	3.1229	3.121	1	1.5	-17.016	
REV D20																
1	3334.57	5.9	70	42N	169	43E	70	42.03N	169	43.62E	2.6633	2.659	1	1.5	30.487	
2	3348.27	5.9					69	48.17N	168	52.80E	2.6633	2.662	1	1.5	30.487	
3	3468.87	5.1	61	46N	163	31E	61	47.04N	163	31.26E	2.5594	2.557	2	1.0	-43.249	
4	3495.87	5.9	59	58N	162	39E	59	58.15N	162	39.33E	3.2175	3.219	1	2.0	11.344	
5	3507.17	5.9					59	12.49N	162	19.04E	3.2175	3.219	1	2.0	11.344	
6	3518.97	5.1	58	24N	161	59E	58	24.75N	161	58.65E	3.3150	3.319	2	2.0	3.545	
7	3538.87	6.3	57	04N	161	26E	57	4.14N	161	25.98E	2.9713	2.969	1	2.0	25.524	
8	3551.77	6.3					56	11.82N	161	5.84E	2.9419	2.945	1	2.0	25.524	
9	3592.27	5.7	53	27N	160	07E	53	27.22N	160	7.24E	2.8840	2.889	2	1.5	-31.905	
10	3604.77	12.9	52	36N	159	50E	52	36.33N	159	50.39E	2.9128	2.913	2	1.5	-31.905	
REV D21																
1	8654.97	5.9	71	07N	147	56E	71	6.52N	147	55.95E	3.1229	3.123	1	1.0	-2.127	
2	8666.27	5.9					70	22.19N	147	12.13E	3.1541	3.156	1	1.0	-2.127	
3	8814.07	5.9	60	32N	140	43E	60	32.23N	140	42.81E	2.4841	2.485	1	2.0	43.958	
4	8830.17	5.9					59	27.21N	140	13.28E	2.4841	2.487	1	2.0	43.958	
5	8951.97	7.0	51	13N	137	12E	51	12.54N	137	11.61E	3.4155	3.415	2	2.5	-1.418	
6	8962.47	9.5	50	30N	136	59E	50	29.71N	136	58.52E	3.4155	3.416	2	2.5	-1.418	
7	8989.97	10.3	48	37N	136	26E	48	37.43N	136	25.66E	3.1229	3.119	?	2.0	-33.323	
8	9003.57	5.3	47	42N	136	10E	47	41.84N	136	10.11E	3.0614	3.058	2	2.5	-29.778	
9	9020.67	6.2	46	31N	135	51E	46	31.08N	135	50.94E	2.6108	2.615	2	1.5	-41.122	
10	9049.67	5.5	44	32N	135	20E	44	32.35N	135	20.18E	3.4496	3.453	2	2.5	4.963	
11	9060.77	8.4	43	48N	135	09E	43	47.69N	135	9.02E	3.4496	3.449	2	2.5	4.963	
12	9657.77	5.9	2	47N	127	37E	2	46.65N	127	36.76E	2.9713	2.972	1	3.0	24.815	
13	9670.27	5.9					1	55.07N	127	28.64E	2.9128	2.917	1	3.0	24.815	
14	9689.27	5.3	0	37N	127	17E	0	36.70N	127	16.31E	3.3150	3.318	2	3.5	.709	
15	9737.67	6.0	2	43S	126	45E	2	42.86S	126	44.91E	2.4841	2.488	2	2.5	41.831	
16	9754.57	5.8	3	52S	126	34E	3	52.50S	126	33.93E	2.4841	2.488	2	2.5	41.831	
17	9765.87	6.3	4	39S	126	27E	4	39.05S	126	26.57E	2.5594	2.564	1	3.0	34.741	
18	9779.97	6.3					5	37.12S	126	17.39E	2.5089	2.507	1	3.0	34.741	

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COMMAND INFORMATION

ACC. NO.	SYSTEM TIME	BURST TIME	COMMAND		POSITION		BEST EPHEMERIS				FILM VELOCITY	IN./SEC.	MIR-ROR	CRAB	ROLL
			LAT	LONG	LAT	LONG	LAT	LONG	LAT	LONG					
	SEC	SEC	DEG	MIN	DEG	MIN	DEG	MIN	DEG	MIN	COMMAND	ACTUAL	POS.	DEG.	ANGLE
REV D22															
1	13958.77	6.7	72	35N	127	22E	72	35.49N	127	22.30E	2.8271	2.827	2	1.5	23.397
2	13971.67	7.0	71	45N	126	25E	71	45.29N	126	24.76E	2.8554	2.852	2	1.0	24.106
3	14008.07	5.9	69	22N	124	06E	69	22.40N	124	5.45E	3.1541	3.152	1	1.5	4.254
4	14019.37	5.9					68	37.73N	123	27.87E	3.1857	3.187	1	1.5	4.254
5	14037.27	6.3	67	26N	122	33E	67	26.73N	122	32.82E	3.0311	3.033	1	1.5	17.725
6	14048.97	6.3					66	40.18N	121	59.47E	3.0311	3.029	1	1.5	17.725
7	14162.97	7.1	59	02N	117	50E	59	2.19N	117	50.18E	2.4841	2.487	1	2.0	40.413
8	14177.87	7.1					58	1.89N	117	24.79E	2.4841	2.488	1	2.0	40.413
9	14275.07	6.7	51	26N	115	04E	51	26.85N	115	3.90E	3.2175	3.216	1	2.5	17.725
10	14285.97	6.7					50	42.40N	114	50.21E	3.1857	3.186	1	2.5	17.725
11	14432.17	5.9	40	43N	112	13E	40	43.96N	112	13.11E	2.9713	2.969	2	2.0	30.487
12	14444.67	9.1	39	52N	112	02E	39	52.64N	112	1.43E	2.9419	2.937	2	2.0	31.196
13	14461.97	5.9	38	41N	111	46E	38	41.58N	111	45.60E	3.1541	3.153	2	2.5	24.106
14	14473.27	9.1	37	55N	111	36E	37	55.14N	111	35.46E	3.1541	3.152	2	2.5	24.106
15	14497.37	5.3	36	15N	111	15E	36	16.05N	111	14.34E	3.7355	3.731	2	2.5	6.381
16	14548.17	5.3	32	46N	110	32E	32	46.99N	110	31.78E	2.7714	2.766	2	2.0	-44.667
17	14593.37	5.3	29	40N	109	56E	29	40.77N	109	55.81E	3.6619	3.659	2	3.0	-19.143
18	14654.07	5.3	25	30N	109	10E	25	30.46N	109	9.76E	2.9713	2.970	2	2.5	35.450
19	14673.27	5.8	24	11N	108	56E	24	11.24N	108	55.65E	3.4155	3.413	2	3.0	12.762
20	14693.97	5.3	22	45N	108	41E	22	45.81N	108	40.65E	3.6256	3.624	2	3.0	-20.561
21	14714.57	5.1	21	20N	108	26E	21	20.78N	108	25.93E	3.2175	3.216	2	3.0	-20.561
22	14733.17	5.3	20	03N	108	13E	20	3.99N	108	12.79E	2.5850	2.581	2	2.0	43.958
23	14753.67	6.7	18	39N	107	59E	18	39.34N	107	58.47E	2.6369	2.633	2	2.0	43.249
24	15058.87	6.7	2	21S	104	37E	2	20.49S	104	36.37E	3.2497	3.249	1	3.5	5.672
25	15069.77	6.7					3	5.41S	104	29.30E	3.1857	3.183	1	3.5	5.672
26	15091.97	7.2	4	37S	104	15E	4	36.87S	104	14.87E	2.6899	2.688	2	2.5	30.487
REV D23															
1	19219.87	5.0	76	45N	111	44E	76	45.33N	111	43.81E	2.7714	2.776	2	.5	-32.614
2	19250.57	8.9	74	50N	108	15E	74	49.74N	108	14.40E	2.5340	2.536	2	1.5	33.323
3	19264.67	9.3	73	56N	106	55E	73	55.79N	106	54.19E	2.5594	0.000	2	.5	33.323
4	19294.87	5.0	71	59N	104	28E	71	58.90N	104	27.59E	3.1857	3.191	2	1.0	-19.143
5	19323.27	6.7	56	46N	94	43E	56	46.02N	94	42.40E	2.5089	2.511	2	1.0	-41.122
6	19337.77	8.7	55	47N	94	20E	55	47.17N	94	20.12E	2.5594	2.559	2	1.5	-41.831
7	19749.07	6.3	41	24N	90	10E	41	24.06N	90	10.30E	3.4155	3.419	1	2.5	-14.180
8	19759.57	6.3					40	40.97N	90	.33E	3.4841	3.490	1	2.5	-14.180
9	19921.36	5.6	29	35N	87	43E	29	35.28N	87	42.69E	3.0920	3.093	2	2.5	36.741
10	19941.77	11.7	28	11N	87	27E	28	11.18N	87	26.97E	2.6899	2.690	2	2.0	43.249
11	19964.67	14.1	26	36N	87	10E	26	36.75N	87	9.65E	3.1541	3.151	2	2.5	31.196
REV D24															
1	24496.76	8.3	79	44N	97	44E	79	43.92N	97	44.00E	2.6633	2.666	3	0.0	-26.942
2	24510.46	8.3					78	56.06N	95	3.00E	2.6369	2.639	3	0.0	-30.487
3	24531.86	6.3	77	39N	91	33E	77	39.00N	91	32.39E	3.0611	3.003	1	.5	-10.635
4	24543.56	6.3					76	55.94N	89	54.21E	3.0311	3.035	1	.5	-10.635
5	24636.46	5.1	71	00N	81	13E	71	.09N	81	12.85E	2.7168	2.720	1	1.5	29.069
6	24550.56	5.1					70	4.72N	80	19.17E	2.6899	2.693	1	1.5	29.069
7	24861.16	8.6	56	00N	72	13E	55	59.89N	72	12.83E	2.8271	2.827	2	2.0	30.487
8	24874.46	10.8	55	06N	71	53E	55	5.87N	71	53.03E	2.8271	2.827	2	1.5	30.487
9	24928.16	6.8	51	27N	70	40E	51	27.28N	70	39.88E	2.6633	2.662	2	1.5	-38.995
10	24941.06	8.0	50	34N	70	24E	50	34.66N	70	23.72E	2.7168	2.721	2	2.0	-29.704
11	24968.66	5.1	48	42N	69	51E	48	41.96N	69	50.67E	3.0311	3.031	2	2.0	-29.069
12	24983.66	5.5	47	40N	69	34E	47	40.65N	69	33.50E	3.4155	3.418	1	2.5	-9.217
13	24994.56	5.5					46	56.07N	69	21.36E	3.4496	3.449	1	2.5	-9.217
14	25040.46	5.5	43	48N	68	33E	43	48.12N	68	32.86E	2.8554	2.853	2	2.5	32.614
15	25053.76	14.8	42	53N	68	20E	42	53.59N	68	19.53E	2.8271	2.829	2	2.0	33.323
16	25072.76	13.8	41	35N	68	01E	41	35.66N	68	.98E	3.0920	3.089	2	2.0	32.614
17	25111.86	6.2	38	55N	67	25E	38	55.11N	67	24.46E	2.9419	2.945	2	2.0	-40.413
18	25210.76	5.9	32	08N	66	00E	32	8.24N	66	.04E	3.5542	3.556	1	3.0	-8.508
19	25220.86	5.9					31	26.64N	65	51.94E	3.5542	3.554	1	3.0	-8.508
20	25962.56	6.7	19	30S	57	26E	19	29.66S	57	25.56E	2.9419	2.942	1	3.5	1.418
21	25974.66	6.7					20	19.18S	57	17.12E	2.9128	2.916	1	3.5	1.418



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COMMAND INFORMATION

ACC. NO.	SYSTEM TIME	BURST TIME	COMMAND LAT		POSITION LONG		BEST EPHEMERIS				FILM VELOCITY	IN./SEC.	MIR-RDR	CRAB	ROLL
			DEG	MIN	DEG	MIN	DEG	MIN	DEG	MIN					
REV D27															
1	40632.46	5.5	69	56N	13	35E	69	56.28N	13	35.55E	2.4841	2.486	1	1.5	38.995
2	40648.16	5.5					68	54.31N	12	41.40E	2.4841	2.486	1	1.5	38.995
3	40657.46	6.3	68	17N	12	11E	68	17.48N	12	11.47E	2.6108	2.611	1	1.5	33.323
4	40671.56	6.3					67	21.50N	11	28.80E	2.6108	2.611	1	1.5	33.323
5	40737.36	6.7	62	58N	8	44E	62	58.33N	8	43.67E	2.6633	2.659	1	1.0	-35.450
6	40750.66	6.7					62	4.82N	8	15.59E	2.6899	2.693	1	1.0	-35.450
7	40768.26	5.3	60	54N	7	40E	60	53.89N	7	40.59E	2.6899	2.693	2	1.0	-35.450
8	40782.56	5.5	59	56N	7	14E	59	56.16N	7	13.76E	2.8271	2.829	1	1.0	-31.196
9	40795.86	5.5					59	2.40N	6	49.98E	2.8840	2.886	1	1.0	-31.196
10	40941.26	7.1	49	11N	3	23E	49	11.05N	3	22.78E	3.1541	3.152	1	2.0	-24.815
11	40952.56	7.1					48	24.88N	3	9.60E	3.1857	3.192	1	2.0	-24.815
12	41145.16	6.3	35	14N	0	01E	35	14.75N	0	1.16E	3.1229	3.123	1	2.5	26.233
13	41156.86	6.3					34	26.60N	0	8.70W	3.0920	3.094	1	2.5	26.233
14	41211.66	5.9	30	41N	0	53W	30	40.92N	0	53.21W	2.8554	2.850	1	2.5	-37.577
15	41224.16	5.9					29	49.40N	1	3.02W	2.9128	2.911	1	2.5	-37.577
REV D29															
1	52810.66	7.1	34	07S	56	17W	34	7.30S	56	16.55W	2.6369	2.637	1	3.0	2.836
2	52824.36	7.1					35	2.85S	56	27.91W	2.6108	2.610	1	3.0	2.836
3	52855.46	6.3	37	09S	56	54W	37	8.81S	56	54.39W	2.4841	2.488	1	2.5	-29.069
4	52871.96	6.3					38	15.55S	57	8.85W	2.4841	2.487	1	2.5	-29.069
REV D31															
1	62334.15	5.0	43	10N	87	01W	43	10.43N	87	1.07W	3.4155	3.419	2	2.5	-26.942
2	62344.75	5.5	42	27N	87	11W	42	26.95N	87	11.56W	3.3817	3.383	1	2.5	-15.598
3	62355.65	5.5					41	42.24N	87	22.16W	3.4155	3.415	1	2.5	-15.598
4	62390.95	5.9	39	17N	87	55W	39	17.30N	87	55.29W	3.5190	3.519	2	3.0	-2.836
5	62401.05	5.8	38	36N	88	04W	38	35.81N	88	4.46W	3.5190	3.529	2	3.0	-2.836
6	62424.55	100.1	36	59N	88	25W	36	59.21N	88	25.33W	3.8106	3.814	2	3.0	-2.127
REV D32															
1	67526.95	24.8	52	14N	106	42W	52	13.39N	106	42.11W	3.3817	3.380	2	0.0	-6.381
2	67582.75	100.1	48	26N	107	50W	48	25.63N	107	50.52W	3.7728	3.776	2	2.5	-7.709
3	67717.15	7.0	39	15N	110	07W	39	14.80N	110	7.71W	3.8106	3.810	2	2.5	-2.127
4	67766.65	6.7	35	52N	110	51W	35	51.31N	110	51.48W	3.3817	3.386	1	2.5	-20.561
5	67777.55	6.7					35	6.47N	111	.76W	3.4155	3.416	1	2.5	-20.561
6	67809.35	7.1	32	56N	111	27W	32	55.57N	111	27.17W	3.8106	3.810	2	3.0	-4.963
7	67826.55	5.1	31	45N	111	41W	31	44.74N	111	41.07W	3.2822	3.283	2	2.5	19.852
REV D33															
1	73867.25	5.5	17	23S	142	01W	17	23.33S	142	1.48W	2.4841	2.483	1	2.5	31.905
2	73882.95	5.5					18	27.62S	142	12.27W	2.4841	2.487	1	2.5	31.905
REV D34															
1	77937.55	100.0	68	19N	143	11W	68	19.43N	143	10.66W	2.8554	2.854	2	1.5	34.032
2	78620.25	5.9	21	53N	157	52W	21	52.72N	157	52.58W	3.5190	3.519	2	3.5	2.127
3	78630.35	5.9	21	11N	158	00W	21	11.03N	157	59.76W	3.5190	3.515	2	3.5	2.127
REV D36															
1	2158.05	6.3	70	11N	174	02E	70	11.54N	174	2.70E	2.9713	2.972	1	1.0	-21.979
2	2169.75	6.3					69	25.41N	173	20.97E	3.0011	3.004	1	1.0	-21.979
3	2231.65	5.1	65	19N	170	19E	65	19.23N	170	18.82E	3.2497	3.249	1	1.5	4.963
4	2243.35	5.1					64	32.38N	169	50.11E	3.2822	3.284	1	1.5	4.963
5	2291.65	5.5	61	18N	168	05E	61	18.17N	168	5.44E	2.7991	2.800	1	2.0	30.487
6	2304.95	5.5					60	24.50N	167	39.90E	2.7991	2.800	1	2.0	30.487
7	2373.65	7.5	55	46N	165	45E	55	46.19N	165	44.77E	2.4841	2.489	3	2.0	42.540
8	2388.95	7.5					54	44.00N	165	22.25E	2.5089	2.507	3	2.0	39.704
REV D37															
1	7504.14	5.5	68	44N	150	35E	68	44.63N	150	34.79E	2.4841	2.487	1	1.5	44.667
2	7521.04	5.5					67	37.57N	149	42.17E	2.4841	2.483	1	1.5	44.667
3	7636.24	5.1	59	55N	145	15E	59	55.34N	145	14.97E	2.9128	2.912	1	1.5	-30.487
4	7649.14	5.1					59	3.17N	144	51.91E	2.9419	2.942	1	1.5	-30.487
5	7735.84	5.3	53	11N	142	39E	53	11.14N	142	38.82E	2.8554	2.849	2	1.5	-39.704
6	7765.84	5.3	51	09N	142	00E	51	8.85N	141	59.63E	3.6985	3.700	2	2.0	4.254
7	7788.64	7.1	49	35N	141	32E	49	35.78N	141	31.66E	3.0614	3.059	2	2.0	-29.069
8	7799.94	6.9	48	49N	141	18E	48	49.61N	141	18.31E	3.0920	3.092	2	2.0	-29.069
9	7830.44	6.0	46	45N	140	44E	46	44.88N	140	43.81E	2.8271	2.825	2	2.0	34.032
10	7843.74	6.0	45	50N	140	30E	45	50.44N	140	29.40E	2.7714	2.770	2	2.0	34.032
11	8588.44	6.8	5	18S	131	10E	5	17.92S	131	9.63E	2.4841	2.485	2	2.0	43.249

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COMMAND INFORMATION

ACC. NO.	SYSTEM TIME		BURST TIME	COMMAND LAT		POSITION LONG		BEST EPHEMERIS				FILM VELOCITY	IN./SEC.	MIR-ROR	CRAB	ROLL
	SEC	SEC		DEG	MIN	DEG	MIN	LAT	MIN	DEG	MIN					
REV 038																
1	12776.14	7.0	72	08N	131	40E	72	0.74N	131	40.65E	2.8840	2.885	2	.5	-24.815	
2	12788.24	7.4	71	20N	130	48E	71	21.49N	130	48.89E	2.9419	2.944	2	1.0	-25.524	
3	12852.64	5.1	67	06N	127	08E	67	7.06N	127	8.21E	2.4841	2.485	2	.5	-41.122	
4	12886.84	5.1	64	49N	125	38E	64	50.46N	125	37.87E	2.6899	2.691	1	1.5	33.323	
5	12900.94	5.1					63	53.93N	125	4.52E	2.6633	2.662	1	1.5	33.323	
6	12920.94	6.7	62	32N	124	20E	62	33.54N	124	20.49E	3.0614	3.061	1	2.0	21.270	
7	12933.04	6.7					61	44.82N	123	55.51E	3.0311	3.031	1	2.0	21.270	
8	12963.04	7.1	59	42N	122	58E	59	43.72N	122	58.23E	3.1229	3.124	1	1.5	-22.688	
9	12974.34	7.1					58	58.01N	122	38.17E	3.1857	3.189	1	1.5	-22.688	
10	12994.94	6.3	57	33N	122	03E	57	34.97N	122	3.49E	2.7168	2.717	2	1.5	-38.286	
11	13103.34	7.6	50	12N	119	31E	50	13.45N	119	31.32E	2.5089	2.510	2	1.0	-43.958	
12	13117.54	7.1	49	14N	119	14E	49	15.45N	119	14.26E	2.5594	2.556	2	2.0	-43.958	
13	13246.94	7.5	40	24N	116	59E	40	25.36N	116	58.72E	3.2497	3.248	2	2.5	-24.106	
14	13257.44	12.5	39	41N	116	49E	39	42.24N	116	48.95E	3.2822	3.284	2	2.5	-24.106	
15	13281.24	5.4	38	03N	116	28E	38	4.45N	116	27.35E	2.7440	2.743	2	2.0	-44.667	
16	13328.34	5.3	34	49N	115	47E	34	50.74N	115	46.54E	2.9128	2.913	2	2.0	-41.122	
17	13356.74	9.8	32	52N	115	23E	32	53.83N	115	23.01E	2.9419	2.940	2	2.0	-41.122	
18	13382.94	6.3	31	05N	115	02E	31	5.92N	115	1.93E	3.2497	3.247	1	2.5	-24.815	
19	13393.44	6.3					30	22.66N	114	53.64E	3.3482	3.347	1	2.5	-24.815	
20	13447.14	8.6	26	40N	114	13E	26	41.29N	114	12.44E	2.9128	2.911	2	2.5	-41.831	
21	13471.74	9.4	24	58N	113	54E	24	58.63N	113	54.17E	3.7728	3.775	2	3.0	-11.344	
22	13485.24	5.3	24	03N	113	44E	24	4.14N	113	44.28E	3.4155	3.416	2	3.0	-17.016	
23	13497.04	5.3	23	14N	113	36E	23	15.45N	113	35.72E	3.6256	3.623	2	3.0	-70.561	
24	13507.44	6.3	22	31N	113	28E	22	32.54N	113	28.23E	3.2497	3.247	2	3.0	-24.815	
25	13519.14	8.4	21	43N	113	20E	21	44.25N	113	19.86E	3.2822	3.284	2	3.0	-24.815	
26	13817.84	5.0	1	10N	110	00E	1	11.53N	109	59.61E	3.3150	3.314	2	2.0	17.725	
27	13921.54	11.3	5	57S	108	52E	5	55.57S	108	52.29E	2.5594	2.556	2	3.0	-38.286	
28	13936.14	10.6	6	57S	108	43E	6	55.63S	108	42.75E	2.5850	2.587	2	3.5	-38.995	
29	13951.34	5.1	7	59S	108	33E	7	58.13S	108	32.81E	2.7714	2.770	2	3.5	-31.905	
REV 039																
1	18067.14	5.9	74	16N	112	13E	74	16.37N	112	13.98E	3.1541	3.152	2	1.0	-6.381	
2	18078.44	9.0	73	32N	111	13E	73	32.91N	111	13.09E	3.1857	3.186	2	1.0	-6.381	
3	18316.14	7.1	57	45N	99	57E	57	45.90N	99	56.58E	3.1857	3.187	1	2.0	17.725	
4	18327.44	7.1					57	.08N	99	38.33E	3.1857	3.187	1	2.0	17.725	
5	18354.84	5.3	55	08N	98	57E	55	8.80N	98	56.58E	3.6619	3.660	2	2.0	2.836	
6	18537.54	5.9	42	41N	95	20E	42	42.39N	95	19.38E	3.1541	3.152	1	2.5	24.106	
7	18548.84	5.9					41	56.03N	95	8.33E	3.1229	3.122	1	2.5	24.106	
8	18563.34	5.8	40	55N	94	55E	40	56.52N	94	54.43E	3.3817	3.384	2	2.5	16.307	
9	18619.14	6.3	37	06N	94	04E	37	7.26N	94	3.62E	3.1541	3.150	1	2.5	26.942	
10	18630.84	6.3					36	19.14N	93	53.44E	3.0920	3.094	1	2.5	26.942	
11	18673.74	5.3	33	22N	93	17E	33	22.60N	93	17.32E	3.0011	3.000	2	2.0	-41.831	
12	18687.74	6.0	32	24N	93	06E	32	24.96N	93	5.90E	2.8840	2.882	2	2.0	-43.958	
13	18698.74	5.3	31	39N	92	57E	31	39.65N	92	57.05E	3.1229	3.124	2	2.5	-38.995	
14	18727.14	5.3	29	42N	92	35E	29	42.64N	92	34.63E	2.8554	2.852	2	2.0	-44.667	
15	18742.94	6.3	28	36N	92	23E	28	37.52N	92	22.42E	3.5190	3.521	2	3.0	-17.016	
16	18752.94	6.3	27	55N	92	15E	27	56.29N	92	14.77E	3.8487	3.846	2	3.0	-14.180	
17	18852.04	7.5	21	06N	91	02E	21	7.48N	91	2.13E	3.0311	3.032	1	3.0	27.651	
18	18863.74	7.5					20	19.19N	90	53.87E	2.9419	2.939	1	3.0	27.651	
REV 040																
1	23341.24	5.9	77	23N	95	46E	77	23.83N	95	47.21E	2.4841	2.489	1	0.0	-38.995	
2	23356.14	5.9					76	28.59N	93	48.40E	2.4841	2.489	1	0.0	-38.995	
3	23450.34	6.8	70	25N	85	30E	70	25.78N	85	29.99E	2.5340	2.533	2	1.5	36.159	
4	23464.44	7.1	69	29N	84	39E	69	30.21N	84	39.00E	2.5594	2.563	2	1.0	36.868	
5	23479.24	5.9	68	31N	83	49E	68	31.66N	83	49.69E	2.4841	2.489	1	1.5	38.995	
6	23494.14	5.9					67	32.50N	83	3.87E	2.4841	2.489	1	1.5	38.995	
7	23675.94	5.0	55	19N	76	49E	55	20.11N	76	49.29E	3.4841	3.489	2	2.0	-19.143	
8	23739.24	5.9	51	01N	75	23E	51	2.30N	75	23.30E	2.4841	2.489	1	2.0	45.376	
9	23755.34	5.9					49	56.58N	75	3.47E	2.4841	2.489	1	2.0	45.376	
10	23804.84	7.1	46	33N	74	07E	46	34.20N	74	6.66E	2.8271	2.827	1	1.5	-30.868	
11	23817.34	7.1					45	43.02N	73	53.18E	2.9128	2.911	1	1.5	-36.868	
12	23851.94	6.3	43	20N	73	17E	43	21.24N	73	17.43E	3.3150	3.315	1	2.5	-17.725	
13	23862.44	6.3					42	38.18N	73	7.00E	3.2822	3.284	1	2.5	17.725	
14	23886.44	5.1	40	59N	72	44E	40	59.69N	72	43.80E	3.1857	3.189	2	2.0	-26.942	
15	23898.14	6.6	40	11N	72	33E	40	11.65N	72	32.80E	3.2175	3.218	2	2.5	-26.942	
16	23911.64	6.3	39	15N	72	20E	39	16.20N	72	20.33E	3.5897	3.591	2	2.5	-25.524	
17	24032.74	99.1	30	57N	70	38E	30	57.93N	70	37.62E	3.3150	3.313	2	2.5	26.942	

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COMMAND INFORMATION

Table with columns: ACC. NO., SYSTEM TIME, BURST TIME, COMMAND LAT, POSITION LONG, BEST EPHEMERIS LAT/LONG, FILM VELOCITY IN./SEC., MIRROR POS., CRAB DEG., ROLL ANGLE. Data includes REV D41 and REV D42 series.

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PERFORMANCE EVALUATION TEAM  
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COMMAND INFORMATION

ACC. NO.	SYSTEM TIME		BURST TIME		COMMAND LAT		POSITION LONG		BEST EPHEMERIS LONG				FILM VELOCITY	IN./SEC.	MIR-RDR	CRAB	ROLL
	SEC	SEC	DEG	MIN	DEG	MIN	DEG	MIN	DEG	MIN	DEG	MIN					
REV D56																	
1	22088.81	5.1	78	23N	103	19E	78	23.58N	103	19.56E	2.4841	2.487	1	1.0	41.831		
2	22105.31	5.1					77	23.46N	100	47.32E	2.4841	2.487	1	1.0	41.831		
3	22197.41	5.1	71	31N	91	38E	71	31.72N	91	37.92E	2.8554	2.854	2	1.0	34.741		
4	22223.61	5.9	69	48N	89	57E	69	48.72N	89	56.74E	3.1229	3.125	1	1.0	-18.434		
5	22234.91	5.9					69	4.05N	89	17.70E	3.1541	3.152	1	1.0	-18.434		
6	22252.11	5.9	67	55N	88	23E	67	55.82N	88	22.77E	3.1229	3.124	1	1.0	-20.561		
7	22263.41	5.9					67	10.85N	87	49.29E	3.1541	3.158	1	1.0	-20.561		
8	22405.51	6.2	57	38N	82	44E	57	38.81N	82	43.75E	2.4841	2.482	2	2.0	41.122		
9	22432.21	5.5	55	50N	82	02E	55	50.40N	82	1.85E	3.1229	3.121	2	2.0	23.397		
10	22444.31	7.1	55	00N	81	44E	55	1.20N	81	43.91E	3.0920	3.091	2	2.0	73.397		
11	22474.21	5.1	52	59N	81	02E	52	59.46N	81	2.07E	3.0614	3.059	2	1.5	33.323		
12	22501.71	7.4	51	06N	80	26E	51	7.30N	80	26.30E	3.1229	3.121	2	2.0	-27.651		
13	22513.01	7.8	50	20N	80	12E	50	21.17N	80	12.28E	3.1541	3.151	2	2.0	-27.651		
14	22537.51	6.3	48	40N	79	43E	48	41.05N	79	43.05E	3.2822	3.285	1	2.5	19.143		
15	22549.21	6.3					47	53.20N	79	29.63E	3.2497	3.245	1	2.5	19.143		
16	22590.21	5.4	45	04N	78	45E	45	5.34N	78	44.97E	3.2175	3.215	2	2.0	-25.524		
17	22601.91	5.4	44	16N	78	33E	44	17.39N	78	32.84E	3.2497	3.246	2	2.0	-25.524		
18	22613.61	7.7	43	28N	78	21E	43	29.42N	78	20.95E	3.1229	3.119	2	2.0	-36.159		
19	22660.41	5.9	40	16N	77	36E	40	17.36N	77	35.62E	2.8554	2.851	1	2.0	-37.577		
20	22672.91	5.9					39	26.01N	77	24.05E	2.9128	2.912	1	2.0	-37.577		
21	22697.31	6.3	37	45N	77	02E	37	45.73N	77	2.04E	3.4841	3.487	2	2.5	9.926		
22	22719.41	5.3	36	14N	76	43E	36	14.85N	76	42.72E	2.6633	2.661	2	2.5	41.122		
23	22737.81	5.4	34	58N	76	27E	34	59.16N	76	27.02E	2.9128	2.908	2	2.5	34.741		
24	22770.61	9.1	32	43N	76	00E	32	44.14N	75	59.87E	2.8840	2.885	2	2.5	-42.540		
25	22785.21	9.0	31	43N	75	48E	31	44.02N	75	48.08E	2.9128	2.908	2	2.5	-41.831		
REV D57																	
1	27369.81	7.5	80	43N	89	31E	80	43.84N	89	33.27E	2.6108	2.609	3	1.0	32.614		
2	27383.91	7.5					79	56.50N	86	12.18E	2.7168	2.726	3	1.0	29.778		
3	27518.41	6.3	71	33N	69	29E	71	33.70N	69	29.39E	2.9419	2.937	1	1.5	-25.524		
4	27530.11	6.3					70	47.81N	68	42.04E	2.9713	2.968	1	1.5	-25.524		
5	27560.01	7.9	68	49N	66	55E	68	49.81N	66	55.11E	3.0920	3.092	3	1.0	-21.979		
6	27572.11	7.9					68	1.79N	66	16.68E	3.0614	3.058	3	1.0	-24.815		
7	27585.41	5.6	67	08N	65	37E	67	8.86N	65	37.17E	3.1229	3.117	2	1.5	-22.688		
8	27717.61	7.7	58	16N	60	49E	58	16.90N	60	48.60E	3.6256	3.624	2	2.0	-12.762		
9	27729.11	6.0	57	29N	60	30E	57	30.26N	60	29.59E	3.6619	3.659	2	2.0	-8.508		
10	27745.31	11.3	56	24N	60	04E	56	24.50N	60	3.92E	3.5542	3.550	2	2.0	14.889		
11	27761.51	5.3	55	18N	59	39E	55	18.65N	59	39.47E	3.5190	3.518	2	2.0	16.307		
12	27783.41	6.6	53	49N	59	08E	53	49.53N	59	8.13E	3.0920	3.091	2	2.0	24.815		
13	27795.51	6.2	52	59N	58	52E	53	.23N	58	51.58E	3.0920	3.091	2	2.0	24.815		
14	27811.31	5.5	51	55N	58	31E	51	55.81N	58	30.73E	3.0920	3.091	2	2.0	25.524		
15	27823.41	6.0	51	05N	58	15E	51	6.44N	58	15.29E	3.0920	3.091	2	2.0	25.524		
16	27850.91	6.7	49	13N	57	42E	49	14.11N	57	41.78E	3.4496	3.445	1	2.5	9.217		
17	27861.81	6.7					48	29.55N	57	29.04E	3.4496	3.449	1	2.5	9.217		
18	27887.41	6.3	46	44N	57	00E	46	44.81N	57	.24E	3.0920	3.092	2	2.5	-31.196		
19	27986.21	5.9	39	59N	55	21E	39	59.62N	55	20.83E	3.2175	3.213	1	2.5	-26.233		
20	27997.51	5.9					39	13.20N	55	10.43E	3.2497	3.249	1	2.5	-26.233		
21	28199.31	5.5	25	21N	52	26E	25	22.28N	52	26.19E	3.0920	3.091	1	3.0	-31.196		
22	28211.41	5.5					24	32.38N	52	17.29E	3.1229	3.119	1	3.0	-31.196		

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COMMAND INFORMATION

ACC. NO.	SYSTEM TIME	BURST TIME	COMMAND			POSITION				BEST EPHEMERIS				FILM VELOCITY IN./SEC.	MIR-ROR	CRAB	ROLL		
			LAT	DEG	MIN	DEG	MIN	DEG	MIN	DEG	MIN	LAT	DEG					MIN	DEG
	SEC	SEC	DEG	MIN	DEG	MIN	DEG	MIN	DEG	MIN	DEG	MIN	DEG	MIN	COMMAND	ACTUAL	POS.	DEG.	ANGLE
REV 058																			
1	32791.51	7.1	74	40N	51	18E	74	40.59N	51	18.97E	2.4841	2.486	1	1.0	39.704				
2	32806.41	7.1					73	43.37N	49	55.82E	2.4841	2.485	1	1.0	39.704				
3	32943.91	6.3	64	39N	41	48E	64	40.36N	41	48.82E	3.1541	3.153	2	1.0	-29.778				
4	32961.51	5.5	63	29N	41	08E	63	29.68N	41	7.92E	3.3817	3.378	1	1.5	-1.418				
5	32972.41	5.5					62	45.83N	40	44.06E	3.3817	3.378	1	1.5	-1.418				
6	32985.21	5.8	61	53N	40	17E	61	54.25N	40	17.34E	3.6619	3.658	2	1.5	0.000				
7	33018.71	6.3	59	38N	39	13E	59	38.93N	39	13.19E	3.2822	3.281	1	1.5	-17.725				
8	33029.21	6.3					58	56.43N	38	54.60E	3.3150	3.310	1	1.5	-17.725				
9	33069.11	5.9	56	14N	37	49E	56	14.56N	37	49.45E	3.3817	3.381	1	2.0	9.926				
10	33080.41	5.9					55	28.63N	37	32.39E	3.3817	3.383	1	2.0	9.926				
11	33100.31	6.3	54	07N	37	03E	54	7.66N	37	3.63E	3.4155	3.413	1	2.0	-10.635				
12	33110.81	6.3					53	24.90N	36	49.07E	3.4496	3.445	1	2.0	-10.635				
13	33143.51	7.1	51	11N	36	06E	51	11.55N	36	6.13E	3.4841	3.487	1	2.0	-7.090				
14	33153.61	7.1					50	30.32N	35	53.54E	3.4841	3.487	1	2.0	-7.090				
15	33175.21	6.3	49	01N	35	27E	49	2.07N	35	27.56E	3.4496	3.448	1	2.5	-11.344				
16	33185.71	6.3					48	19.13N	35	15.36E	3.4841	3.485	1	2.5	-11.344				
17	33225.31	5.5	45	36N	34	32E	45	37.04N	34	31.62E	3.4155	3.414	1	2.5	-17.016				
18	33236.21	5.5					44	52.38N	34	20.14E	3.4496	3.448	1	2.5	-17.016				
19	33442.81	5.5	30	42N	31	15E	30	43.19N	31	14.76E	3.3482	3.346	1	3.0	-21.270				
20	33453.71	5.5					29	58.28N	31	6.20E	3.3817	3.378	1	3.0	-21.270				
REV 059																			
1	37998.71	5.5	81	27N	48	59E	81	27.58N	49	1.17E	3.0920	3.094	1	.5	9.217				
2	38010.81	5.5					80	48.87N	45	36.86E	3.1229	3.122	1	.5	9.217				
3	38419.11	5.3	54	17N	14	56E	54	17.34N	14	56.25E	3.3817	3.382	2	2.0	-26.233				
4	38435.51	5.0	53	10N	14	33E	53	10.54N	14	33.54E	3.3482	3.346	2	2.0	-27.651				
5	38453.41	5.4	51	49N	14	07E	51	49.40N	14	7.23E	3.7355	3.733	2	2.0	0.000				
6	38465.41	19.3	51	08N	13	54E	51	8.60N	13	54.48E	3.5190	3.522	2	2.0	-2.836				
7	38487.91	5.1	49	36N	13	27E	49	36.71N	13	26.86E	3.7728	3.773	2	2.0	-4.254				
REV D63																			
1	59376.40	9.1	75	50N	57	37W	75	50.81N	57	37.61W	3.3482	3.347	2	.5	-15.598				
2	59638.00	60.1	58	34N	72	08W	58	34.27N	72	8.41W	3.6985	3.696	2	2.0	-4.254				
3	59827.30	5.1	45	42N	76	20W	45	42.43N	76	20.43W	3.5897	3.591	2	2.0	17.016				
4	59900.70	9.1	40	41N	77	34W	40	41.38N	77	33.84W	3.5897	3.591	2	2.5	16.307				
5	60147.10	7.9	23	46N	80	55W	23	46.95N	80	54.88W	2.8554	2.855	2	2.5	-36.868				
6	60159.60	7.9	22	55N	81	04W	22	55.40N	81	3.91W	2.9128	2.910	2	2.5	-36.868				
REV D64																			
1	64762.50	24.8	71	35N	85	43W	71	35.86N	85	43.26W	3.3817	3.381	2	0.0	-6.381				
2	64818.30	60.1	67	55N	89	03W	67	55.58N	89	3.00W	3.3482	3.345	2	1.5	-8.508				
3	64934.70	60.0	60	08N	93	37W	60	8.28N	93	37.68W	3.4155	3.418	2	2.0	-9.217				
4	65031.40	5.5	53	35N	96	11W	53	35.67N	96	11.33W	3.4155	3.418	1	2.0	-12.762				
5	65042.30	5.5					52	51.26N	96	26.12W	3.4496	3.448	1	2.0	-12.762				
6	65095.90	5.5	49	12N	97	33W	49	12.45N	97	33.43W	3.3817	3.384	1	2.0	-17.725				
7	65106.80	5.5					48	27.89N	97	46.15W	3.4155	3.412	1	2.0	-17.725				
8	65231.50	5.9	40	00N	99	53W	40	.83N	99	53.59W	3.5542	3.552	2	3.0	2.836				
9	65240.60	5.9	39	19N	100	03W	39	19.36N	100	2.89W	3.5190	3.521	2	3.0	2.836				
10	65339.40	5.1	32	32N	101	28W	32	32.95N	101	27.79W	3.8106	3.816	2	3.0	2.836				

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	DMIN	DMAX	BASE + FOG	(DEGREES)	RANGE	DMAX	(NM)
REV D 4							
1	.45	1.73	.13	17	3.75		96.2
2	.41	1.72	.14	17	4.02		95.8
3	.68	1.98	.22	17	4.04		95.5
4	.79	1.20	.12	24	1.44	1.53	93.0
5	.98	1.42	.22	24	1.54	1.90	92.7
6	.88	1.08	.22	35	1.21	2.52	88.9
7	.94	1.28	.22	35	1.38	2.52	88.7
REV D 5							
1	.56	1.90	.22	13	4.15	1.60	96.2
2	.56	1.87	.22	13	3.99	1.45	97.2
3			.12	27		1.68	91.8
4			.10	27		1.61	91.5
5	.42	1.21	.22	33	2.60	2.03	89.6
6	.43	1.20	.22	33	2.53	2.25	89.3
7	.43	1.42	.22	37	3.17	1.57	88.2
8	.57	1.50	.22	39	2.62	2.32	87.7
9	.34	.90	.22	42	2.46		87.1
10	.46	.94	.22	43	1.86	2.32	86.8
11	.52	1.14	.22	43	1.99	2.32	86.6
12	.29	1.16	.22	45	3.75		86.2
13	.35	1.32	.22	45	3.56		86.0
14	.36	1.30	.22	47	3.36		85.8
15	.35	1.55	.22	47	4.52	1.30	85.6
16			.12	79		1.76	89.9
17			.12	79		1.82	90.3
REV D 6							
1	.76	1.58	.11	13	2.13	1.41	96.2
2	.36	1.40	.11	13	3.56	1.32	97.2
3			.10	33		1.72	89.5
4			.20	35		1.98	89.2
5			.22	34		1.93	88.9
6	1.01	1.48	.21	47	1.60	2.06	85.6
7	.62	.99	.12	47	1.46	1.74	85.4
8			.11	51		1.34	85.0
9			.10	54		1.56	84.6
10			.10	54		1.60	84.6
11	.42	.97	.21	57	2.08	2.16	84.4
12	.32	.87	.22	59	2.56	2.32	84.3
13	.28	1.28	.21	61	4.57	2.28	84.3
14	.40	.98	.21	61	2.19	2.28	84.3
15	.81	.97	.21	63	1.17	2.01	84.4
16	.32	.82	.22	64	2.43	2.28	84.4
17	.60	.72	.22	79	1.15	2.30	88.2
18	.55	.70	.22	79	1.20	2.22	88.5
19	.49	1.72	.21	78	3.88	2.20	89.2
20	.39	1.70	.23	78	4.66	1.91	89.5
REV D 7							
1	.28	1.04	.21	9	3.63	1.39	96.2
2	.57	.97	.21	9	1.55	1.09	99.2
3			.21	14		1.80	97.3
4	.56	.97	.21	14	1.57	1.89	96.8
5			.21	26		1.78	92.1
6			.21	28		2.20	91.5
7	.60	.93	.21	28	1.44	1.78	91.2
8	.40	.52	.21	30	1.28	1.84	90.8
9			.11	31		2.00	90.2
10			.11	31		2.00	89.9
11			.11	32		1.89	89.4
12	.60	1.24	.21	36	1.93	2.39	88.6
13	.45	1.02	.11	35	1.90	2.28	88.4
14			.11	44		1.45	86.5
15			.10	44		1.60	86.3
16	.44	1.42	.21	56	3.11	2.06	84.5
17	.40	1.40	.21	56	3.30	2.00	84.5
18	.34	1.96	.21	70	7.57	1.82	85.2
19	.48	1.76	.21	70	4.14	1.78	85.3
20	.40	1.82	.21	75	5.21	1.87	86.5
21	.36	.84	.21	75	2.16	1.87	86.7
REV D 8							
1	.35	.78	.22	6	2.09	1.31	96.2
2	.36	1.06	.22	6	2.67	1.04	100.3
3			.22	8		1.80	99.6
4			.22	8		1.66	99.2
5	.65	1.50	.22	29	2.39	2.06	90.7
6	.48	1.15	.22	30	2.18	2.36	90.4
7	.60	1.24	.22	30	1.93	2.36	90.1
8			.10	35		1.66	88.8
9			.10	35		1.76	88.4
10	.69	1.46	.22	40	2.19	2.16	87.2
11	.66	1.64	.22	40	2.75	2.14	87.0
12	.52	1.82	.22	46	4.08		85.8
13	.44	1.74	.22	47	4.40		85.6
14	.66	1.32	.22	48	1.95	1.70	85.3
15	.42	1.22	.22	50	2.63		85.1
16	.34	1.44	.22	51	4.17		85.0
17	.44	1.52	.22	51	3.45		84.9

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	DMIN	DMAX	BASE + FDG	(DEGREES)	RANGE	DMAX	(NM)
REV D 9							
1	.46	.78	.22	5	1.58	1.68	96.2
2	.38	.74	.22	5	1.80	1.30	100.5
3	.94	2.14	.22	8	3.82	2.01	99.4
4	.86	1.82	.22	8	2.67	2.08	99.0
5	.42	2.22	.22	11	8.88		97.9
6	.86	1.86	.20	13	2.79	1.96	97.4
7	.76	1.76	.20	12	2.80	1.90	97.0
8			.11	14		1.26	96.5
9			.12	15		1.38	96.1
10			.11	18		1.56	94.9
11			.10	18		1.42	94.6
12	.66		.13	27	1.13	2.10	91.6
13	.40	1.03	.12	27	2.12	2.34	91.2
14			.20	30		2.38	90.4
15			.12	30		2.16	90.1
16			.10	33		1.81	89.2
17	.60	1.32	.22	34	2.09	2.24	88.9
18	.72	1.28	.20	34	1.75	2.14	88.6
19	.65	1.73	.20	36	3.07	2.42	88.3
20	.34	1.52	.20	37	4.54		88.0
21	.38	1.58	.20	37	4.21		87.8
22	.32	.78	.20	40	2.32		87.1
23	.30	1.10	.20	41	3.41	1.68	86.7
24	.26	1.42	.20	43	6.29		86.5
25	.28	1.56	.22	43	6.13	1.82	86.3
26	.27	.92	.21	44	3.54	2.00	86.0
27	.42	1.50	.20	49	3.52		85.4
28	.46	1.66	.20	49	3.86		85.2
29	.74	1.54	.20	53	2.24	1.62	84.7
30	.75	1.80	.20	65	2.96		84.5
31	.80	1.42	.20	65	1.85		84.6
32	.68	1.35	.12	66	1.87	1.25	84.8
REV D10							
1			.21	3		1.05	96.2
2			.21	3		.72	101.2
3	.37	1.58	.22	17	4.36	2.10	95.6
4	.37	1.50	.22	16	4.01	2.06	95.3
5	.58	1.20	.22	18	1.90	2.20	94.7
6	.38	.65	.22	19	1.62	2.15	94.2
7	.32	.80	.22	23	2.37	2.10	92.9
8	.32	1.48	.22	23	4.66	1.98	92.6
9	.34	1.20	.22	25	3.26		92.1
10	.42	1.35	.22	25	3.01		91.8
11	.34	1.18	.22	27	3.20	2.12	91.4
12	.39	1.21	.22	27	2.77	2.10	91.1
13	.60	1.60	.22	27	2.81	2.40	90.7
14	.47	1.10	.22	29	2.12	2.18	90.4
15	.49	1.38	.22	29	2.69	2.34	90.1
16	.40	1.45	.22	31	3.48	2.20	89.7
17	.46	1.58	.22	31	3.53	2.15	89.4
18	.45	1.52	.21	33	3.38		88.9
19	.44	1.58	.22	35	3.68		88.5
20	.50	2.20	.22	36	7.29		88.1
21	.32	1.20	.22	36	3.49		87.9
22	.44	1.50	.22	37	3.38		87.7
23	.44	1.77	.22	38	4.54		87.4
24	.37	1.26	.22	38	3.12		87.2
25	.26	1.15	.22	42	4.78		86.6
26	.29	1.32	.22	42	4.40		86.5
27	.31	1.15	.22	44	3.45		86.1
28	.33	1.12	.22	44	3.13		86.0
29	.36	1.42	.22	71	3.81		85.6
30	.48	1.50	.22	71	3.11		85.7
31	.42	.58	.22	75	1.36	2.06	86.5
32	.36	.75	.22	75	1.95	2.26	86.8
33	.36	.70	.22	76	1.84	2.24	87.5
34	.38	.80	.22	76	1.93	2.05	87.7
35	.50	.66	.22	77	1.27	2.28	88.1
36	.62	.88	.22	77	1.34	2.28	88.4
37			.22	78		2.22	88.7
38			.12	79		1.69	90.6
39			.12	79		1.65	91.1
40			.12	80		1.62	91.4
REV D11							
1	.34	2.16	.21	7	10.11	1.99	96.2
2	.41	1.99	.20	7	6.37	1.87	99.2
3	.39	.99	.21	24	2.26		92.7
4	.40	1.10	.20	24	2.45		92.3
5	.46	1.00	.20	26	1.97	2.18	91.9
6	.49	1.16	.21	26	2.15	2.22	91.5
7	.54	1.12	.21	27	1.88		91.1
8	.49	.80	.21	41	1.52	2.32	86.9
9	.54	1.13	.20	41	1.89	2.35	86.7

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	DMIN	DMAX	BASE + FDG	(DEGREES)	RANGE	DMAX	(NM)
REV D13							
1	.36	1.36	.20	59	3.58		96.2
2	.37	1.51	.20	59	4.05		107.4
3	.34	1.49	.21	58	4.40		108.0
4	.52	2.03	.21	53	5.39		112.0
5	.49	1.72	.21	53	3.88		112.7
REV 014							
1	.34	.92	.21	64	2.51	2.26	96.2
2	.40	.94	.21	64	2.11	2.30	84.6
REV D15							
1	.30	1.12	.22	39	3.47		96.2
2	.32	1.15	.22	39	3.33		86.8
3	.35	1.46	.22	41	4.12		86.6
4	.36	1.52	.22	40	4.23		86.4
5	.36	1.34	.22	44	3.51		85.9
6	.34	1.50	.22	44	4.44		85.7
7	.27	1.10	.22	46	4.19		85.4
8	.28	1.10	.22	48	3.84		85.2
9	.25	.82	.22	50	3.83	.85	84.9
9			.14	50		1.64	84.9
9			.12	50		1.38	84.9
REV 016							
1	.38	.98	.21	33	2.31	2.10	96.2
2	.48	1.09	.20	37	2.06	1.57	87.5
3	.30	1.40	.20	44	4.59		85.7
4	.30	1.20	.21	48	3.74		85.1
5	.36	1.28	.21	48	3.29		85.0
6	.41	1.44	.21	50	3.37		84.8
7	.48	1.42	.21	51	2.86		84.7
REV D17							
1	.44	1.80	.21	73	4.70	2.26	96.2
2	.33	1.89	.20	73	7.11	2.28	98.1
REV D19							
1	.51	2.16	.22	21	6.67	2.25	96.2
2	.56	2.10	.22	21	5.48	2.26	92.0
REV D20							
1	.51	2.26	.12	16	6.75		96.2
2	.58	2.22	.12	16	5.77		94.2
3	.57	.80	.13	24	1.32	1.92	91.0
4			.12	27		2.10	90.4
5			.12	27		2.02	90.1
6			.12	27		1.52	89.9
7			.10	30		1.84	89.4
8			.12	30		1.76	89.1
9	.48	1.95	.12	33	4.56	2.20	88.3
10	.75	1.20	.20	33	1.56	2.40	88.1
REV D21							
1	.43	1.74	.23	15	4.49		96.2
2	.38	1.69	.23	15	4.76		94.3
3	.49	1.31	.15	26	2.32	1.67	90.5
4	.70	.93	.15	26	1.24	2.10	90.2
5	.35	1.72	.23	35	5.46	2.07	87.6
6	.37	1.52	.22	35	4.09	2.17	87.4
7	.44	1.38	.23	37	2.98	1.10	87.0
8	.72	1.63	.23	38	2.53		86.7
9	.47	1.15	.23	39	2.22		86.5
10	.43	1.44	.23	42	3.24		86.0
11	.46	1.52	.23	42	3.31		85.9
12			.15	78		2.16	88.6
13			.14	78		1.62	88.9
14	.71	.99	.22	78	1.34	2.04	89.4
15	.46	.73	.23	80	1.49	2.36	90.7
16	.50	.80	.22	80	1.49	2.40	91.2
17	.48	1.92	.23	79	5.02	2.13	91.5
18	.47	1.97	.23	79	5.49	2.13	92.0

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	DMIN	DMAX	BASE + FOG	(DEGREES)	RANGE	DMAX	(NM)
REV D22							
1			.23	14		2.11	96.2
2			.13	14		1.85	94.8
3			.20	17		1.65	93.8
4			.21	17		1.55	93.5
5	.50	1.10	.21	19	2.00	1.90	93.1
6	.47	.88	.21	19	1.73	1.12	92.7
7	.50	2.11	.21	28	6.27	2.31	90.0
8	.64	2.22	.21	28	6.10	2.10	89.6
9	.92	1.28	.21	35	1.41	1.93	87.7
10	.56	1.19	.12	35	1.89	1.78	87.5
11	.40	1.23	.21	46	2.76	1.93	85.3
12	.41	1.35	.20	46	3.07	1.96	85.2
13	.36	.95	.21	47	2.41	2.01	85.0
14	.40	1.11	.21	47	2.47	2.08	84.9
15			.11	49		1.29	84.7
16			.11	52		1.62	84.4
17			.11	55		1.45	84.3
18	.46	1.16	.21	60	2.29	2.10	84.3
19	.36	1.33	.20	60	3.47	2.20	84.4
20	.31	1.55	.21	62	5.19	2.22	84.4
21	.36	1.65	.21	63	4.88	2.00	84.5
22	.40	1.58	.20	65	3.99	2.10	84.7
23	.40	1.45	.21	66	3.48	2.21	84.8
24	.80	.95	.21	78	1.17	2.14	90.6
25	.77	1.11	.21	78	1.40	2.22	90.9
26	.61	1.10	.13	79	1.63	2.14	91.6
REV D23							
1			.21	9		1.64	96.2
2			.13	12		1.57	96.1
3			.21	12		1.57	95.7
4	.32	1.56	.21	14	5.07		94.8
5	.47	.73	.12	29	1.41	2.09	89.2
6	.66	.90	.12	29	1.28	1.82	88.9
7	.69	1.37	.20	44	1.99	1.87	85.4
8	.66	1.70	.21	44	2.94	1.66	85.3
9	.28	1.42	.21	56	5.29	2.10	84.3
10	.33	2.36	.21	57	15.23	2.07	84.3
11	.31	1.23	.21	59	3.72	1.91	84.3
REV 024							
1			.21	6		1.10	96.2
2	.27	.67	.21	6	2.69	1.05	97.8
3			.21	8		1.54	97.2
4			.21	8		1.18	96.8
5	.46	.68	.21	16	1.35	1.95	94.2
6	.47	.76	.21	16	1.51	2.12	93.9
7	.82	1.28	.21	31	1.56	2.16	88.8
8	.76	1.43	.21	31	1.95	2.21	88.5
9	.72	1.00	.13	35	1.29	2.12	87.5
10	.30	.90	.11	34	2.48	1.75	87.3
11			.11	37		1.78	86.8
12	.86	1.68	.22	38	2.28		86.6
13	.94	1.80	.21	38	2.42		86.4
14	.30	1.00	.21	43	3.10		85.8
15	.47	1.39	.21	43	2.83		85.6
16	.39	1.09	.21	44	2.48	1.58	85.4
17	.45	1.39	.21	46	2.95		85.0
18	.35	1.45	.21	53	4.07		84.4
19	.46	1.50	.21	53	3.24		84.3
20	.36	1.35	.21	70	3.54	2.08	100.0
21	.30	1.45	.20	70	4.84	2.04	100.5



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PHOTOGRAPHIC EVALUATION

ACC. NO.	DENSITY		DENSITY	SUN ANGLE	BRIGHTNESS	CLOUD	ALTITUDE
	DMIN	DMAX	BASE + FOG	(DEGREES)	RANGE	DMAX	(NM)
REV 025							
1			.20	5		1.26	96.2
2			.20	5		1.08	97.9
3			.20	8		1.70	96.9
4			.20	8		1.58	96.6
5	.74	1.05	.20	9	1.37	1.55	96.2
6	.72	2.08	.20	16	4.43	2.25	93.8
7	.46	1.58	.20	18	3.53		92.7
8			.10	26		1.68	89.9
9			.10	26		1.48	89.6
10	.56	1.63	.20	30	3.05	2.32	88.8
11	.58	1.32	.22	30	2.14	2.38	88.6
12	.52	1.53	.20	32	2.96	2.22	88.2
13	.47	.74	.11	33	1.43	1.75	87.8
14	.45	.72	.11	33	1.45	1.64	87.5
15	.75	1.57	.20	37	2.29	2.22	86.9
16	.94	1.70	.20	37	2.16	2.30	86.7
17	.46	1.00	.20	41	1.97		85.8
18	.50	1.14	.20	41	2.07		85.7
19	.35	1.62	.20	42	4.89		85.5
20	.33	1.66	.20	44	5.47		85.3
21	.26	1.16	.20	46	4.83		85.1
22	.30	1.23	.20	46	3.85		85.0
23	.44	1.40	.20	50	3.04	1.90	84.6
24	.50	1.38	.20	50	2.64	2.02	84.6
25	.52	1.46	.20	52	2.75	1.90	84.4
26	.70	1.35	.20	52	1.93	2.04	84.4
27			.10	53		1.44	84.3
28	.33	1.68	.20	66	5.60	1.90	84.9
29	.56	1.42	.20	67	2.44	2.31	85.1
30	.47	1.64	.20	72	3.70	2.34	86.7
31	.44	1.30	.20	72	2.74	2.22	87.0
32	.50	1.10	.20	77	2.00	2.34	88.2
33	.68	1.03	.20	77	1.44	2.24	88.5
34	.40	1.30	.20	80	2.97	2.28	91.7
35	.44	1.56	.20	80	3.60	2.37	92.2
36	.57	1.40	.20	73	2.36	2.33	97.1
37	.42	1.00	.20	73	2.14	2.28	97.6
38	.53	1.58	.20	71	3.06		98.4
39	.50	1.52	.20	71	3.05		98.9
40	.76	2.04	.20	70	4.00	2.24	99.8
41	.30	1.70	.20	70	6.35	2.24	100.4
42	.40	1.84	.20	67	5.33	2.32	102.3
43	.33	1.78	.20	67	6.26	2.15	102.9
44	.63	1.80	.20	64	3.40	2.30	104.5
REV D26							
1	.22	.38	.20	3	4.02	.82	96.2
2	.24	.38	.20	3	2.28	.68	98.8
3	.28	.48	.20	4	1.85	.90	98.3
4			.20	16		1.84	93.5
5			.21	16		2.06	93.2
6	.30	1.14	.21	18	3.54	1.78	92.5
7	.52	.91	.21	21	1.61	2.28	91.8
8	.52	1.80	.14	21	3.61	2.10	91.5
9	.28	.74	.21	24	2.67		90.6
10	.30	.54	.21	25	1.85	1.54	90.3
11	.31	.62	.21	25	1.99	1.82	90.1
12	.36	1.32	.21	27	3.43	1.76	89.6
13	.43	1.40	.21	27	3.10	1.92	89.3
14	.54	1.38	.20	28	2.43	1.50	89.1
15	.45	1.50	.21	29	3.31		88.8
16	.32	1.23	.21	32	3.59	2.32	88.1
17	.37	1.20	.21	32	2.94	2.16	87.9
18	.32	1.64	.21	34	5.54	1.80	87.5
19	.39	1.20	.21	35	2.75	2.10	87.3
20	.45	1.22	.21	35	2.47	2.12	87.1
21	.60	1.28	.21	36	2.01	1.87	86.9
22	.39	1.01	.21	39	2.30		86.4
23	.28	.94	.21	40	3.31	2.01	86.1
24	.32	.98	.21	41	2.84		85.9
25	.53	1.72	.21	42	3.58		85.7
26	.34	.90	.21	42	2.46		85.6
27	.34	1.20	.21	43	3.26		85.4
28	.27	.89	.21	44	3.44		85.3
29			.11	77		1.34	88.0
30			.11	77		1.67	88.4
31	.34	.65	.21	78	1.86	2.10	90.3
32	.34	.58	.21	78	1.72	2.00	90.6
33			.11	78		1.72	91.5
34			.11	78		1.57	91.8
35	.43	.89	.21	77	1.89	2.14	92.9
36	.38	.72	.21	77	1.76	2.00	93.3
37	1.24	1.74	.21	75	1.71	1.26	94.7
38	.97	1.58	.21	75	1.84	1.14	95.2
39	.48	1.33	.13	75	2.40	1.70	95.6
40	.78	1.40	.21	75	1.85	1.90	96.0
41	.38	1.52	.21	73	3.95	2.02	97.0
42	.34	1.51	.21	73	4.49	2.12	97.5

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## PHOTOGRAPHIC EVALUATION

ACC. NO.	DENSITY		DENSITY	SUN ANGLE (DEGREES)	BRIGHTNESS RANGE	CLOUD DMAX	ALTITUDE (NM)
	DMIN	DMAX	BASE + FOG				
REV D27							
1			.13	17		1.8D	96.2
2			.13	17		1.63	93.1
3	.62	1.04	.21	18	1.56	2.06	92.9
4	.51	1.21	.21	18	2.17	2.00	92.5
5			.12	23		1.35	90.9
6			.12	23		1.22	90.6
7			.12	25		1.59	90.2
8			.12	26		1.66	89.9
9			.12	26		1.62	89.6
10	.72	1.88	.22	37	3.36	2.42	86.8
11	.76	2.00	.20	37	3.78	2.42	86.6
12	.32	1.86	.21	51	7.08	2.30	84.5
13	.40	2.66	.21	51	59.57	2.28	84.5
14	.62	1.78	.21	54	3.36		84.3
15	.40	1.44	.23	54	3.44		84.3
REV D29							
1	.36	1.18	.20	57	2.98		96.2
2	.33	1.32	.22	57	3.81		111.2
3	.40	1.90	.21	54	5.74		112.8
4	.38	1.20	.22	54	2.84		113.7
REV D31							
1	.44	1.82	.20	42	4.80		96.2
2	.26	.96	.20	43	4.01		85.3
3	.30	1.32	.20	43	4.23		85.2
4	.24	1.34	.20	46	7.47		84.9
5	.30	.94	.20	46	2.93		84.8
6	.27	.92	.20	48	3.54	1.04	84.7
REV D32							
1	.36	1.42	.21	33	3.81		96.2
2	.27	1.39	.21	37	5.59		86.4
3	.42	1.72	.21	46	4.48		84.9
4	.27	.89	.20	50	3.44		84.5
5	.27	1.34	.21	49	5.30		84.5
6	.35	1.36	.21	52	3.71		84.4
7	.44	1.39	.21	53	3.01		84.3
REV D33							
1			.12	72		1.81	96.2
2			.12	72		1.9D	99.9
REV D34							
1			.20	18		1.92	96.2
2	.42	2.06	.20	63	6.88	2.46	84.6
3	.30	1.96	.20	63	8.69	2.30	84.7
REV D36							
1	.57	2.12	.20	16	5.56		96.2
2	.76	2.33	.20	16	6.47		92.2
3	.53	1.90	.20	21	4.41		90.8
4	.54	1.98	.21	21	4.83		90.5
5	.39	2.30	.21	25	10.86		89.4
6	.42	2.29	.21	25	10.02		89.2
7			.11	31		1.70	87.8
8			.10	31		1.60	87.5
REV D37							
1	.59	1.00	.21	18	1.56	2.00	96.2
2	.55	1.50	.21	18	2.70	1.85	91.5
3	.40	1.00	.21	26	2.23	1.73	89.0
4	.45	1.23	.21	26	2.50	1.50	88.7
5	.29	.95	.13	32	2.52	1.61	87.1
6	.35	1.12	.21	34	2.92	2.12	86.6
7	.47	1.15	.21	36	2.22		86.3
8	.50	1.66	.21	36	3.56		86.1
9	.29	.55	.21	39	1.97	1.80	85.7
10	.34	.58	.21	39	1.72	1.90	85.5
11	.70	1.62	.21	80	2.56	2.15	92.8

Handle via Byeman  
Controls Only~~TOP SECRET - GAMBIT~~

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PHOTOGRAPHIC EVALUATION

ACC. NO.	DENSITY		DENSITY	SUN ANGLE	BRIGHTNESS	CLOUD	ALTITUDE
	DMIN	DMAX	BASE + FOG	(DEGREES)	RANGE	DMAX	(NM)
REV D38							
1			.11	14		.94	96.2
2			.11	14		1.06	92.8
3	.72	.96	.21	19	1.29	1.94	91.2
4	.45	1.42	.21	21	3.04	1.55	90.5
5	.42	1.30	.21	21	2.85	1.48	90.2
6	.36	1.80	.21	24	5.77	1.82	89.7
7	.36	1.52	.21	24	4.23	1.49	89.5
8	.66	1.08	.21	26	1.55	1.38	88.9
9	.50	1.03	.21	26	1.87	1.91	88.6
10	.80	1.80	.21	27	2.79	2.16	88.2
11			.21	35		2.16	86.4
12			.13	35		2.01	86.2
13	.38	1.50	.21	45	3.87	1.40	84.7
14	.39	1.02	.21	45	2.32	1.60	84.7
15	.62	1.30	.21	47	2.00	1.70	84.5
16			.12	50		1.40	84.3
17			.12	52		1.60	84.2
18			.12	54		1.60	84.2
19			.12	54		1.60	84.2
20			.11	57		1.76	84.3
21	.34	1.33	.21	59	3.72	2.14	84.4
21			.11	59		1.56	84.4
22	.43	.80	.21	60	1.72	2.18	84.5
23	.43	.62	.21	61	1.40	2.34	84.5
24	.44	1.42	.22	62	3.11	2.32	84.6
25	.44	1.48	.22	62	3.31	2.40	84.7
26			.12	78		1.80	90.0
27	.80	1.20	.22	77	1.47	2.44	93.2
28	.70	1.16	.22	77	1.59	2.44	93.7
29	.63	1.60	.22	77	2.72	2.20	94.2
REV D39							
1	.47	.86	.21	11	1.70	1.74	96.2
2	.50	.88	.21	11	1.63	1.42	93.5
3			.12	28		1.48	88.2
4			.11	28		1.46	88.0
5	.54	1.30	.22	30	2.24	2.15	87.5
6	.32	1.55	.21	43	5.02		85.0
7	.30	1.54	.21	43	5.32		84.9
8	.49	1.32	.22	44	2.53	1.53	84.8
9	.28	1.58	.22	49	6.26		84.4
10	.32	1.54	.22	49	4.96		84.4
11	.34	2.22	.21	51	11.22	2.25	84.2
12	.40	2.32	.21	52	11.00	2.26	84.2
13	.36	2.08	.22	53	8.34	1.78	84.2
14	.30	1.80	.22	54	7.10	2.08	84.2
15	.25	2.31	.22	56	22.19	1.90	84.2
16	.25	2.14	.22	56	16.55	2.20	84.2
17			.10	64		1.60	84.8
18			.11	64		1.62	84.9
REV D40							
1			.22	8		1.30	96.2
2			.22	8		1.50	94.6
3	.62	1.50	.22	16	2.47	1.96	92.2
4	.44	1.32	.22	16	2.80	1.99	91.9
5			.11	18		1.53	91.5
6			.11	18		1.32	91.2
7	.52	1.12	.23	30	1.95	1.32	87.5
8	.48	1.52	.22	35	3.18		86.4
9	.59	1.52	.22	35	2.61		86.2
10	.67	1.69	.23	39	2.87		85.6
11	.65	1.90	.22	39	3.74		85.4
12	.26	2.08	.22	42	13.74		85.1
13	.27	1.81	.21	42	8.82		85.0
14	.40	1.10	.22	44	2.45		84.7
15	.48	1.23	.22	44	2.35		84.7
16	.28	2.26	.22	46	15.56	1.76	84.6
17	.44	1.26	.22	54	2.63		84.2
17	.72	1.10	.14	54	1.41		84.2
17	1.10	1.26	.22	54	1.17		84.2
17	.27	.42	.11	54	1.43		84.2
17	.73	1.62	.22	54	2.48		84.2
17	.28	.34	.12	54	1.18		84.2
17	.46	1.29	.22	54	2.60		84.2
17	.34	.52	.12	54	1.41		84.2
17	.86	1.28	.22	54	1.49		84.2
17	.28	1.48	.12	54	4.76		84.2
17	.73	1.98	.22	54	3.81		84.2

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PHOTOGRAPHIC EVALUATION

ACC. NO.	DENSITY		DENSITY	SUN ANGLE	BRIGHTNESS	CLOUD	ALTITUDE
	DMIN	DMAX	BASE + FOG	(DEGREES)	RANGE	DMAX	(NM)
REV D41							
1			.22	5		.78	96.2
2			.22	5		.74	95.6
3			.22	8		1.47	94.7
4			.21	9		2.00	94.4
5			.21	9		1.88	94.0
6	.56	1.38	.21	14	2.34	2.01	92.7
7	.48	1.16	.22	14	2.20	1.94	92.3
8			.12	15		1.36	91.7
9	.64	1.72	.22	18	3.07	2.03	91.0
10			.11	26		1.78	88.7
11			.11	26		1.84	88.5
12			.22	28		2.40	88.2
13			.22	27		2.32	88.0
14	.54	1.24	.22	30	2.10	2.25	87.5
15			.11	32		1.72	87.2
16			.11	32		1.84	87.0
17	.56	1.40	.12	33	2.30	2.30	86.7
18			.11	33		2.00	86.5
19	.64	1.83	.22	37	3.47		85.9
20	.73	2.04	.22	37	4.14		85.7
21	.68	1.72	.22	38	2.93		85.6
22	.55	1.48	.22	42	2.65	1.72	85.0
23	.58	1.48	.22	42	2.53	1.85	84.9
24	.30	1.10	.22	44	3.41		84.7
25	.40	1.56	.22	44	3.90		84.6
26	.41	1.44	.22	45	3.37		84.5
27	.29	1.22	.22	46	3.97	1.96	84.4
28			.11	52		1.65	84.1
29	.40	1.64	.22	54	4.26		84.1
30	.40	1.61	.22	54	4.12		84.1
31	.38	1.34	.13	59	2.94	1.40	84.4
32	.74	1.81	.22	59	3.03	1.57	84.4
33	.48	1.63	.22	68	3.58	2.28	86.1
34	.60	1.62	.22	68	2.88	2.27	86.3
35	.43	1.84	.22	72	5.01	2.34	86.9
36	.47	1.65	.22	72	3.74	2.34	87.1
37	.54	1.60	.21	77	3.07	2.10	90.6
38	.42	1.32	.22	77	2.91	2.15	91.0
39	.46	1.65	.22	74	3.82	2.34	96.9
40	.40	1.50	.22	74	3.66	2.37	97.4
41	.41	1.91	.22	74	5.71	1.92	98.1
42	.34	1.75	.22	74	5.85	2.22	98.5
REV D42							
1	.38	1.14	.22	4	2.68		96.2
2	.41	1.10	.22	4	2.40	1.18	96.0
3	.34	.80	.21	17	2.21	2.01	91.4
4	.35	.97	.22	17	2.54	2.11	91.1
5	.51	.83	.22	20	1.51	1.80	90.4
6	.46	1.31	.23	20	2.66	2.02	90.0
7	.38	.83	.23	22	1.99	1.95	89.4
8	.33	1.10	.23	23	3.07	1.80	89.2
9	.40	1.10	.23	25	2.45	1.57	88.9
10	.47	1.12	.23	25	2.16	1.63	88.6
11	.29	.75	.21	27	2.50	1.98	88.1
12	.41	1.11	.22	27	2.42	2.03	87.9
13	.35	1.18	.23	29	3.09		87.7
14	.53	1.51	.22	30	2.84		87.3
15	.57	1.60	.23	31	2.91		87.0
16	.38	1.38	.24	33	3.41		86.7
17	.49	1.52	.23	33	3.12		86.6
18	.48	1.19	.22	34	2.26		86.3
19	.38	1.37	.22	35	3.38		86.2
20	.49	1.45	.23	35	2.90		86.0
21	.44	1.42	.21	37	3.11		85.8
22	.55	1.58	.22	39	2.94		85.5
23	.60	1.83	.22	39	3.64		85.3
24	.56	1.25	.22	40	2.04	2.18	85.2
25	.34	.90	.23	42	2.46	1.98	85.0
26	.39	1.04	.22	42	2.36	1.62	84.9
27	.36	1.01	.22	43	2.55	2.03	84.8
28	.33	1.13	.23	46	3.16		84.5
29	.35	1.15	.22	46	3.00		84.5
30	.90	1.50	.22	55	1.81		84.2
31	.78	1.45	.22	55	1.95		84.2
32	1.30	1.60	.22	64	1.37		85.1
33	.90	1.22	.13	64	1.33		85.2
34	.41	.95	.23	73	2.08	1.82	87.3
35	.39	.82	.22	73	1.91	2.05	87.6
36	.33	1.00	.23	78	2.80	2.20	89.6
37	.40	1.11	.22	78	2.47	2.01	90.0
38	.38	.95	.23	77	2.25	2.14	90.8
39	.46	.94	.23	79	1.86	1.98	93.6
40			.12	78		1.86	95.1
41			.12	78		1.87	95.6
42			.12	76		1.66	96.2
43			.12	76		1.78	96.7
44	.50	1.58	.23	75	3.26	2.14	98.0
45	.45	1.51	.23	75	3.34	2.06	98.6
46	.90	1.68	.23	64	2.20	1.00	105.7
47	.77	1.53	.23	64	2.14	.86	106.4

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PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65

PHOTOGRAPHIC EVALUATION

ACC. NO.	DENSITY		DENSITY BASE + FOG	SUN ANGLE (DEGREES)	BRIGHTNESS RANGE	CLOUD DMAX	ALTITUDE (NM)
	DMIN	DMAX					
REV D43							
1			.22	3		1.28	96.2
2			.22	3		.94	96.3
3			.22	14		2.25	92.3
4	.34	.54	.22	16	1.61	2.13	91.5
5	.36	.64	.22	16	1.72	2.15	91.3
6	.52	2.24	.22	18	7.50	2.00	91.0
7	.50	.84	.22	20	1.56	2.35	90.4
8	.27	.45	.14	20	1.52	2.12	90.1
9	.36	1.35	.22	22	3.54		89.7
10	.32	1.46	.22	22	4.57		89.5
11			.12	25		1.96	88.6
12			.12	25		1.97	88.4
13	.68	1.12	.14	29	1.51	2.17	87.7
14	1.05	1.45	.23	29	1.49	2.40	87.5
15	.48	.78	.12	31	1.50	1.85	87.3
16	.43	.72	.12	31	1.55	1.97	87.0
17	.64	1.44	.22	33	2.27	2.38	86.7
18	.62	1.40	.22	33	2.22	2.42	86.5
19	.57	1.42	.23	34	2.41	2.46	86.2
20	.38	1.25	.22	42	2.98	2.44	85.0
21	.43	1.56	.22	42	3.67	2.44	84.9
22	.30	1.15	.22	48	3.57	1.42	84.4
23	.31	1.08	.22	48	3.23	.85	84.3
24	.85	1.88	.22	52	2.89		84.2
25	.54	1.65	.13	52	3.02		84.2
26	.89	1.75	.22	53	2.40	1.77	84.2
27			.11	61		.98	84.6
28			.11	61		.95	84.8
29	.72	1.22	.22	69	1.64	2.22	86.1
30	.90	1.45	.22	69	1.71	2.20	86.3
31	.42	.82	.22	74	1.79	2.26	88.0
32	.48	.67	.22	74	1.34	2.17	88.3
33	.40	1.68	.22	75	4.46	2.28	88.6
REV D44							
1	.43	1.56	.22	6	3.67		96.2
2	.24	1.90	.22	5	13.85		95.2
3	.81	1.00	.22	23	1.21	2.22	89.4
4	.64	.86	.22	23	1.29	2.08	89.1
5	.34	.63	.22	52	1.82	2.00	84.1
6	.40	.66	.22	52	1.55	2.20	84.1
REV D47							
1			.21	7		.97	96.2
2			.21	7		.88	94.0
3	.38	1.76	.21	23	5.15		88.9
4	.60	1.72	.22	23	3.22		88.6
5			.10	36		1.02	85.8
6			.13	36		1.56	85.7
7	.28	.60	.21	42	2.28	1.86	84.8
8	.35	1.28	.21	48	3.41		84.1
9	.28	1.48	.21	50	5.63		84.1
10	.26	1.53	.21	52	7.05		84.1
11	.31	1.50	.22	52	4.93		84.1
12			.22	72		2.06	87.2
13			.22	72		1.66	87.5
14	.46	1.28	.22	77	2.58	2.20	91.1
15	.36	1.58	.22	77	4.51	2.18	91.4
REV D48							
1			.13	21		2.05	96.2
1			.23	21		2.10	89.4
1			.13	21		1.72	89.4
2			.22	25		1.52	88.3
3			.22	25		1.50	88.1
4	.33	.80	.22	35	2.29	1.43	86.0
5	.31	.79	.22	35	2.43	1.15	85.8
6	.45	1.35	.21	40	2.83		85.1
7	.55	1.46	.22	40	2.59		85.0
8	.65	1.39	.22	43	2.13		84.6
9	.81	1.55	.22	43	2.09		84.5
10	.56	1.77	.22	44	3.57		84.5
11	.32	1.56	.22	45	5.07		84.4
12	.32	1.16	.20	52	3.36	2.00	84.1
REV D49							
1			.22	68		2.29	96.2
2			.22	68		2.27	103.8
REV D50							
1			.22	16		1.44	96.2
2			.23	16		1.50	90.5
3	.23	1.76	.23	24	14.20	1.86	88.3
4	.23	1.70	.23	24	13.29	1.87	88.1

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PHOTOGRAPHIC EVALUATION

ACC. NO.	DENSITY		DENSITY BASE + FOG	SUN ANGLE (DEGREES)	BRIGHTNESS RANGE	CLOUD DMAX	ALTITUDE (NM)
	DMIN	DMAX					
REV D52							
1	.37	2.37	.22	14	13.60		96.2
2	.38	2.26	.22	14	10.47		90.9
3	.48	2.06	.22	16	6.09		90.4
4	.56	2.26	.22	20	7.17		89.2
5	.54	2.47	.22	20	12.10		89.0
6	.94	2.18	.22	23	4.10	1.76	88.6
7	.88	2.14	.22	23	4.04	1.66	88.3
REV D53							
1			.23	18		1.76	96.2
2			.23	17		1.76	89.7
3	.55	2.16	.22	25	6.15	1.97	87.9
4	.55	2.10	.22	25	5.58	2.06	87.7
5	.49	.63	.22	25	1.25	1.98	87.5
6			.13	41		1.14	84.6
7			.13	41		1.04	84.5
8	.86	1.34	.23	77	1.59	2.20	91.4
9	1.09	1.42	.23	77	1.39	2.20	91.8
10	.46	1.16	.23	75	2.29		97.5
11	.49	1.06	.23	75	1.96		98.0
12	.80	1.30	.23	69	1.63	1.76	103.1
13	.46	1.18	.23	69	2.33	1.68	103.7
14	.59	1.10	.15	61	1.67		109.4
15	.86	1.33	.23	61	1.57		110.0
REV D54							
1			.12	12		1.60	96.2
2			.21	12		1.95	91.5
3			.12	13		1.20	91.0
4	.86	1.36	.21	21	1.62		88.7
5	.77	1.55	.22	21	2.19		88.5
6	.41	1.01	.22	23	2.20	2.00	88.2
7	.47	1.04	.22	23	2.01	2.03	88.0
8			.11	25		1.18	87.7
9			.11	26		1.43	87.3
10			.11	26		1.36	87.1
11	.49	1.06	.22	36	1.96	2.40	85.4
12	.58	1.08	.21	36	1.70	2.32	85.2
13	.33	1.36	.21	38	3.97	1.93	85.0
14	.42	1.68	.21	39	4.28	2.30	84.8
15	.45	1.84	.21	39	4.81	2.36	84.7
16	.49	1.78	.22	41	4.15		84.5
17			.11	43		1.32	84.4
18			.11	43		1.18	84.3
19			.11	44		1.20	84.2
20	.56	1.20	.22	46	1.94	1.82	84.1
21	.49	1.10	.22	46	2.04	1.95	84.1
22	.32	1.12	.21	47	3.24	1.08	84.0
23	.40	1.00	.22	48	2.23	2.13	84.0
24	.40	1.32	.21	48	3.03	2.06	83.9
25			.21	51		2.20	83.9
26			.11	53		1.74	83.9
27			.11	53		1.72	84.0
28			.11	55		1.78	84.0
29			.11	57		1.42	84.2
30			.11	58		1.36	84.3
31	.48	1.12	.22	59	2.12	2.24	84.4
32	.33	.96	.21	76	2.69	2.33	89.4
33	.30	.90	.21	76	2.83	2.30	89.7
34			.21	78		2.32	91.0
35			.21	78		2.38	91.4
36	.44	.70	.21	79	1.50	2.31	91.9
37	.40	1.18	.22	78	2.64	1.90	92.5
38	.43	.98	.21	78	2.06	2.08	92.8
39	.86	1.88	.21	77	2.86	1.61	94.4
40	.72	1.72	.22	77	2.80	1.56	94.9
REV D55							
1			.13	11		1.54	96.2
2			.13	11		1.43	91.4
3	.40	1.02	.22	31	2.27	2.18	86.3
4	.38	1.15	.22	33	2.71		85.9
5	.43	1.00	.22	33	2.10		85.8
6	.66	1.86	.22	43	3.51	2.34	84.3
7	.74	1.68	.22	43	2.62	2.34	84.2
8	.46	1.37	.22	44	2.83	1.84	84.2
9	.83	1.12	.13	46	1.29	1.76	84.1
10	.53	2.30	.22	48	8.16	2.24	84.0
11	.58	2.08	.22	48	5.20	2.25	84.0
12	.28	2.32	.22	51	17.28	2.30	83.9
13	.25	.95	.22	53	3.97	1.97	84.0
14	.24	1.88	.22	54	13.47	2.00	84.0
15	.28	2.28	.22	54	16.11	2.08	84.1
16	.26	1.38	.22	55	6.03	1.88	84.1
17	.28	1.26	.22	58	4.47	2.28	84.3
18	.32	1.34	.22	58	4.03	2.30	84.4
19	.40	1.22	.22	61	2.74	2.25	84.7
20	.46	1.20	.22	61	2.38	2.38	84.8
21	.40	1.35	.22	66	3.13	2.32	85.6
22	.40	1.16	.22	66	2.59	2.38	85.8

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ACC. NO.	DENSITY		DENSITY BASE + FOG	SUN ANGLE (DEGREES)	BRIGHTNESS RANGE	CLOUD DMAX	ALTITUDE (NM)
	DMIN	OMAX					
REV 056							
1	.56	1.10	.22	7	1.77	1.56	96.2
2	.65	1.05	.21	7	1.52	.70	93.0
3			.22	14		1.86	90.9
4	.48	1.08	.23	16	2.04	1.80	90.3
5	.44	1.70	.22	16	4.20	1.93	90.1
6			.22	18		1.82	89.7
7			.22	17		1.70	89.5
8	1.00	1.40	.22	28	1.48		86.9
9	.52	1.29	.22	30	2.30		86.5
10	.31	1.54	.22	30	5.14		86.4
11	.51	.98	.22	32	1.75		85.9
12	.51	1.42	.21	34	2.69		85.6
13	.86	2.03	.22	34	3.52		85.5
14	.43	1.80	.22	37	4.80		85.2
15	.62	1.90	.22	37	3.87		85.1
16	.62	1.48	.22	40	2.42		84.7
17	.46	1.17	.22	40	2.31		84.6
18	.40	1.05	.22	41	2.33		84.5
19			.13	45		1.73	84.2
20			.13	45		1.82	84.2
21	.48	1.09	.22	47	2.06	1.48	84.1
22	.50	2.00	.22	49	5.38	2.02	84.0
23	.42	2.26	.22	50	9.52		84.0
24	.38	1.38	.22	51	3.41		84.0
25	.56	1.56	.22	52	2.82		84.1
REV 057							
1			.22	4		1.10	96.2
2	.30	.40	.22	4	1.39	.80	93.8
3			.22	14		1.24	90.8
4			.22	14		1.13	90.5
5			.12	17		1.02	89.9
6			.14	17		1.24	89.6
7	.40	1.16	.22	17	2.59	2.32	89.4
8	.56	1.28	.22	27	2.10	2.03	87.0
9	.56	.72	.22	28	1.20	2.16	86.8
10			.12	29		1.64	86.6
11	.53	.90	.22	30	1.56	2.32	86.3
12	.55	.87	.22	32	1.45	2.16	86.0
13	.66	.76	.21	32	1.12	2.16	85.9
14	.26	.46	.12	34	1.60	1.71	85.7
15	.86	1.20	.22	34	1.37	2.38	85.5
16	.79	.99	.22	36	1.22	2.33	85.2
17			.12	36		2.10	85.1
18			.12	38		1.88	84.8
19	.36	1.59	.21	45	4.57	1.71	84.2
20	.32	1.60	.21	45	5.30	2.10	84.1
21	.52	2.14	.22	59	6.31		84.5
22	.59	2.12	.22	59	5.43		84.6
REV 058							
1	.50	2.20	.22	11	7.29	1.96	96.2
2	.47	2.36	.22	11	10.53	1.83	91.4
3	.38	.80	.22	20	1.93	2.23	88.6
4	.33	1.24	.22	22	3.50	2.33	88.2
5	.35	1.90	.22	22	6.73	2.32	88.0
6	.32	1.80	.22	23	6.62	2.16	87.8
7			.22	26		2.23	87.2
8			.22	26		2.14	87.1
9	.34	1.32	.21	29	3.68	1.15	86.5
10	.38	1.66	.22	29	4.61	.76	86.3
11	.34	1.34	.22	31	3.76		86.0
12	.43	1.26	.22	31	2.68		85.9
13	.38	1.03	.22	34	2.42		85.5
14	.48	1.25	.22	34	2.40		85.4
15	.48	1.64	.22	36	3.62		85.1
16	.46	1.66	.22	36	3.86		85.0
17	.51	1.51	.22	40	2.96		84.7
18	.73	1.70	.22	40	2.71		84.6
19	.84	1.46	.22	54	1.84	1.76	84.1
20	.78	1.53	.22	54	2.12	2.06	84.1
REV 059							
1			.24	3		1.18	96.2
2			.23	3		.98	94.0
3	.26	1.27	.23	31	5.37	2.39	86.0
4	.35	1.30	.23	32	3.48	1.83	85.8
5	.44	1.11	.23	33	2.28		85.6
6	.52	1.30	.23	33	2.33	2.20	85.4
7			.23	35		2.22	85.2

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PERFORMANCE EVALUATION TEAM  
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PHOTOGRAPHIC EVALUATION

ACC. NO.	DENSITY		DENSITY	SUN ANGLE	BRIGHTNESS	CLOUD	ALTITUDE
	DMIN	DMAX	BASE + FOG	(DEGREES)	RANGE	DMAX	(NM)
REV U63							
1			.22	9		.73	96.2
2			.22	26		1.60	86.7
2	.62	.84	.13	26	1.28	.79	86.7
2	1.09	1.36	.22	26	1.31	1.36	86.7
3	.26	.90	.22	39	3.79	.86	84.5
4	.25	1.02	.22	44	4.66	1.02	84.1
5	.34	1.39	.22	60	3.96	2.28	84.8
6	.38	1.49	.22	60	3.83	2.30	84.9
REV D64							
1	.42	.63	.23	13	1.44	1.22	96.2
2	.38	1.58	.23	17	4.21	1.40	89.0
2			.14	17		1.10	89.0
2			.23	17		1.38	89.0
3			.23	24		1.70	87.0
4	.28	.36	.23	32	1.39	.44	85.6
5	.29	.36	.23	32	1.28	.47	85.5
6	.42	1.40	.22	36	3.17		84.9
7	.43	1.44	.23	36	3.24		84.8
8	.30	1.08	.23	45	3.34	.90	84.1
9	.37	1.15	.22	45	2.80	.70	84.0
10	.33	1.36	.23	52	3.97	1.67	84.0



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APPENDIX B

Handle via Byeman  
Controls Only

~~TOP SECRET - GAMBIT~~

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PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65

PHOTOGRAPHS NO. 14 & NO. 15

These two photographs provide a comparison of duplicate photography obtained of radio towers and bridge at Milwaukee, Wisconsin, on Missions 4017 and 4022.

PHOTOGRAPH NO. 14  
(Mission 4017)

Rev D31            Frame 003

Altitude: 87.55 NM      Obliquity: +10.0°  
Scale:    1:84,670          Strip

20X

PHOTOGRAPH NO. 15

Rev D31            Frame 001

Altitude: 85.4 NM        Obliquity: -24.4°  
Scale:    1:89,200            Strip

20X

Handle via Byeman  
Controls Only

~~TOP SECRET - GAMBIT~~

PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65



PHOTOGRAPH 14

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~~TOP SECRET - GAMBIT~~

PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65



PHOTOGRAPH 15

E-3

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PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65

PHOTOGRAPHS NO. 16 & NO. 17

These two photographs provide a comparison  
of duplicate photography obtained of a gas  
storage tank in Chicago, Illinois

PHOTOGRAPH NO. 16  
(Mission 4017)

Rev D31                      Frame 004

Altitude: 87.3 NM                      Obliquity: 26.6°  
Scale: 1:92,300                      Strip

40X

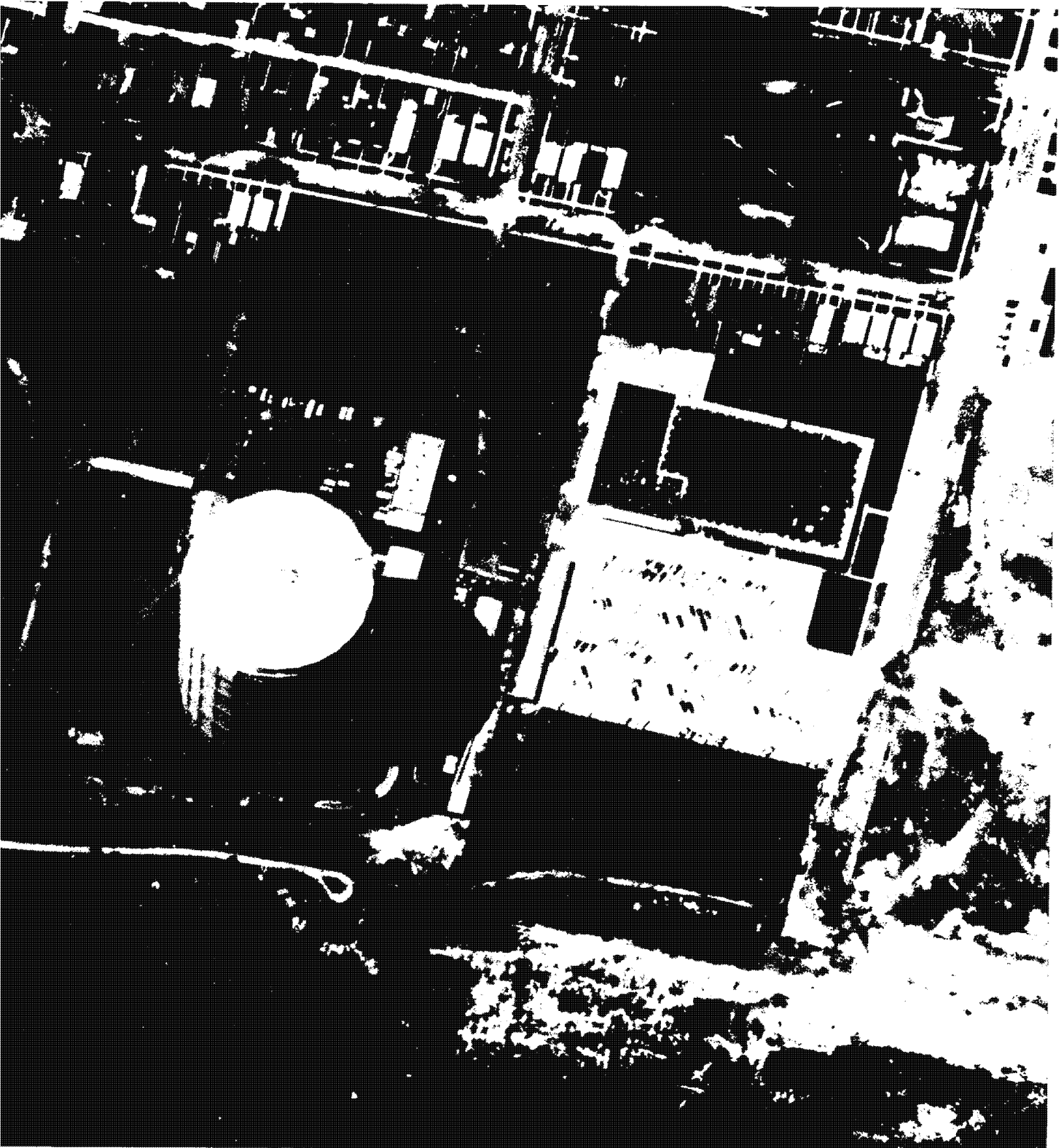
PHOTOGRAPH NO. 17

Rev D31                      Frame 003

Altitude: 85.3 NM                      Obliquity: -13.0°  
Scale: 1:86,100                      Stereo

40X

PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65



PHOTOGRAPH 16

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~~TOP SECRET - GAMBIT~~

PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65



PHOTOGRAPH 17

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PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65

PHOTOGRAPH NO. 18

Football stadium with game in progress -  
Chicago, Ill.

Rev D31

Frame 003

Altitude: 85.2 NM  
Scale: 1:85,900

Obliquity: -13.0°  
Stereo

20X



~~TOP SECRET - GAMBIT~~

PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65



PHOTOGRAPH 18

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~~TOP SECRET - GAMBIT~~

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PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65

PHOTOGRAPH NO. 19

Midway Airport, Chicago, Illinois

Rev D31

Frame 002

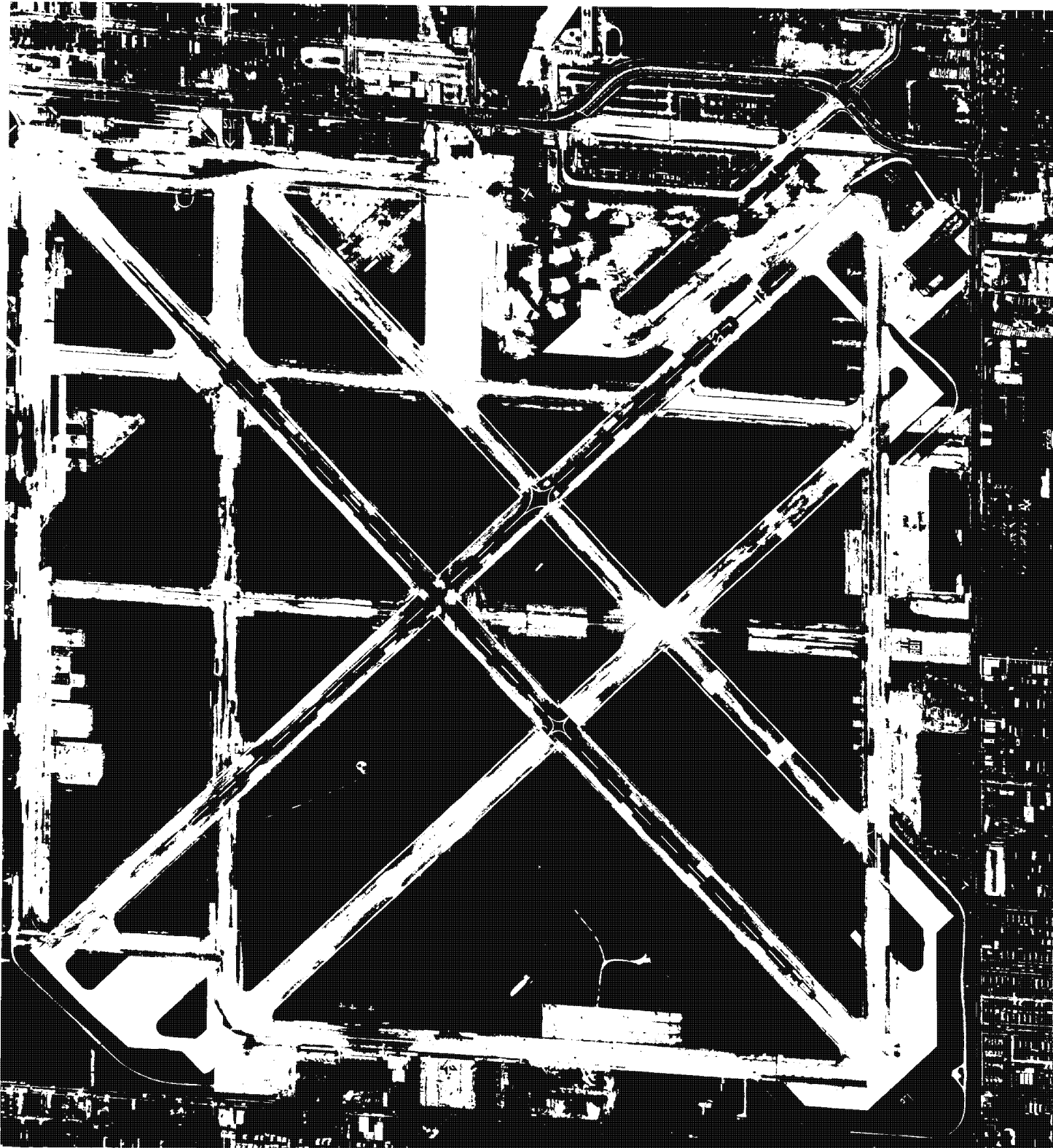
Altitude: 85.3 NM  
Scale: 1:86,100

Obliquity: -13.0°  
Stereo

10X

~~TOP SECRET - GAMBIT~~

PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65



PHOTOGRAPH 19

~~TOP SECRET - GAMBIT~~

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PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65

PHOTOGRAPH NO. 20

Hangar buildings, Lowry AF Base, Colorado

Rev D48

Frame 011

Altitude: 84.4 NM

Obliquity: +6.0°

Scale: 1:80,500

Strip

20X

PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65



PHOTOGRAPH 20

~~TOP SECRET - GAMBIT~~

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PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65

PHOTOGRAPH NO. 21

Drive-in Theatre at Phenix City, Alabama  
Note Speaker Stands

Rev D47      Frame 010

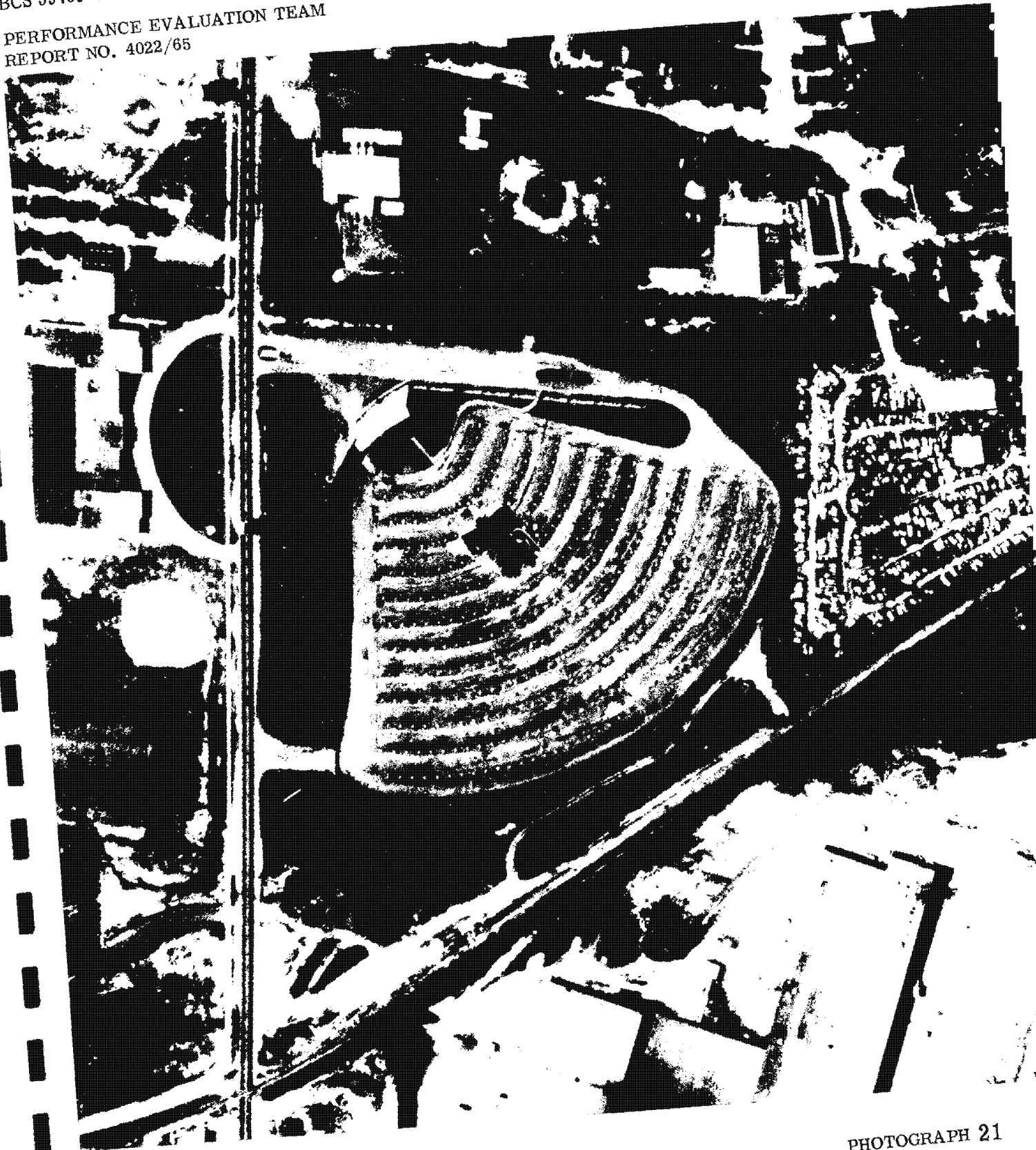
Altitude: 84.1 NM  
Scale: 1:87,907

Obliquity: -19.7°  
Strip

40X

BCS 39739-65

PERFORMANCE EVALUATION TEAM  
REPORT NO. 4022/65



PHOTOGRAPH 21

B-14

~~TOP SECRET - GAMBIT~~